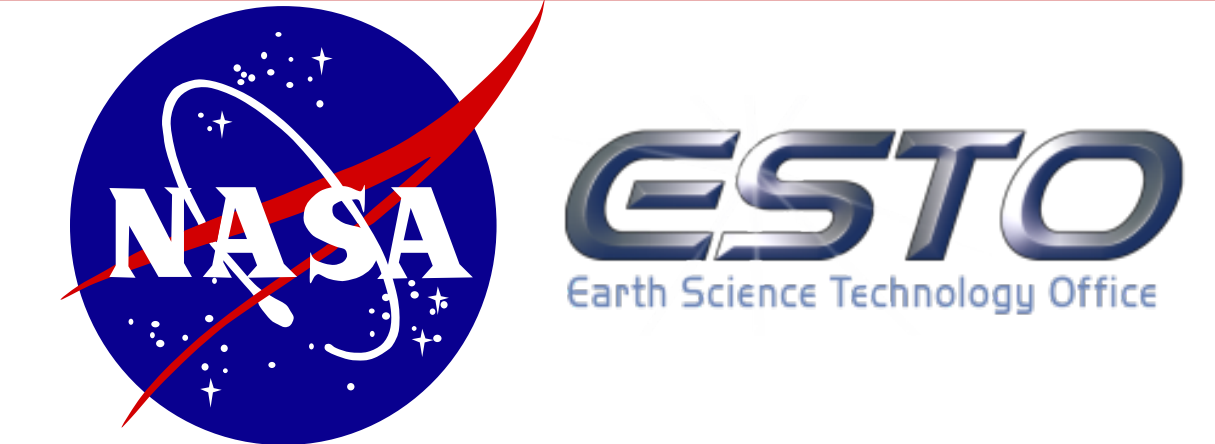




# The CubeSat Radiometer Radio Frequency Interference Technology Validation (CubeRRT) Mission: Status and Continuing Operations



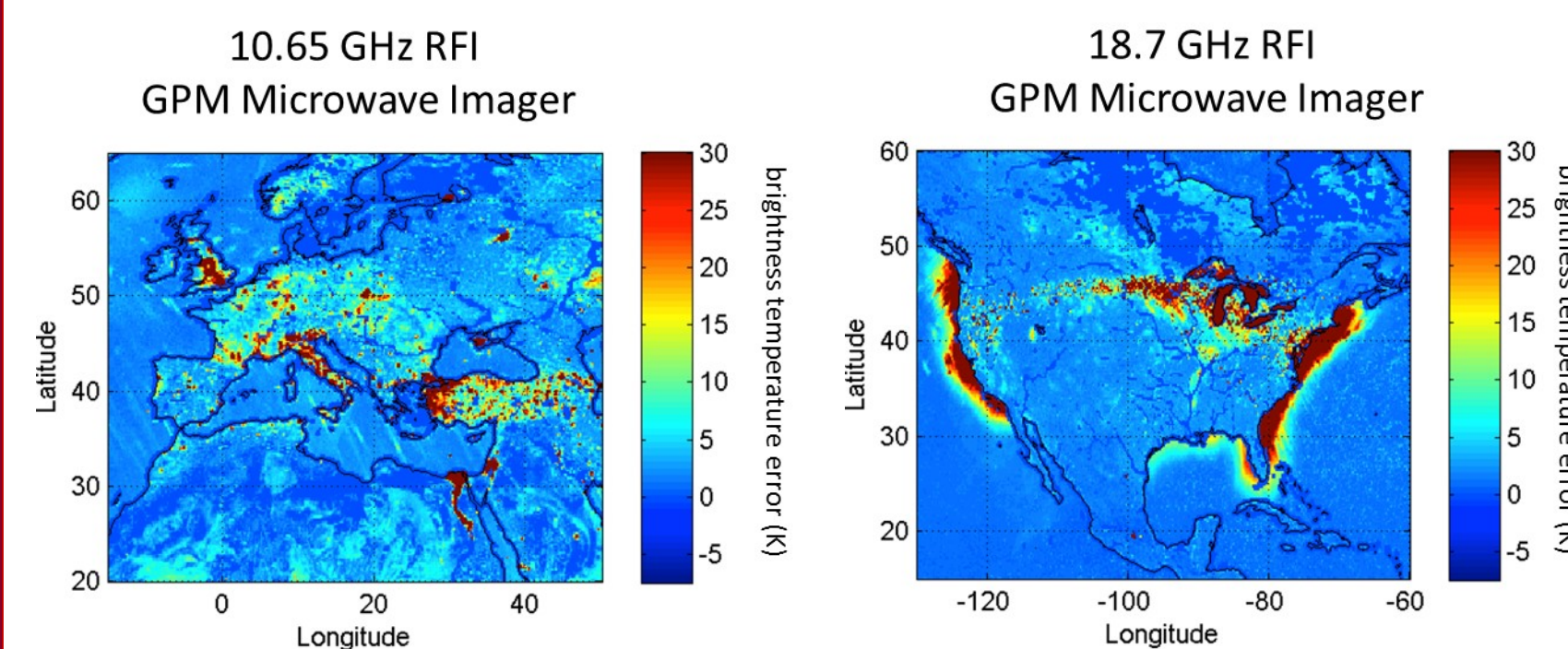
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## INTRODUCTION

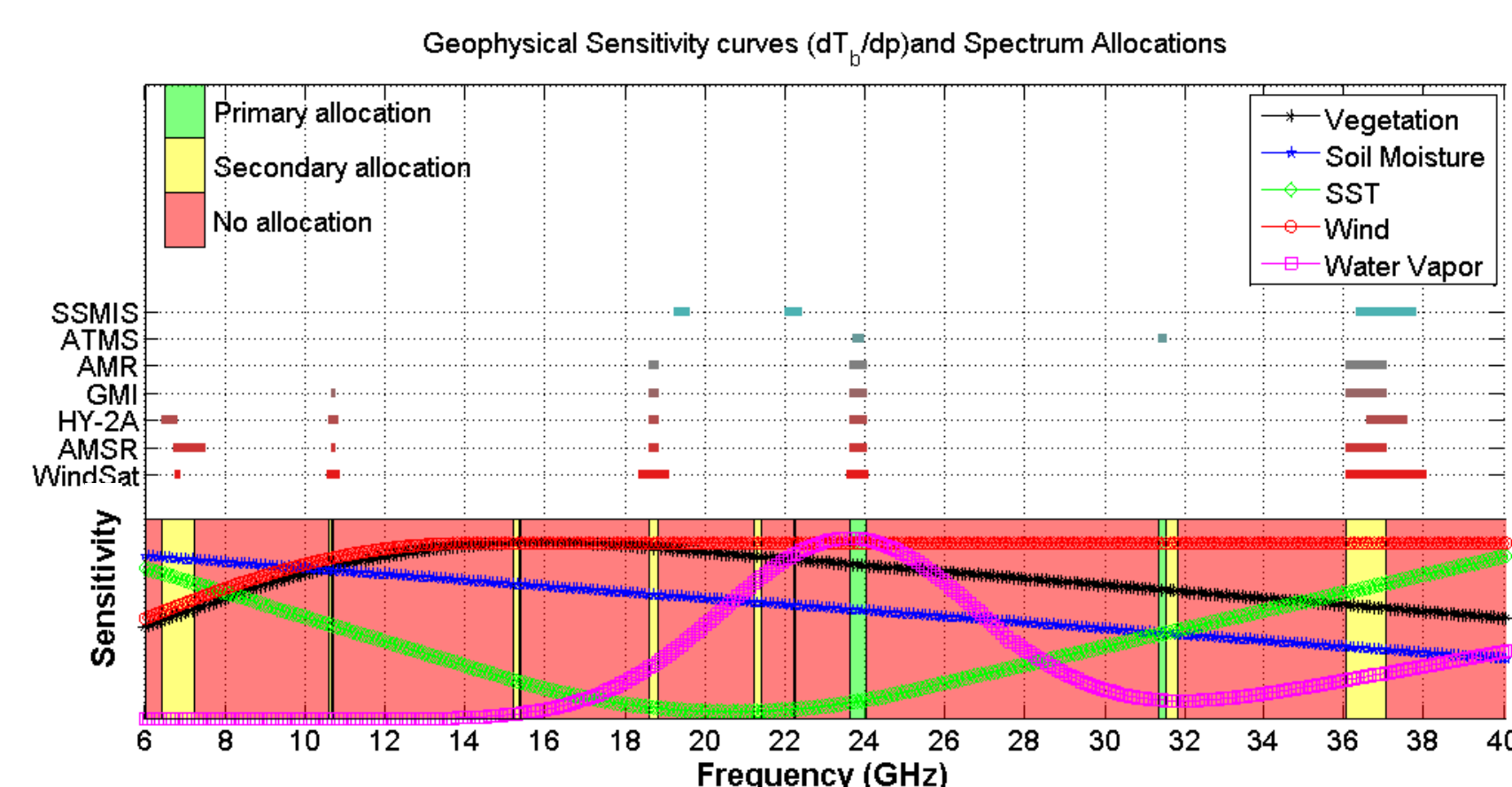
Passive microwave systems provide important data products for Earth science applications by measuring the naturally generated microwave thermal emission from Earth. Data from these highly sensitive instruments can be corrupted by radio-frequency interference (RFI). Recent measurements below 40 GHz have shown an increasing presence of RFI, making it more difficult to perform radiometry without an RFI mitigation capability. To address this challenge, the CubeRRT mission has demonstrated a real-time onboard RFI processor with low SWaP requirements operating on a 6U CubeSat.

## RFI PROBLEM FOR MICROWAVE RADIOMETRY



Images from NASA's GMI radiometer (above) at 10.65 and 18.7 GHz show the clear corrupting influence of RFI. Ideally, radiometers avoid RFI by operating in bands where transmission is prohibited. In practice, radiometers must co-exist with RFI sources due to shared spectrum allocations or operation in unprotected bands. Co-existence in some cases may be possible provided that a subsystem for mitigating RFI is included in future systems.

## Spectrum allocations 6-40 GHz

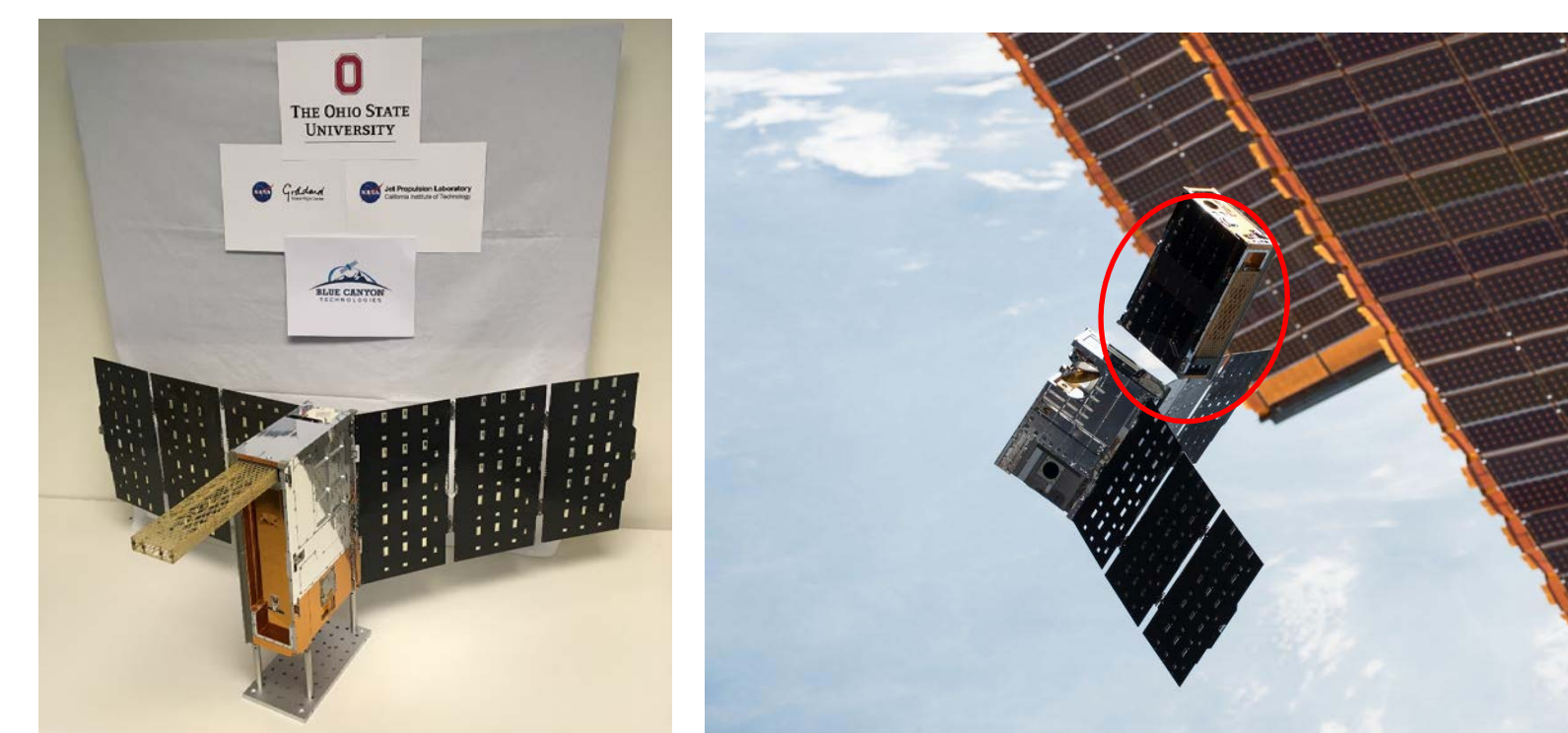


## CUBERTT MISSION

The CubeRRT mission's primary objective is to demonstrate an RFI processor for future spaceborne microwave radiometers that provides real-time, onboard RFI filtering.

- 6U CubeSat (30 cm x 20 cm x 10 cm) housing radiometer front end, digital backend, and wideband antenna systems
- 1 GHz instantaneous bandwidth
- Real-time on-board RFI detection and removal using digital backend
- 6 to 40 GHz operational range, tunable
- CubeRRT launched to ISS on May 21<sup>st</sup>, 2018 and was deployed into orbit on July 13<sup>th</sup>

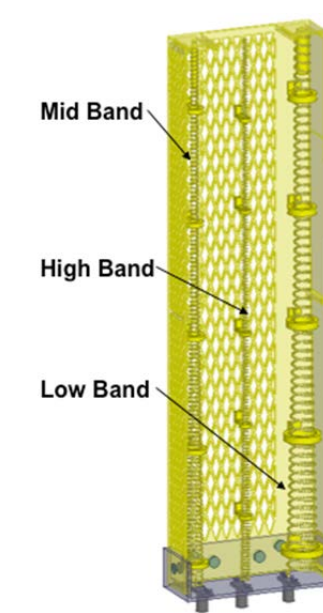
## CubeRRT satellite and ISS Deployment



Mission Properties	
Frequency	6 to 40 GHz Tunable, 1 GHz instantaneous
Polarization	Circular polarization
Observation angle/Orbit (ISS launch)	0° Earth Incidence Angle 400 km altitude, 51.6° orbit inclination
Spatial Resolution	80 km (40 GHz) to 240 km (6 GHz)
Integration time	100 msec
Ant Gain/ Beamwidth	12dBi/30° (6 GHz), 21 dBi/10° (40 GHz)
Interference Mitigation	On-board Nyquist sampling of 1 GHz spectrum; On-board real-time Kurtosis and Cross-Frequency Detection Downlink of frequency resolved power and kurtosis in 128 channels to verify on-board performance
Calibration	Internal: Reference load and Noise diode sources External: Cold sky and Ocean measurements
Noise equiv dT	0.8 K in 100 msec (each of 128 channels in 1 GHz)
Average Payload Data Rate	9.375 kbps (including 25% duty cycle) ~102 MB per day, ~ 37 GB over 1 year mission life
Downlink	135 MB per daily ground contact [6 minute contact with 3 Mbps UHF cadet Radio] 32% margin over payload data

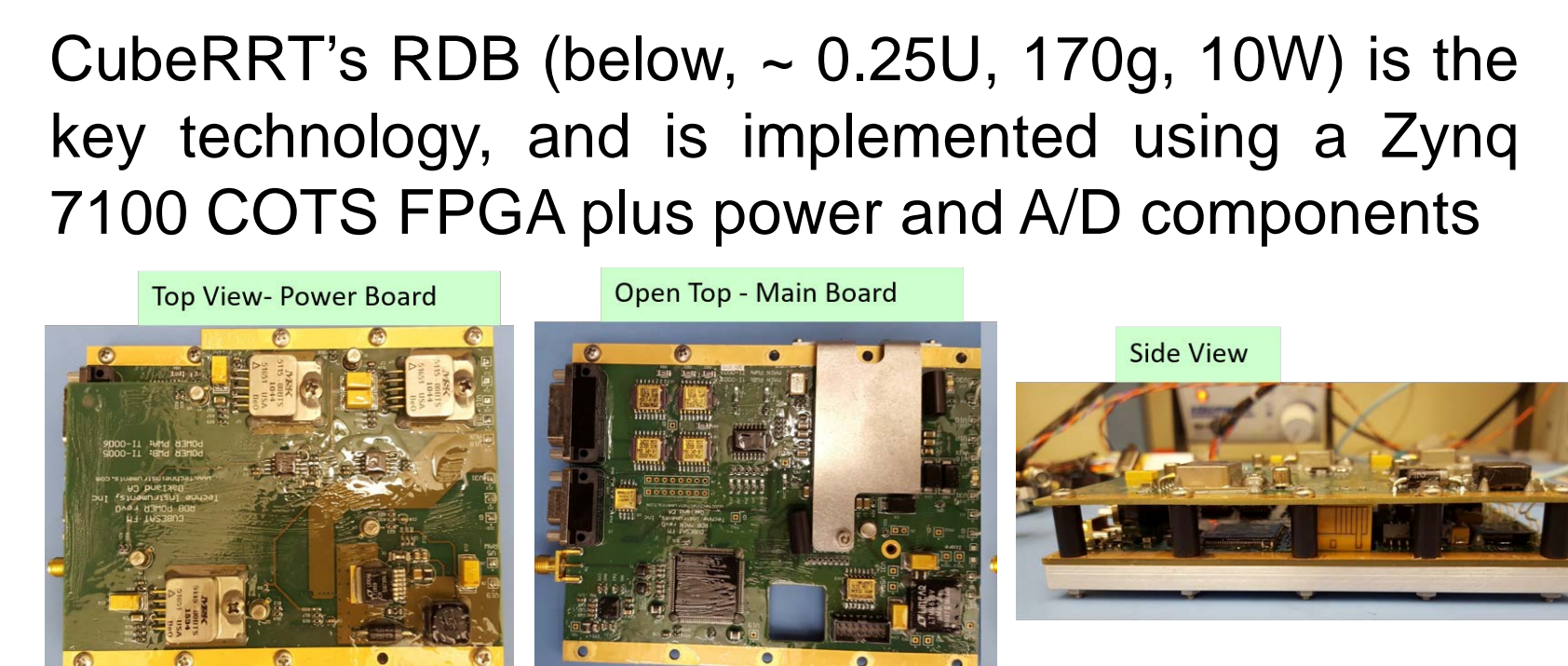
## CUBERTT PAYLOAD SUBSYSTEMS

Subsystems include the 6-40 GHz antenna (ANT), the Radiometer Front End (RFE), and the Radiometer Digital Backend RFI (RDB) processor.

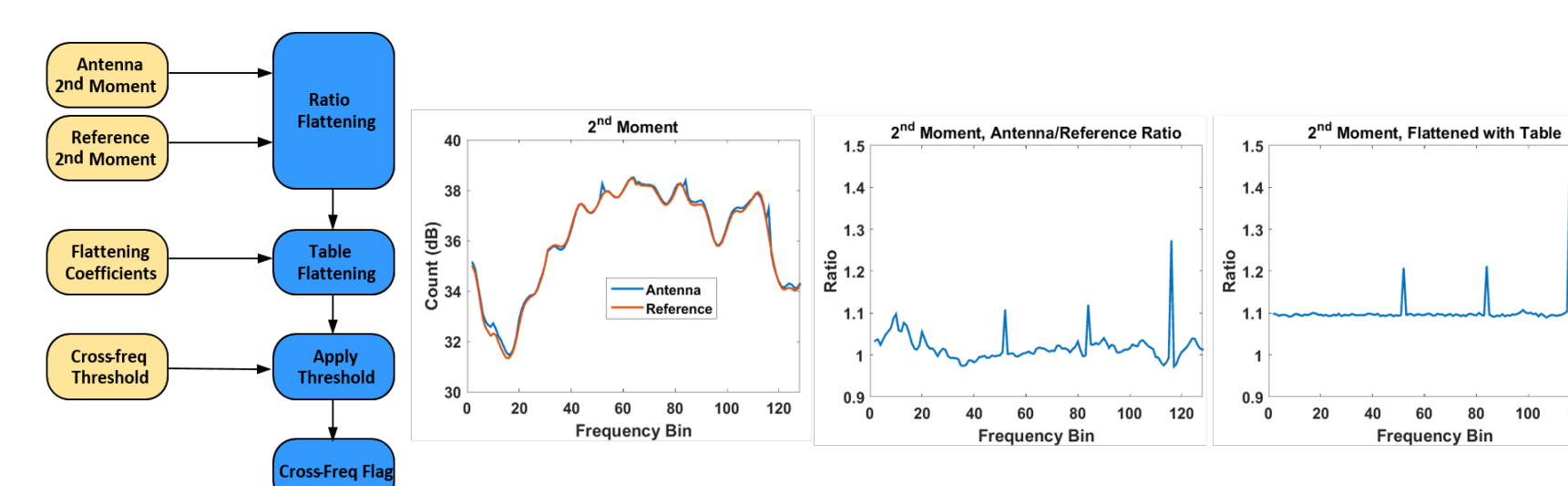
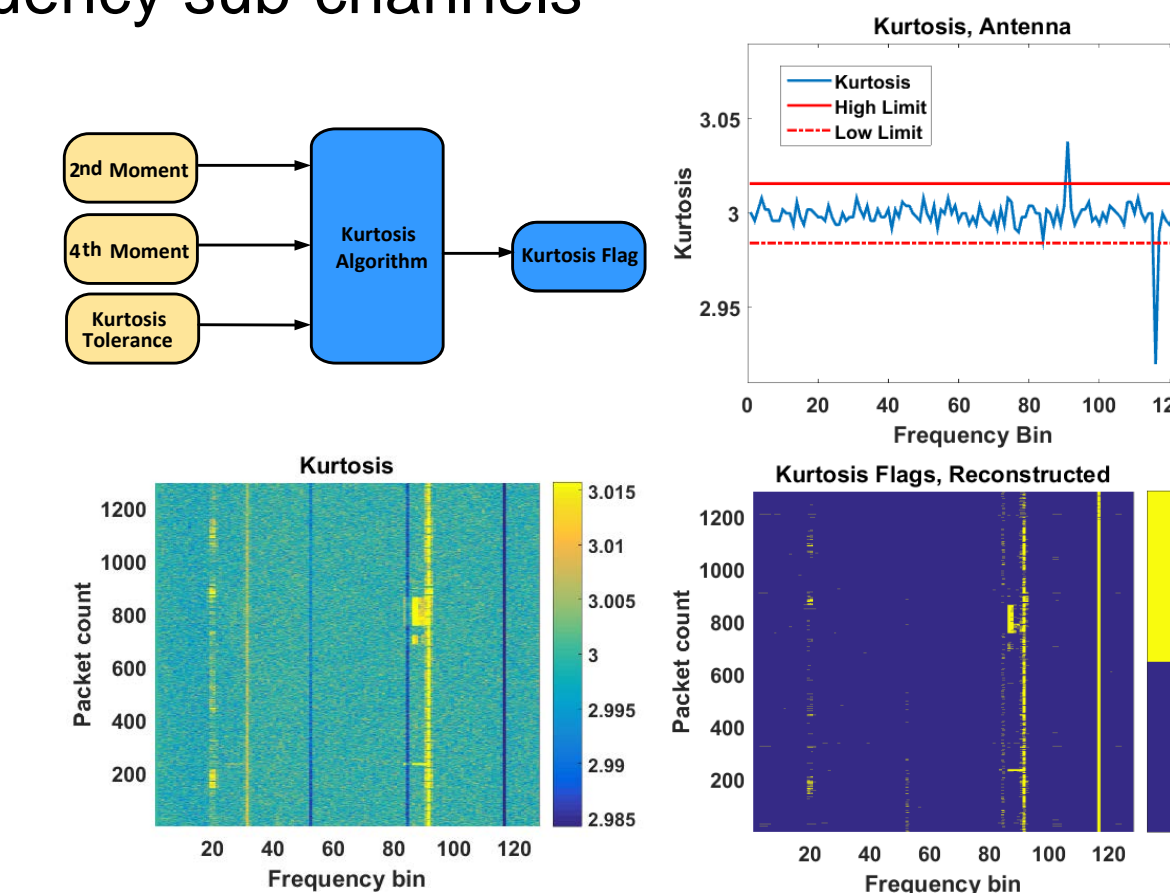


CubeRRT's payload antenna (left) includes 3 tapered helix elements to cover 6-40 GHz with gain 7 to 21 dBi in a 0.25U, 116g package

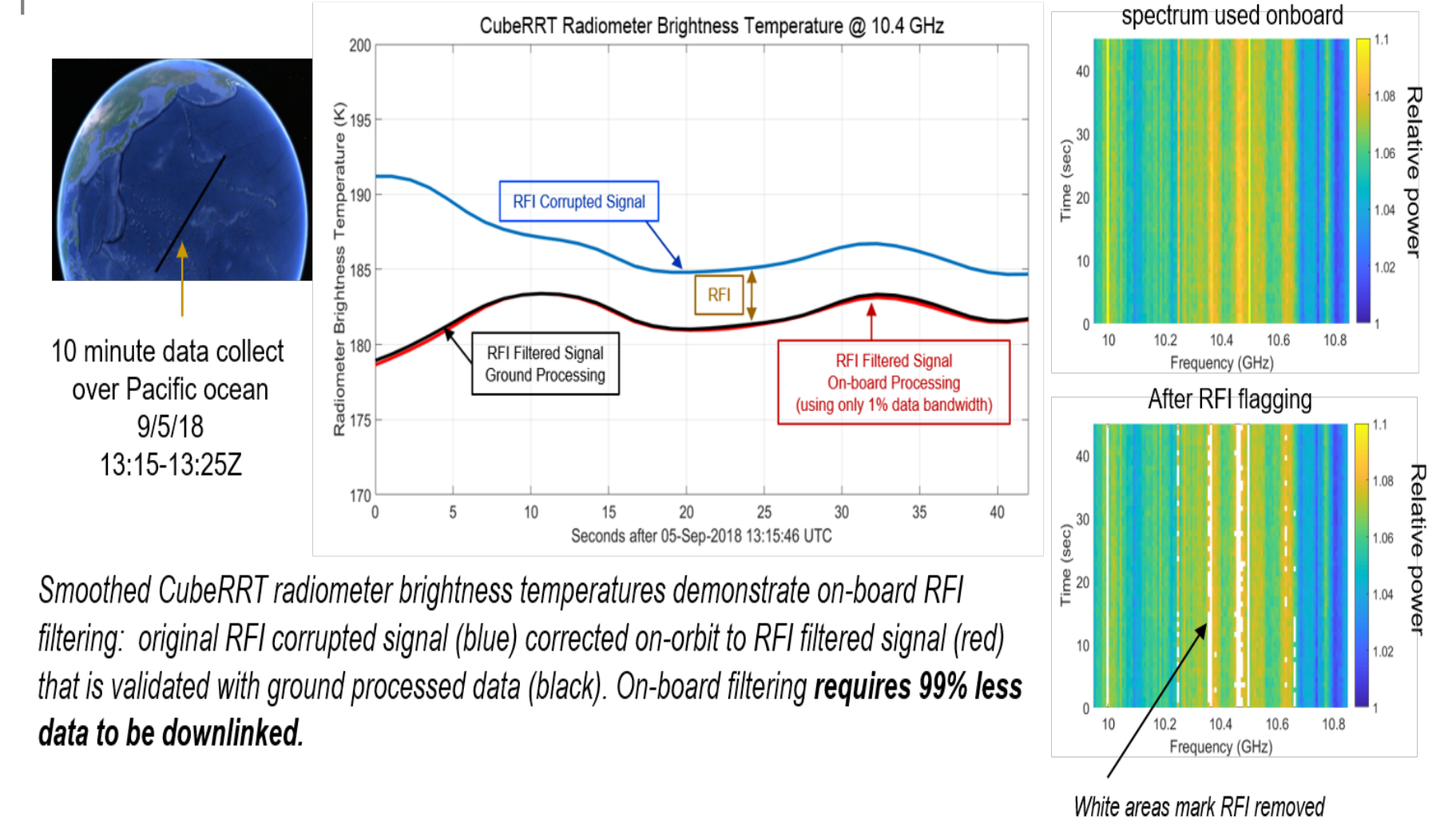
CubeRRT's RFE (right) tunes 6-40 GHz signals to 1-2 RF GHz IF of RDB



The RDB implements both kurtosis and cross-frequency methods for RFI detection using 128 frequency sub-channels



## FIRST LIGHT DATASET

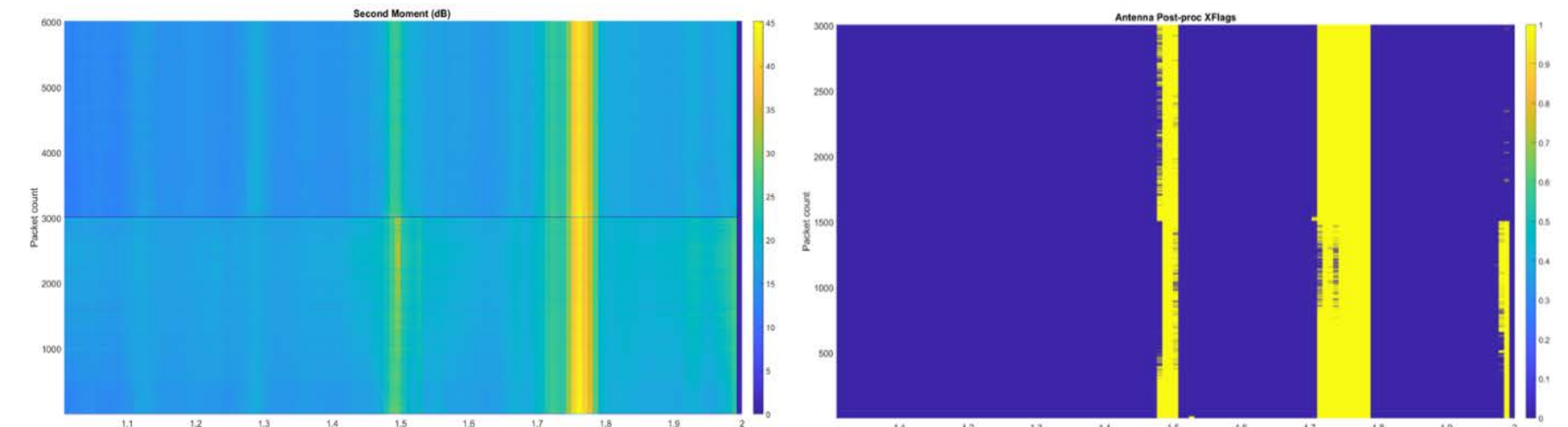


CubeRRT's first light dataset was acquired on September 5<sup>th</sup>, 2018, and demonstrates successful real-time onboard RFI flagging for the first time in space!

## CURRENT STATUS AND CONCLUSIONS

CubeRRT has successfully met its mission objectives by demonstrating the RFI processor for over 400 hours of measurements in orbit with no evidence of radiation induced degradation. This is an important milestone for the use of COTS FPGA systems for radiometer RFI processing in space.

CubeRRT's RFE ceased operation on September 8<sup>th</sup>, 2018, limiting the range of observed RDB inputs to a self generated "test pattern". Ground processing nevertheless confirms that successful on-board flag generation by the RDB continues.



The success of CubeRRT's RDB paves the way for its adoption as an essential component of future Earth observing microwave radiometers to enable continued operation in the presence of RFI.

## ACKNOWLEDGEMENTS

This work was supported by the NASA Earth Science Technology Office (ESTO). The design, development and test of the CubeRRT spacecraft bus is by Blue Canyon Technologies. The radiometer front end is developed by NASA Goddard Space Flight Center, and the radiometer digital backend is developed by NASA Jet Propulsion Laboratory.