# Status of the Signals of Opportunity Airborne Demonstrator (SoOp-AD)

#### **Purdue University**

Simulation, Retrieval Algorithms, Requirements Def.

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#### **NASA GSFC**

Systems Engineering, RF Design, Aircraft Integration

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#### Harris (Formerly Exelis, Inc)

Digital Receiver Design

George Alikakos

Co-I: Steve O'Brien

#### Langley Research Center

Aircraft Operations

**Bruce Fisher** 

#### Dr. Stephen Katzberg – Consultant

Scattering Model, Signal Processing



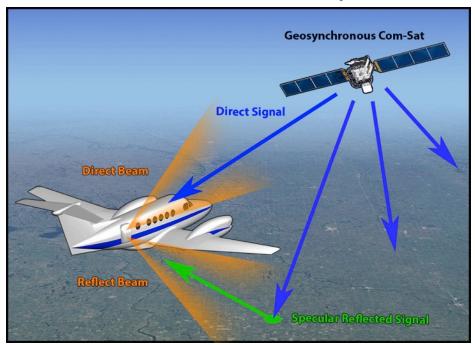






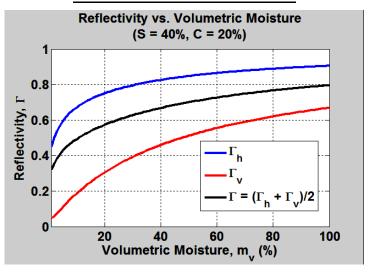
# SoOp-AD Measurement Overview

#### **P-Band Reflectometry**



We plan to measure Root Zone Soil Moisture (RZSM) through cross-correlation of direct and reflected P-Band geosynchronous communication satellite signals.

#### **Basis of Measurement**



#### **Expected Performance**

Parameter	SoOp Airborne	SoOp Spaceborne
Resolution*	100m	870m
Antenna Size	75 x 75 cm	75 x 75 cm
Sensing Depth	0-30cm	0-30cm
Sensing Precision**	0.04m <sup>3</sup> /m <sup>3</sup>	0.04m <sup>3</sup> /m <sup>3</sup>

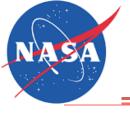
<sup>\*</sup>Specular Reflection Assumed

<sup>\*\*</sup>SMAP Requirement



# SoOp-AD Solution

- SoOp-AD will use geostationary P-Band SATCOM systems
  - 225-420MHz allocation for government use, SoOp-AD will focus on 240-270MHz band: 18 x 25-kHz channels, 20 X 5kHz channels.
  - Continuous use by US since 1978, follow-on systems planning legacy support
  - SoOp-AD method measures correlation of direct and reflected signals - does not require demod / decode of the transmission.



### **Comparison to Conventional Methods**

#### L-Band

- L-band (SMAP) penetrates only few cm of soil
- Saturation at L-band limits the ability to sense soil moisture through vegetation
- RZSM from SMAP Level 4 assimilation product

#### P-Band Radar

- Difficult to find allocation in heavily utilized spectrum
- ESA-BIOMASS cannot operate in North America or Europe due to interference with Space Object Tracking Radar
- RFI
- Expensive from space



# SoOp-AD Project Highlights

#### • IIP Timeline

- Awarded in April '14.
- System I&T at GSFC is underway.
- Science flights in Fall of '16.

#### Instrument

- Antennas: Patch, Dual Linear Pol, Null Steering
- Receivers: Standard P-Band Receivers w/ internal calibration. S-Band receiver for XM Radio included
- Digital System: FPGA based. 7TB Storage for raw and/or correlation data
- Two aircraft racks: 12U Total

#### Aircraft Campaign

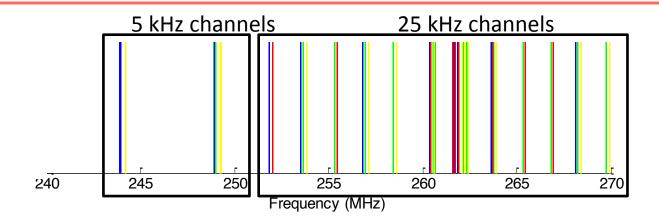
- Flying on NASA Langley B200.
- Co-Flying with SLAP instrument (GSFC's Active / Passive L-Band).
- Science flights over the St. Joseph's Watershed.



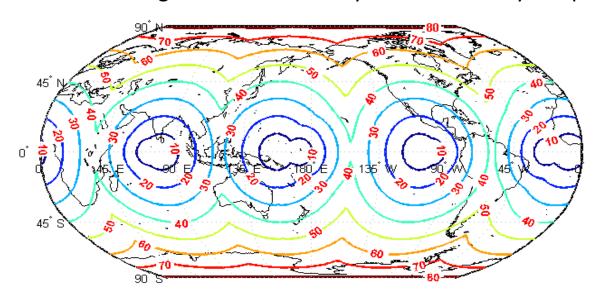




# Signal Bands and Coverage

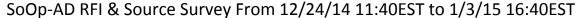


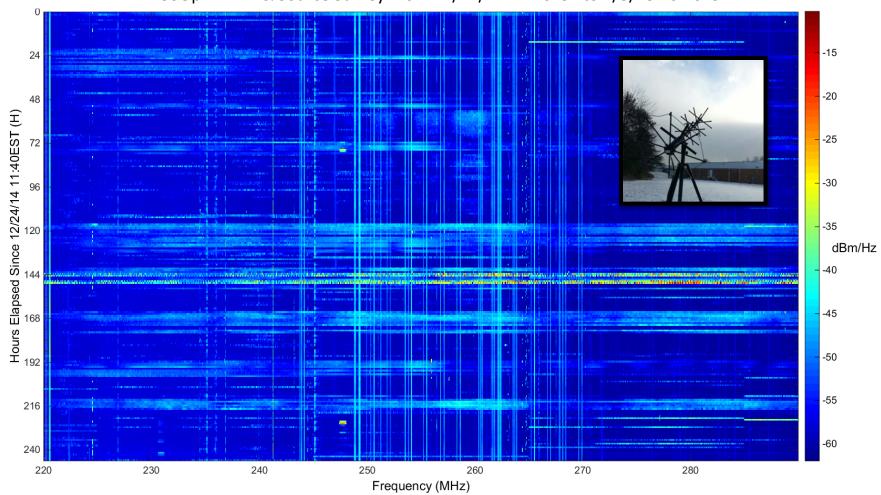
Incidence Angle for Geostationary Sources used by SoOp-AD.





# Measured Signal Details & RFI



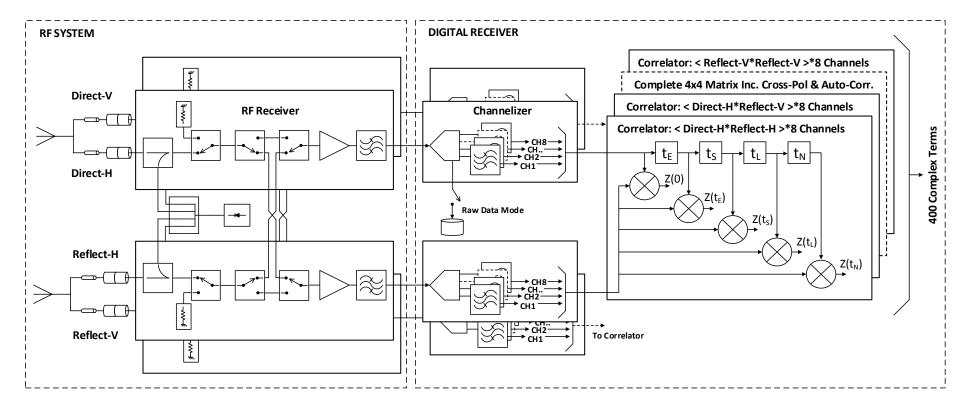


Waterfall spectrum measured at GSFC over 11 days. Note persistence of SATCOM signals and broad-band RFI.



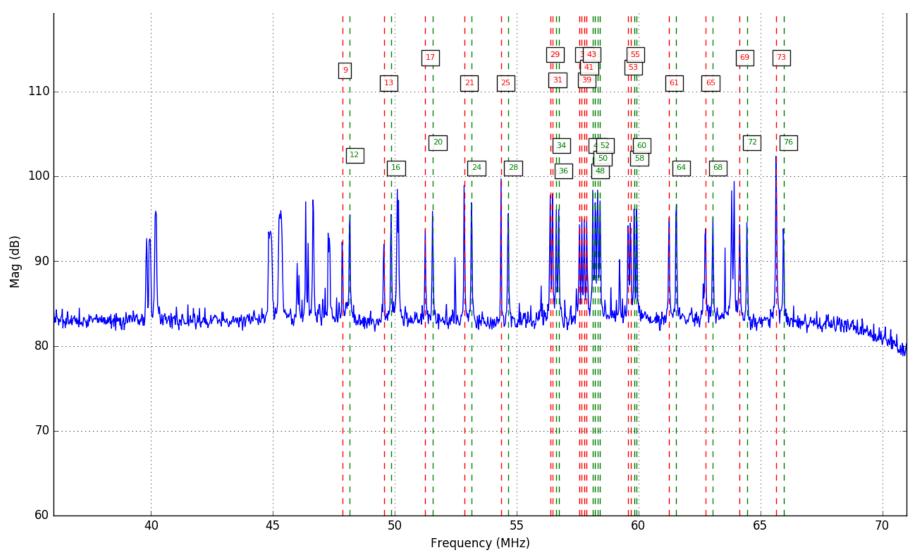
# SoOp-AD System Architecture

Reflectometry: 
$$\Gamma_2 = (\frac{|\tilde{Z}_{12}(\tau_{RD}|)}{\tilde{Z}_{11}(0) - G_1 \sigma_1^2})^2 \frac{G_1}{G_2} \frac{G_{S,D}}{G_{E,R}}$$



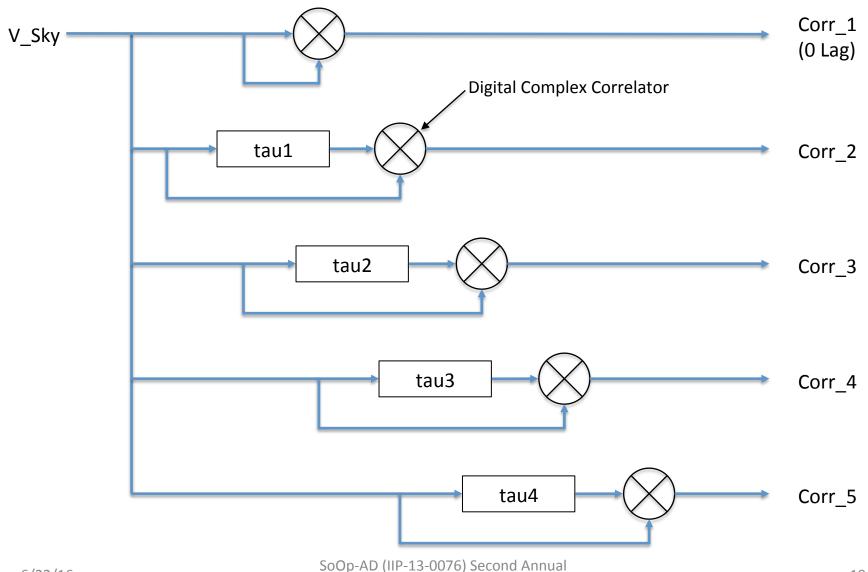


### Spectrum from SoOp-AD Raw Data Mode





# "Auto" Example: (V\_Sky, V\_Sky\*)

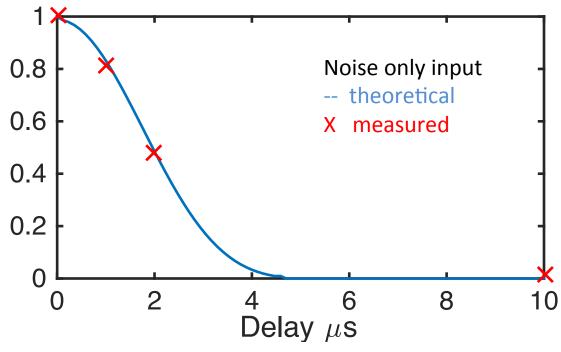


Review



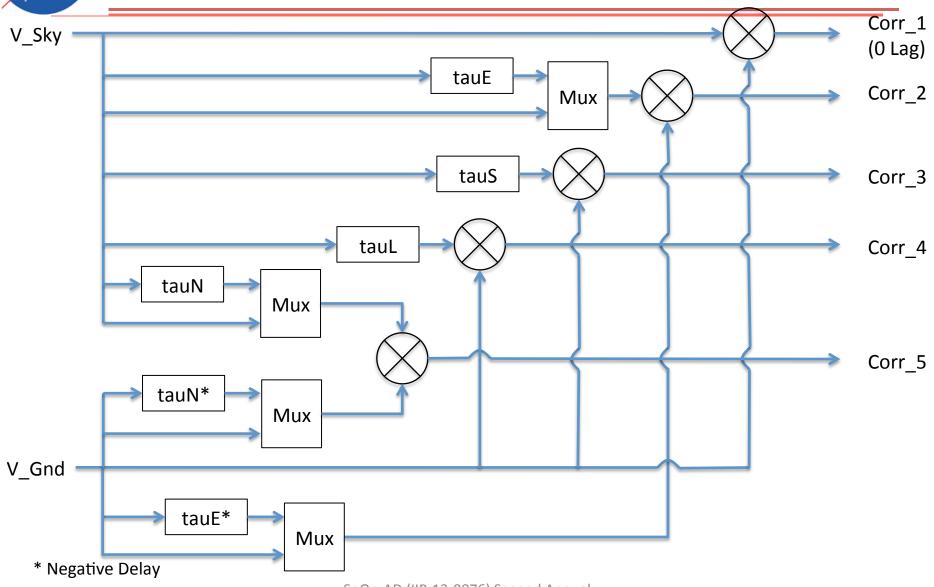
### SoOp-AD Auto-Correlator Verification

- Correlators have programmable 4 lags + 0
- 300-kHz noise detection bandwidth
- Test: 0, 1, 2 and 10 us (400 us not shown)





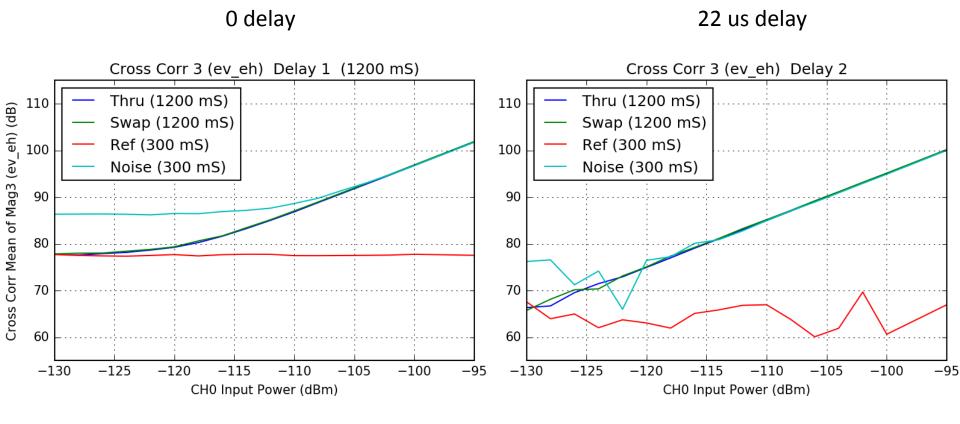
# "Cross" Example: (V\_Sky, V\_Gnd\*)





# SoOp-AD Cross-Correlator Verification

### AWG QPSK waveform into V&H inputs





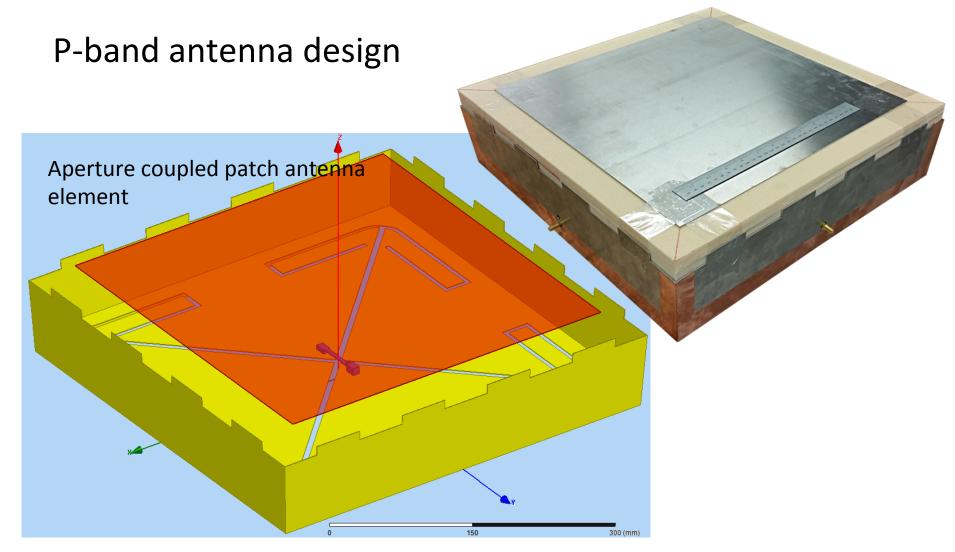
### **Technology Development: Antennas**

Antenna radome design for B200 aircraft





### **Technology Development: Antennas**



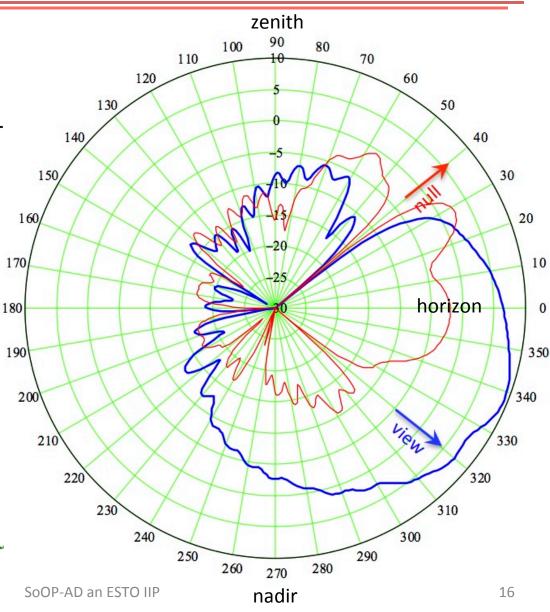


# **Antenna System Considerations**

- Direct-to-Reflect isolation is driving requirement – But not in orbit!
- Using two-element interferometer to synthesize a two-element array with null steering in postprocessing.
- Simulation: Earth View Beam
  - Co-pol (blue): LHCP
  - X-pol (red): RHCP

ESTF June 14-16, 2016

 Results simulate a post-processed 190/ pattern with a null steered to +40°





### Technology Development: Antenna Radome

- Radome designed and fabricated.
- Test-fit Successful.
- Awaiting test flight







### **Next Steps**

- Ground Testing
- Aircraft Safety Test
- Aircraft Campaign in Fall of 2016



