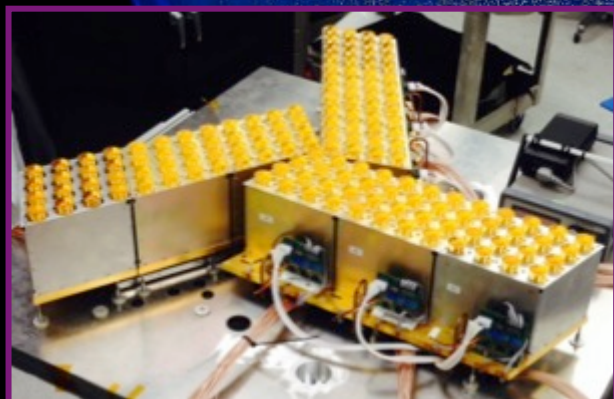




# GeoSTAR Results

Not quite final yet...

Bjorn Lambrigtsen, Todd Gaier, Alan Tanner, Pekka Kangaslahti\*  
Chris Ruf, Darren McKague, Mike Flynn, Roger Backhus†



\* Jet Propulsion Laboratory  
California Institute of Technology  
† University of Michigan

Earth Science Technology Forum  
Pasadena; June 23-25, 2015



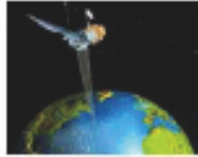


# GeoSTAR Development History

## Decadal survey: PATH mission

Decadal Survey Mission	Mission Description	Orbit	Instrument	Rough Cost Estimate
<b>Timeframe: 2010 – 2013. Missions listed by cost</b>				
CLARREO (NASA portion)	Solar radiation: spectrally resolved forcing and response of the climate system	LEO, Precessing	Absolute, spectrally-resolved interferometer	\$200 M
SMAP	Soil moisture and freeze-thaw for weather and water cycle processes	LEO, SSO	L-band radar L-band radiometer	\$300 M
ICESat-II	Ice sheet height changes for climate change diagnosis	LEO, Non-SSO	Laser altimeter	\$300 M
DESDynI	Surface and ice sheet deformation for understanding natural hazards and climate, vegetation structure for ecosystem health	LEO, SSO	L-band InSAR Laser altimeter	\$700 M
<b>Timeframe: 2013 – 2016. Missions listed by cost</b>				
HypIRI	Land surface composition for agriculture and mineral characterization, vegetation types for ecosystem health	LEO, SSO	Hyperspectral spectrometer	\$300 M
ASCENDS	Day/night, all-latitude, all-season CO <sub>2</sub> column intensity for climate emissions	LEO, SSO	Multifrequency laser	\$400 M
SWOT	Ocean, lake, and river water levels for ocean and inland water dynamics	LEO, SSO	Ka-band wide swath radar C-band radar	\$450 M
GEO-CAPE	Atmospheric gas columns for air quality forecasts, ocean color for coastal ecosystem health and climate emissions	GEO	High and low spatial resolution hyperspectral imagers	\$350 M
ACE	Aerosol and cloud profiles for climate and water cycle, ocean color for open ocean biogeochemistry	LEO, SSO	Backscatter lidar Multiple polarimeter Doppler radar	\$800 M
<b>Timeframe: 2016-2020. Missions listed by cost</b>				
LIST	Land surface topography for landslide hazards and water runoff	LEO, SSO	Laser altimeter	\$300 M
PATH	High frequency, all-weather temperature and humidity soundings for weather forecasting and SST <sup>a</sup>	GEO	MW array spectrometer	\$450 M
GRACE-II	High temporal resolution gravity fields for tracking large-scale water movement	LEO, SSO	Microwave or laser ranging system	\$450 M
SCLP	Snow accumulation for fresh water availability	LEO, SSO	Ku and X-band radars K and Ka-band radiometers	\$500 M
GACM	Ozone and related gases for intercontinental air quality and stratospheric ozone layer prediction	LEO, SSO	UV spectrometer IR spectrometer Microwave limb sounder	\$600 M
3D-Winds (Demo)	Tropospheric winds for weather forecasting and pollution transport	LEO, SSO	Doppler lidar	\$650 M

Precipitation and All-weather Temperature and Humidity (PATH)  
 Launch: 2016-2020  
 Mission Size: Medium



Sea surface temperature

Temperature and humidity profiles

Constraints on models for boundary layer, cloud, and precipitation processes

More accurate, longer-term weather forecasts

Improved storm track and intensification prediction and evacuation planning

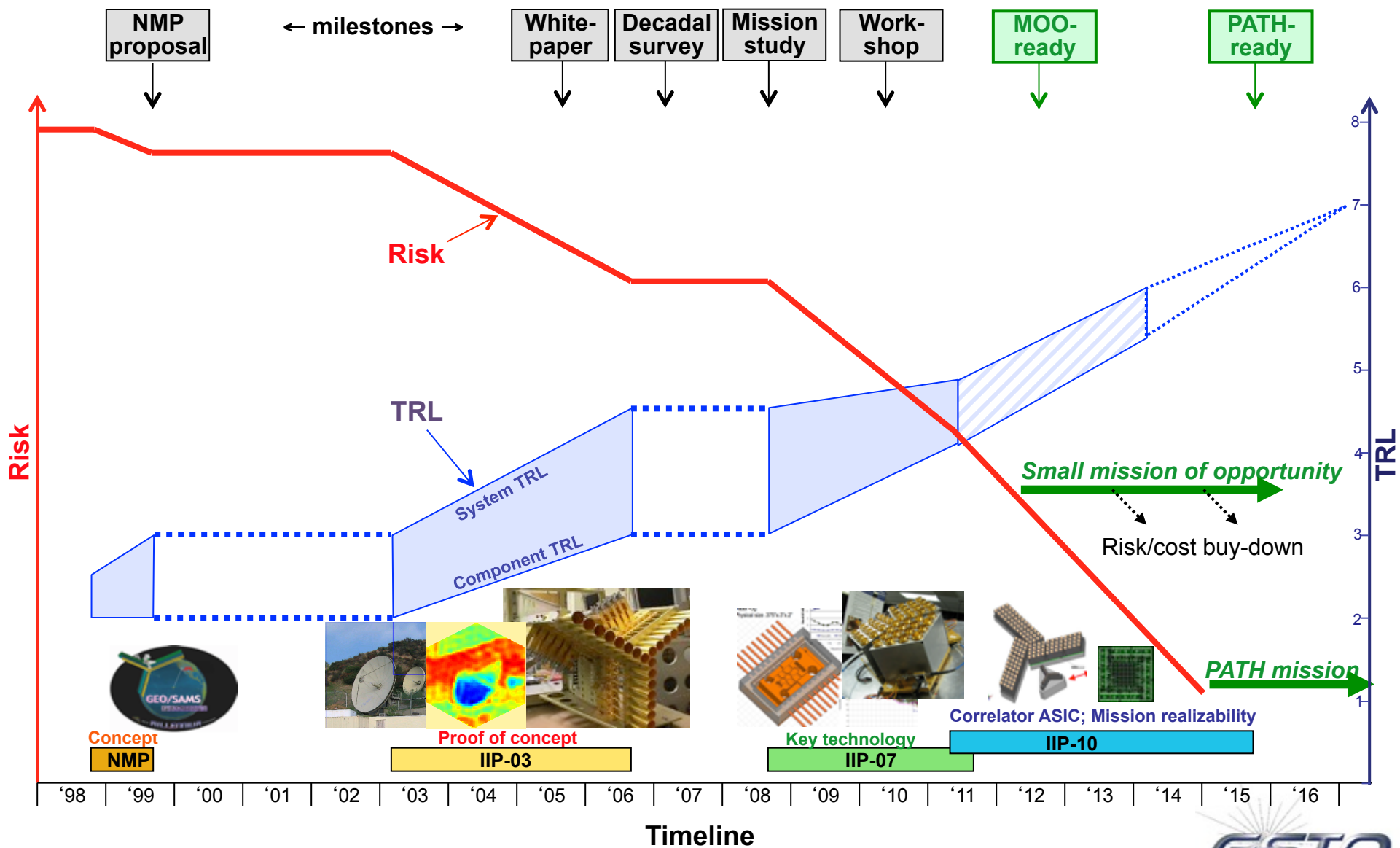
Determination of geographic distribution and magnitude of storm surge and rain accumulation

PATH	High frequency, all-weather temperature and humidity soundings for weather forecasting and SST <sup>a</sup>	GEO	MW array spectrometer = GeoSTAR	\$450 M
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# GeoSTAR timeline

- **Concept development**
  - NMP/EO-3 proposal (1998-1999)
  - NRC white paper (2005)
  - NRC Decadal Survey (2007)
- **ESTO technology development**
  1. IIP-03 (2003-2006): Proof-of-concept prototype
  2. ACT-05 (2006-2008): MIMRAM receivers
  3. IIP-07 (2008-2011): Key technology
  4. IIP-10 (2011-2015): Risk reduction
- **Space implementation**
  - *Venture mission (2020?)*
  - *PATH mission (~202X?)*

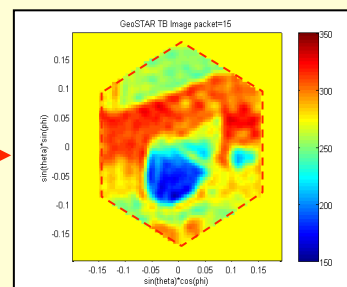
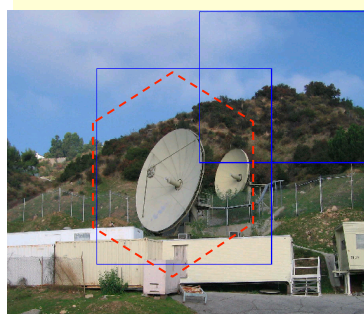
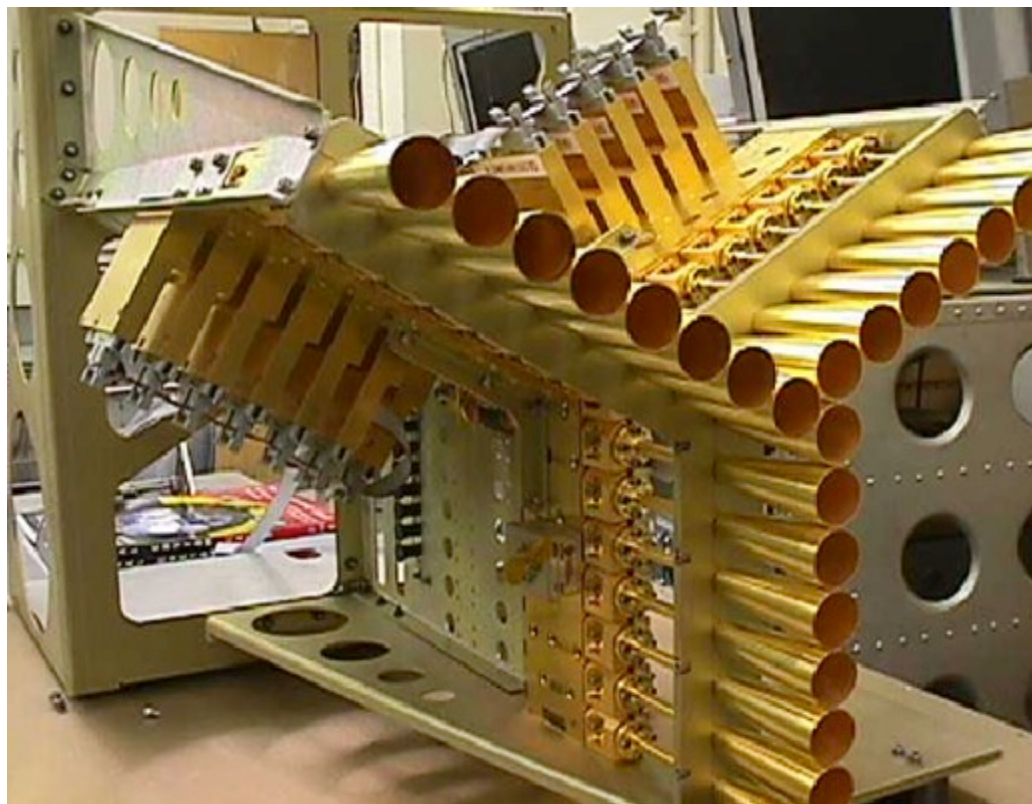
# GeoSTAR timeline





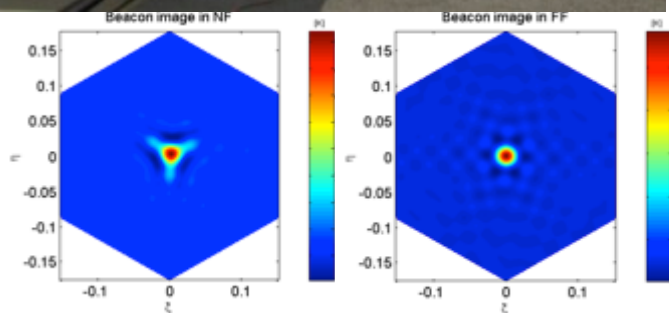
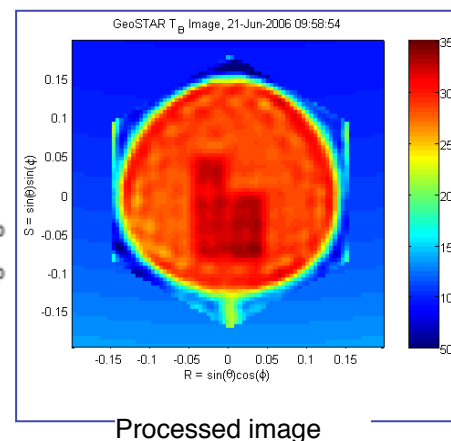
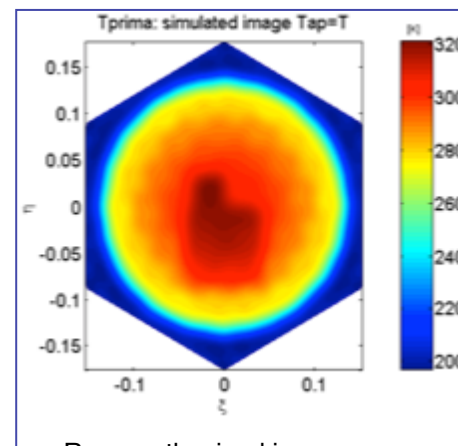
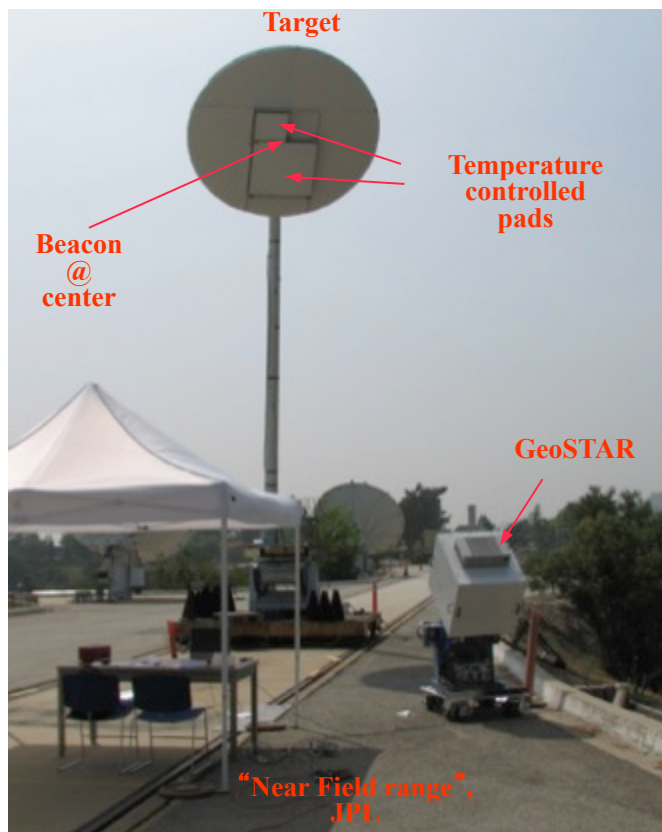
National Aeronautics and  
Space Administration  
Jet Propulsion Laboratory  
California Institute of Technology  
Pasadena, California

# IIP-03: Proof of concept





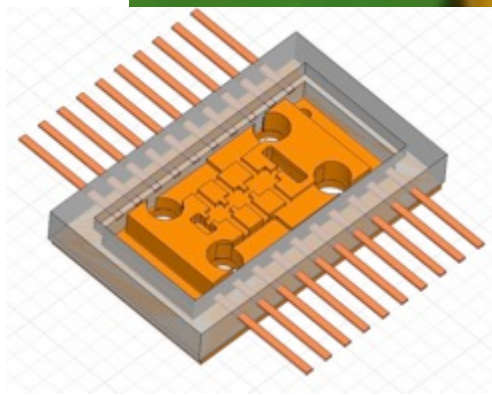
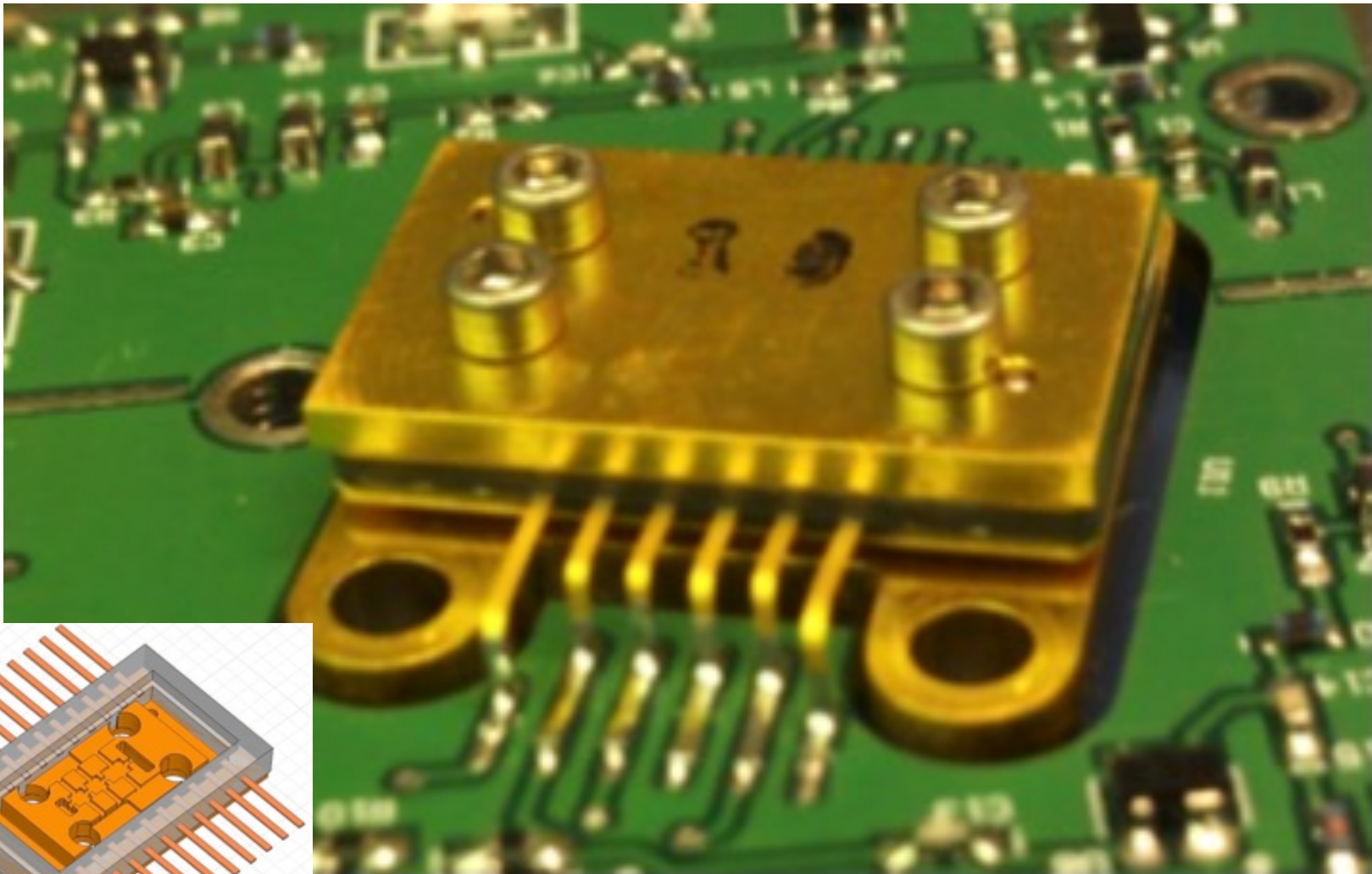
# JPL R&TD: Calibration





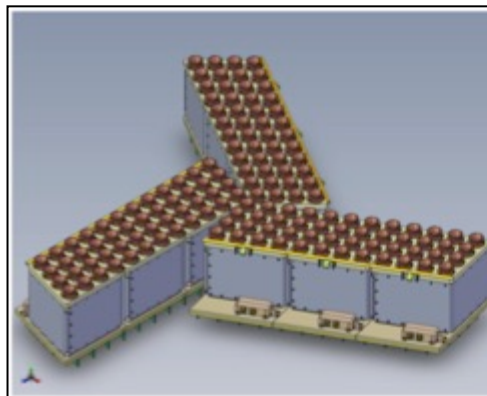
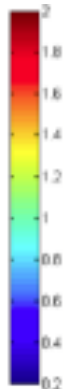
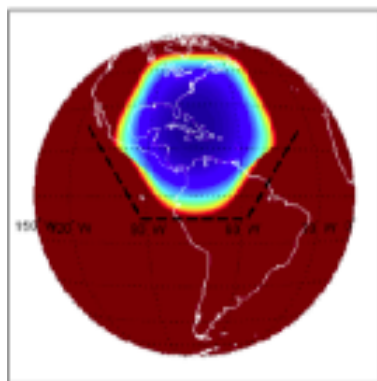
National Aeronautics and  
Space Administration  
Jet Propulsion Laboratory  
California Institute of Technology  
Pasadena, California

# ACT-05: MIMRAM

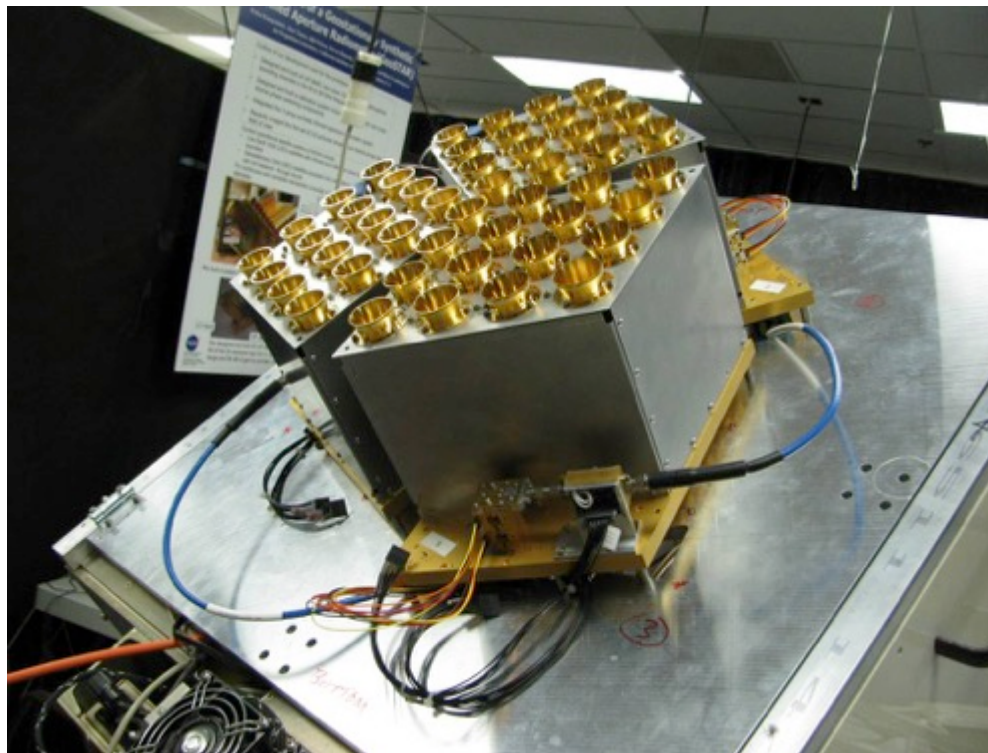




# IIP-07: Architecture & technology



- Develop 50 low-noise 183-GHz receivers
- Develop 3 2x8-element receiver sub-array modules
- Develop low-power Application-Specific Integrated Circuit (ASIC) correlator chips
- Develop low-mass/power signal distribution system
- Develop functional 183-GHz 2D STAR prototype





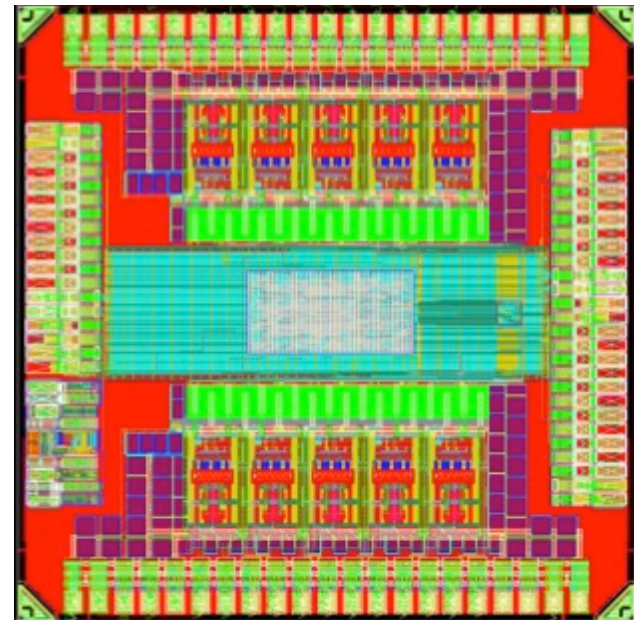
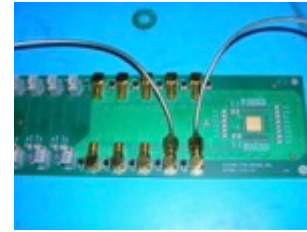
# IIP-10: Correlator – 5x5 test chip

A 5x5 digitizer/correlator and evaluation board was built to provide risk reduction for the development of the larger A/D correlator ASIC.

- Test A/D and correlator cells together to uncover design or implementation flaws
- Determine crosstalk between channels

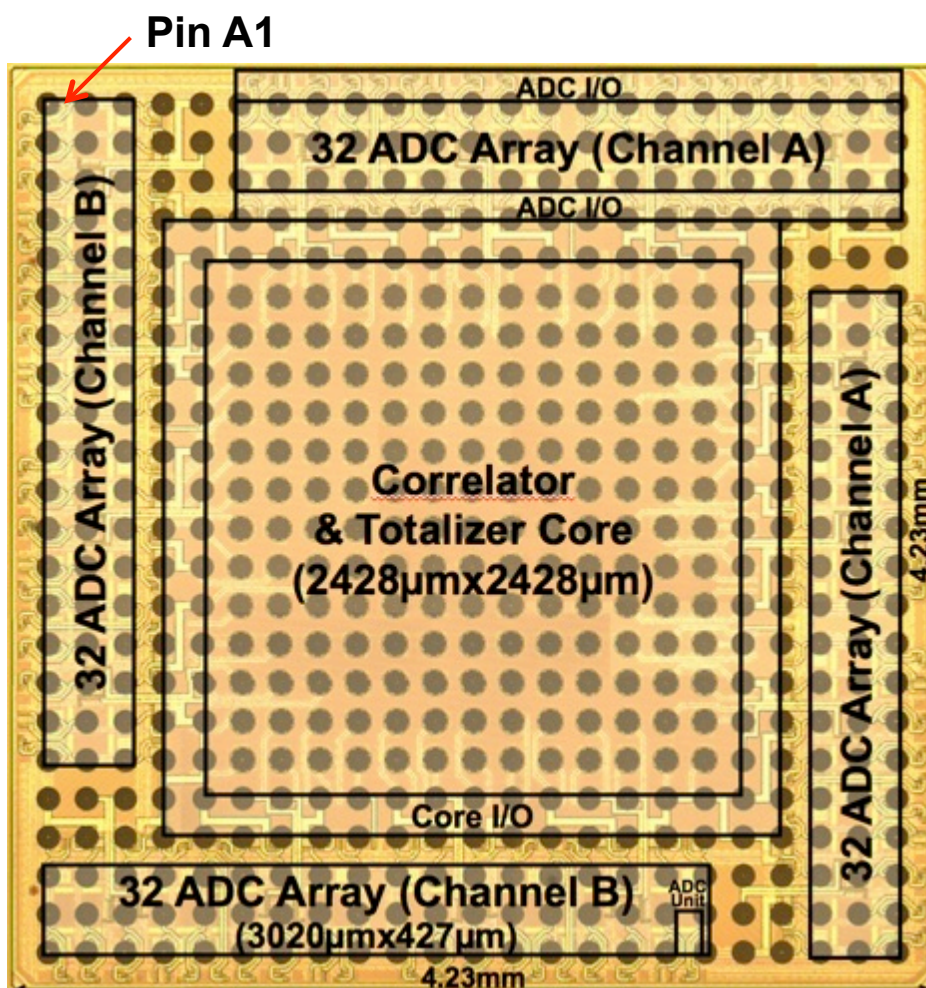
Initial tests indicated problems, but design was fixed, chip respun and tested

Tested for rad-hardness: OK

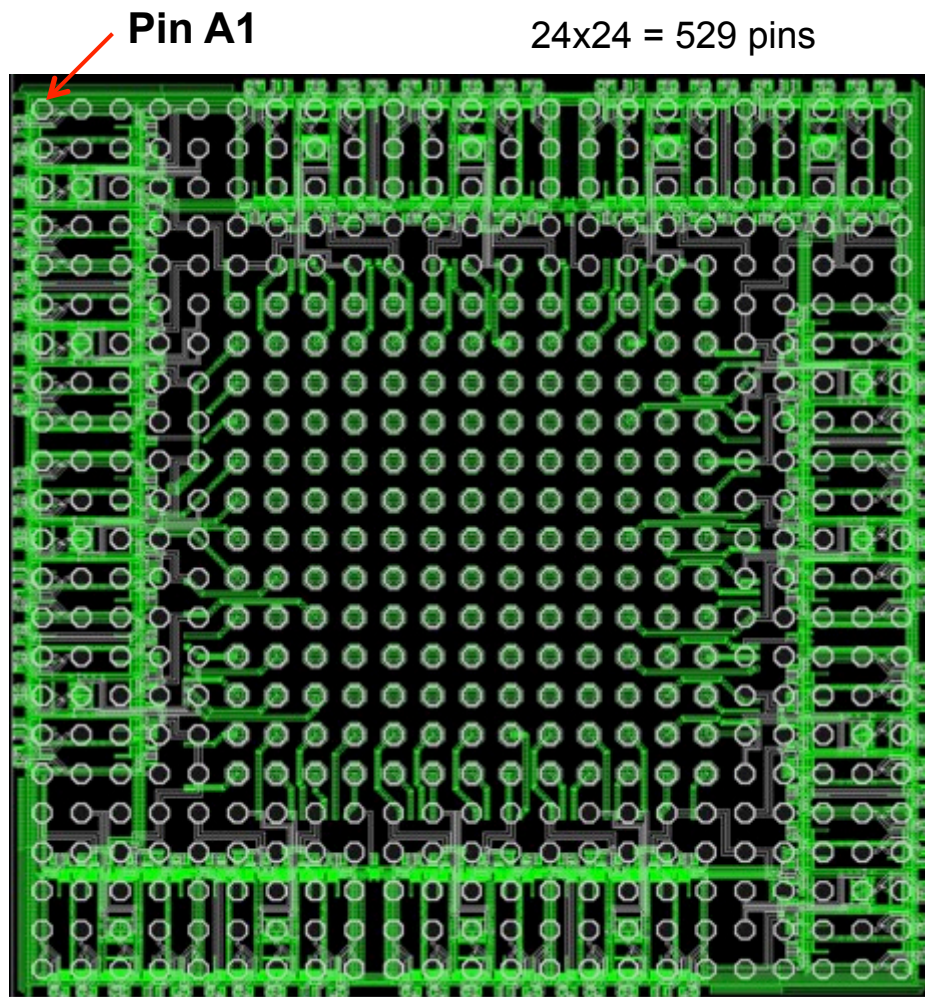


# IIP-10: Correlator 64x64 chip

## Chip orientation



- Die Photo -



- Layout (M9, AP) -



# Correlator Test Board

Test system allowed only four independent inputs and on-board switches and dividers to excite all ASIC inputs

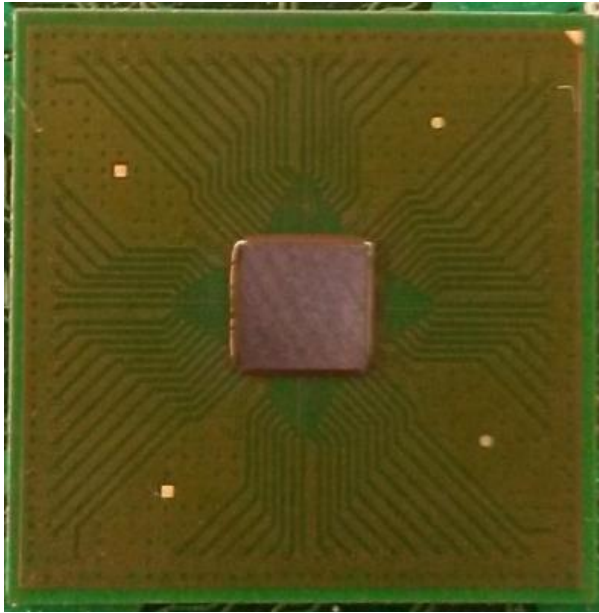


Figure S2: The correlator die is flip chip bonded to a custom-designed 576 pin 8-layer substrate.

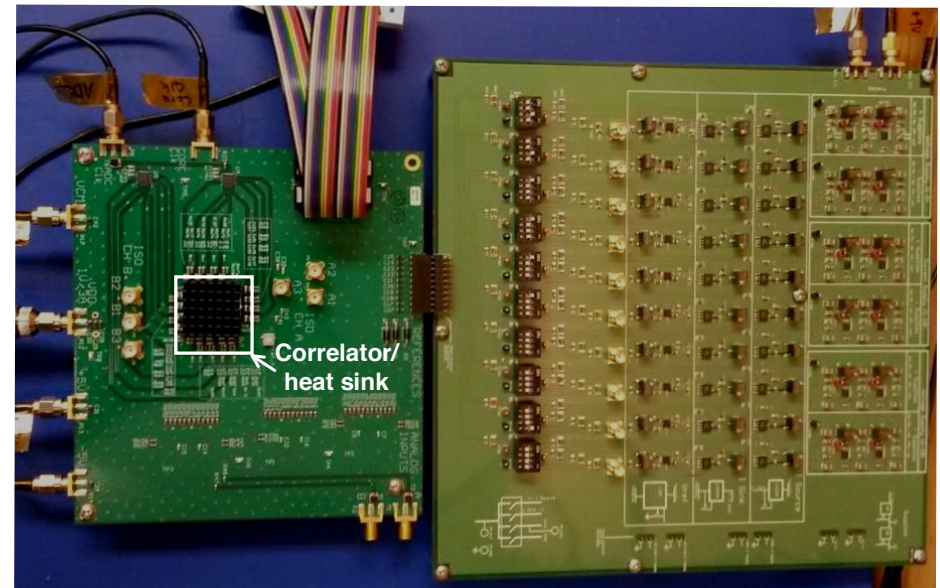
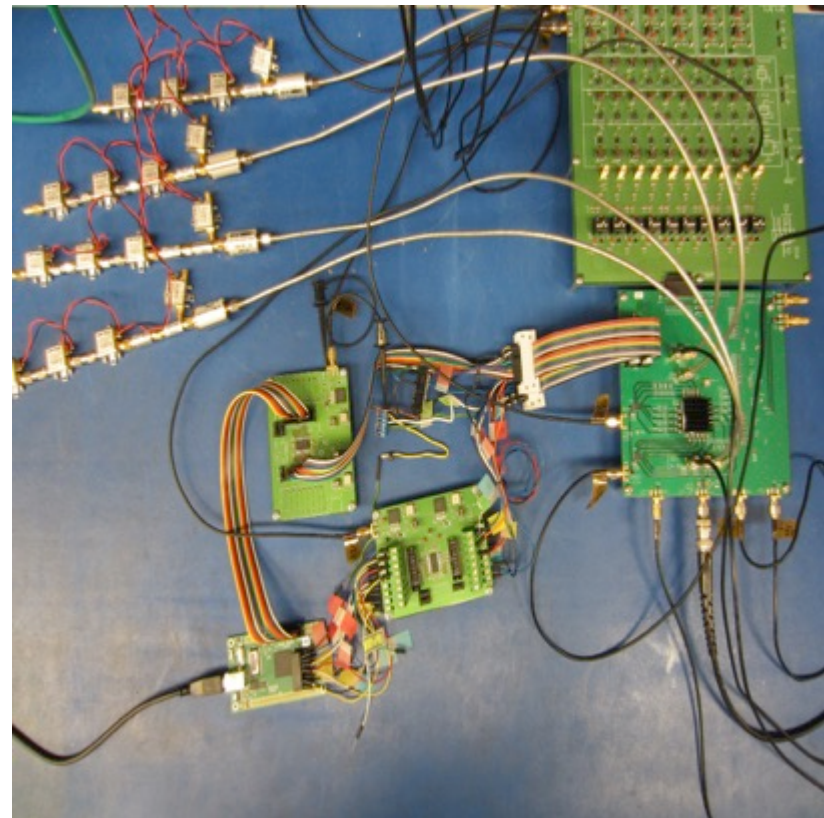


Figure S3: Test circuit board and test setup.



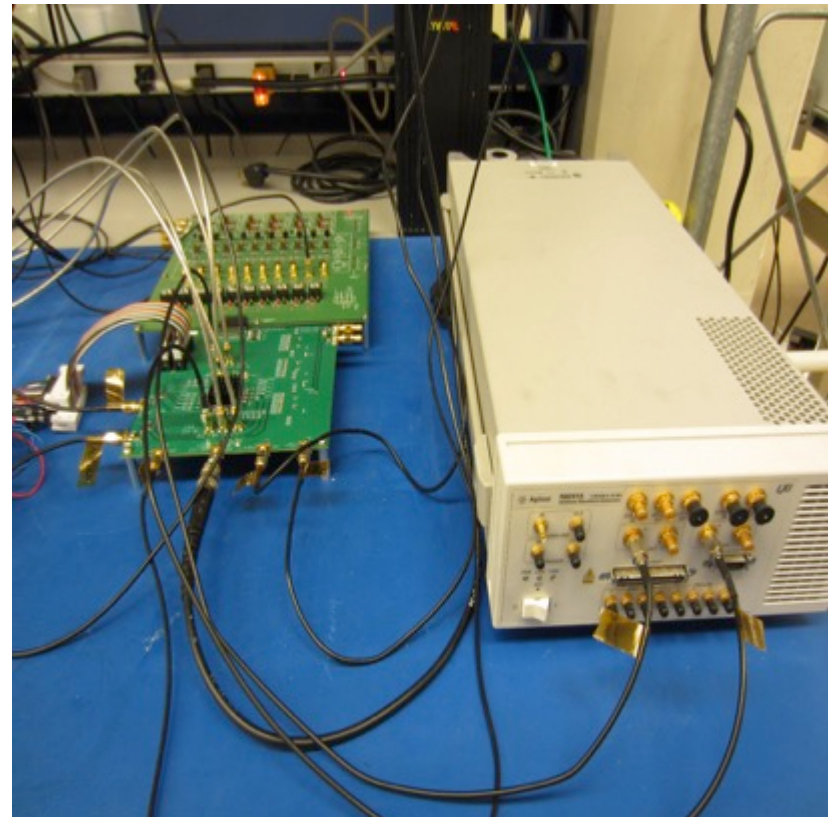
# General Test Configuration

- Test board designed by EECS
  - Hard-wired input ports for a 3x3 subarray of adjacent channels
  - A13-A15, B13-B15
- Inputs
  - 2-channel AWG on one A-side, one B-side input, drives the correlator cell of interest
  - Adjacent inputs driven by independent amplified Johnson noise (uncorrelated white noise)
- Nominal clock: 1 GHz (sig gen max of 990 MHz)
- Nominal integration time: 10 ms



# Correlation Efficiency

- 2-channel AWG sends 100% correlated or anti-correlated white noise signals to two ASIC input channels with input power of -25 dBm
- Resulting digital correlations are mapped to analog using totalizers
- Ratio between the analog correlation obtained and the analog correlation of input signal (here, unity) is the correlation efficiency.
  - Observed to be between 97% and 98% for the nine cells tested (averaged in 25 repeated samples)



# IIP-10: ASIC preliminary test results

Chips assembled and tested on EECS test board

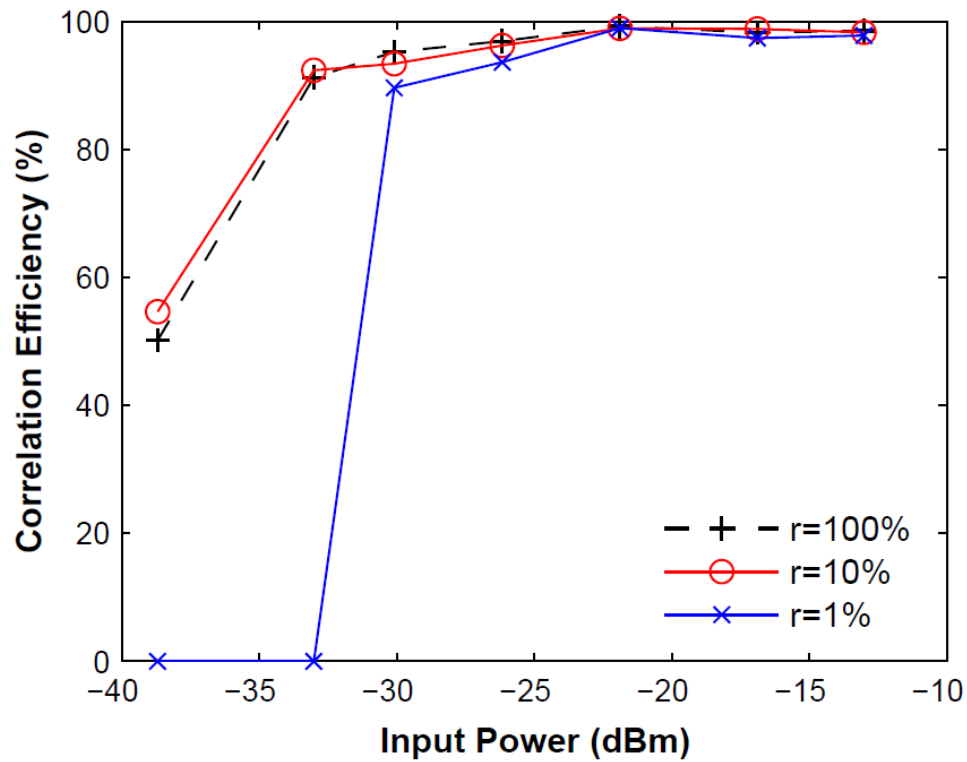
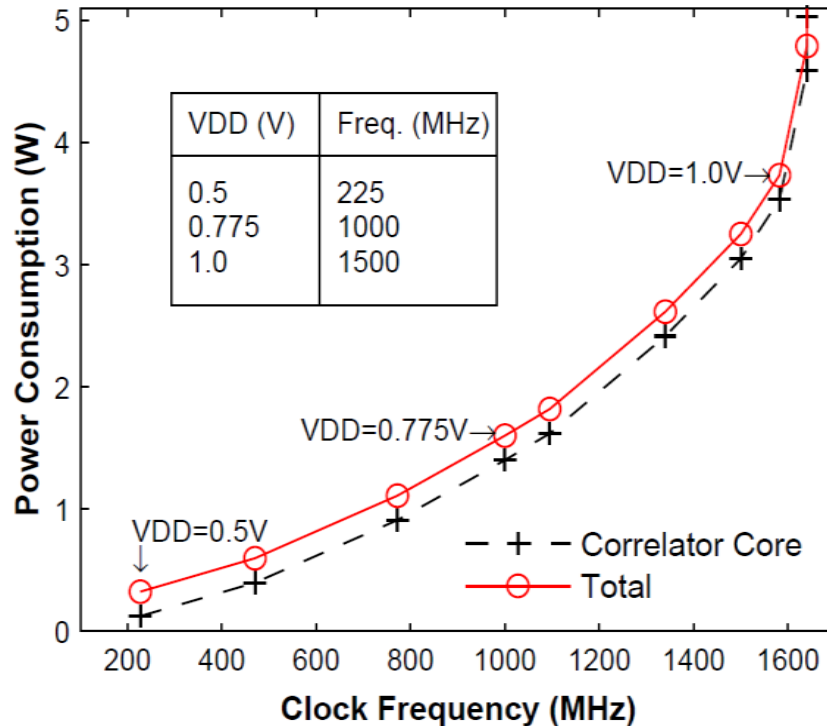


Figure 5: Correlation efficiency of the test chip. ( $r = 100\%$  represents when the two channels receive 100% correlated inputs;  $r = 10\%$  represents the two channels receive 10% correlated inputs and so on.)





# ASIC Correlator Measured Performance

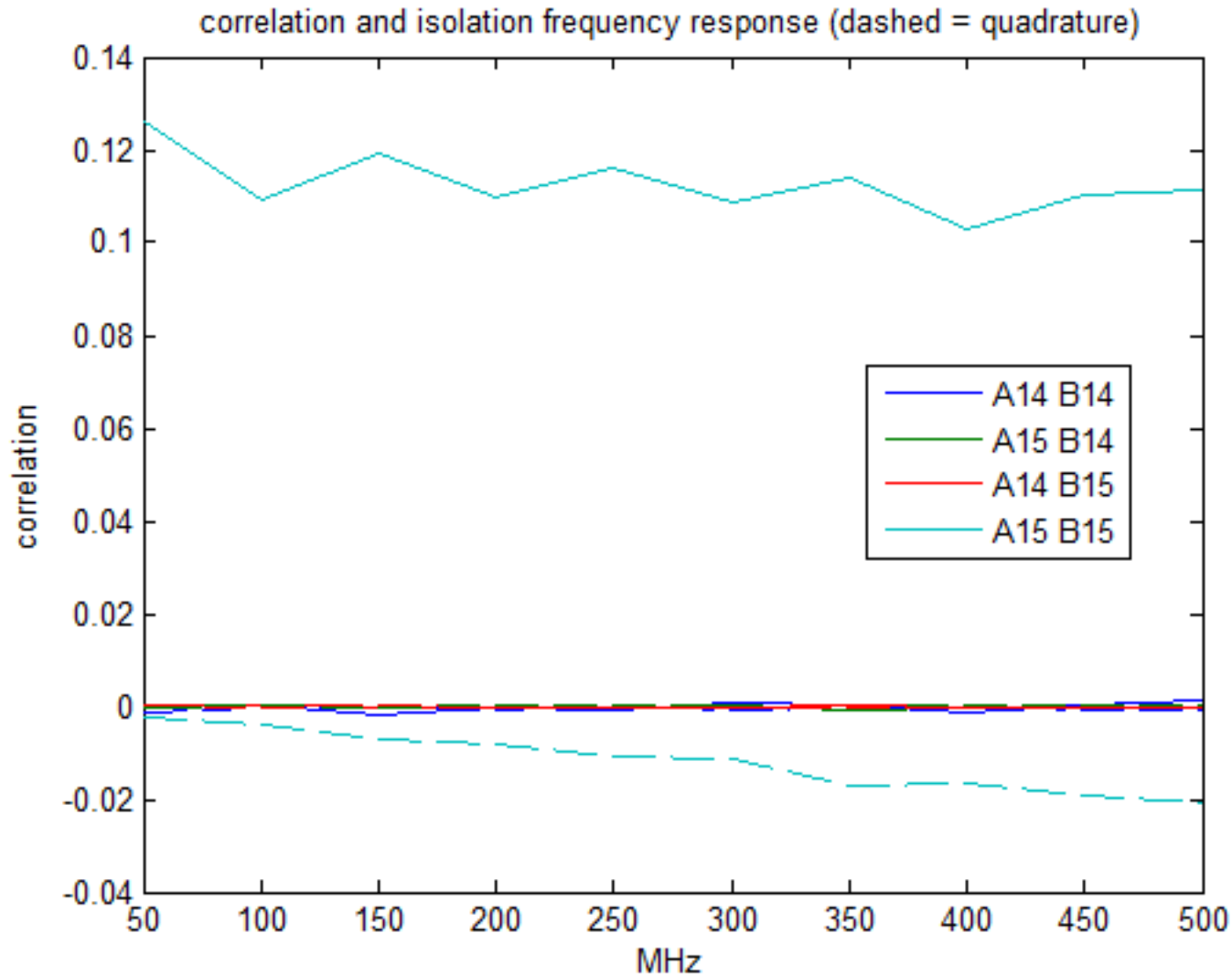


	This work
# of Channels	128
# of Correlators	4096
Channel Bitwidth	2
On-Chip ADCs	yes, 128
Logic Family	static
Correlation Efficiency (%)	>90% @ >30dBm
Isolation (dB)	-42.4
Technology	65nm
Total Power (W)	1.44 @775mV, 1GHz 3.73@1V, 1.5GHz
Energy per Correlation (pJ/correlation/cycle)	0.35 @775mV, 1GHz 0.61 @1V, 1.5GHz (2b corr + ADC)
Core Area (mm <sup>2</sup> )	5.9
Chip Area (mm <sup>2</sup> )	17.9
Max Performance (T correlation/s)	6.14 @1V, 1.5GHz

<sup>a</sup>: a 1-bit correlation is just XOR



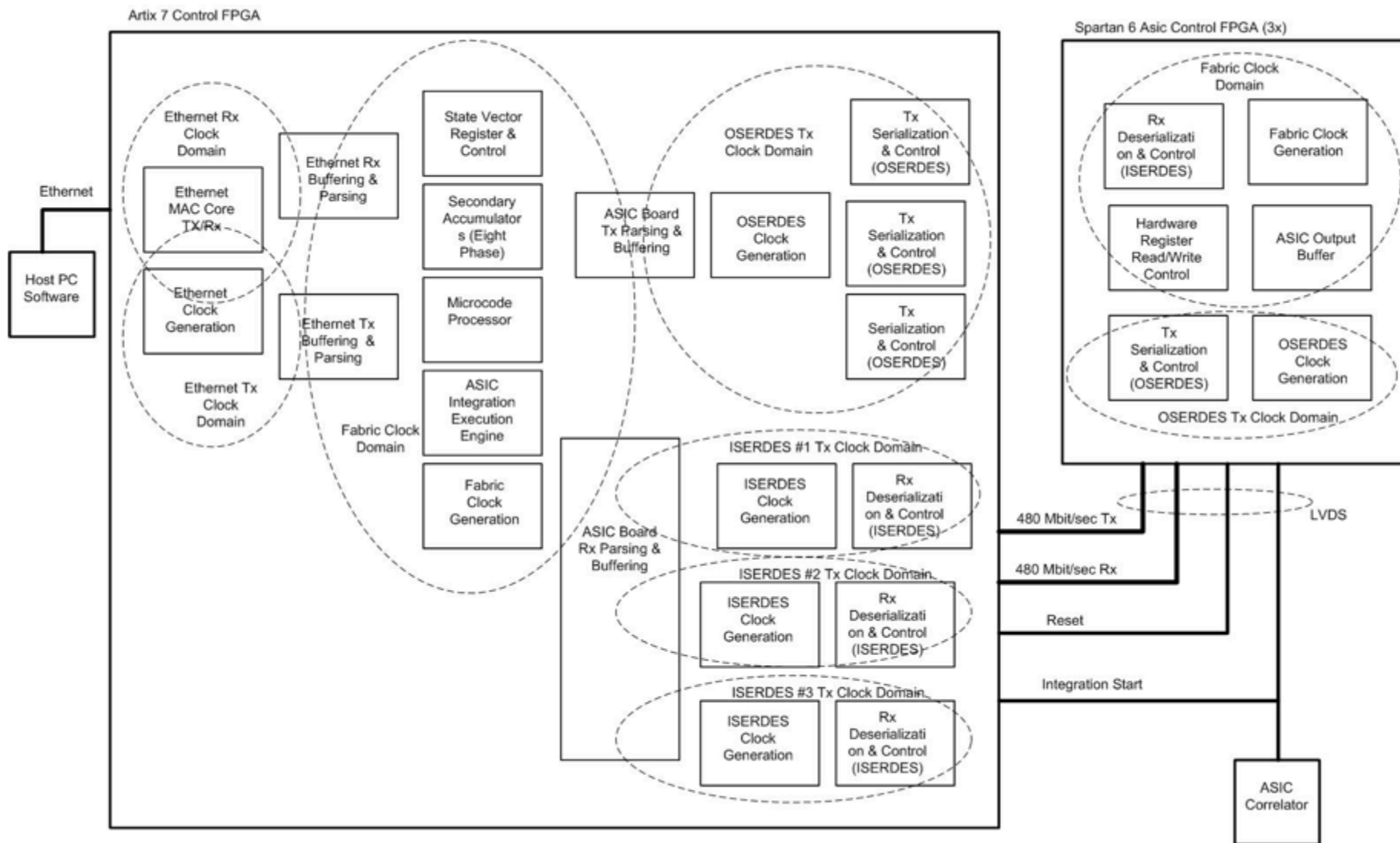
# Correlator ASIC Frequency Response



Note: quadrature channel should be 0, but has small leakage. That is in the test set .



# IIP-10: ASIC full system test boards







National Aeronautics and  
Space Administration

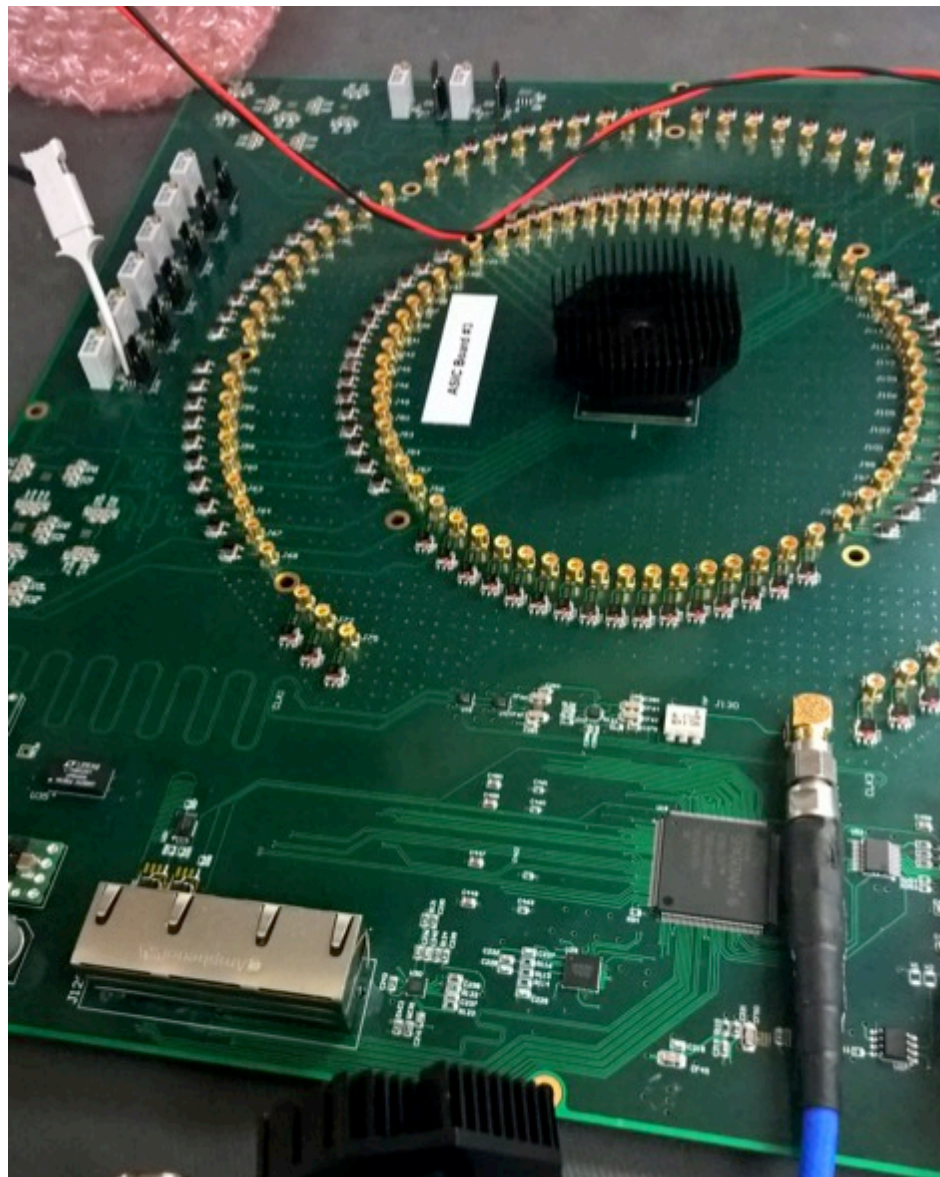
Jet Propulsion Laboratory  
California Institute of Technology  
Pasadena, California

# IIP-10: ASIC full system test boards





# IIP-10: ASIC system test board

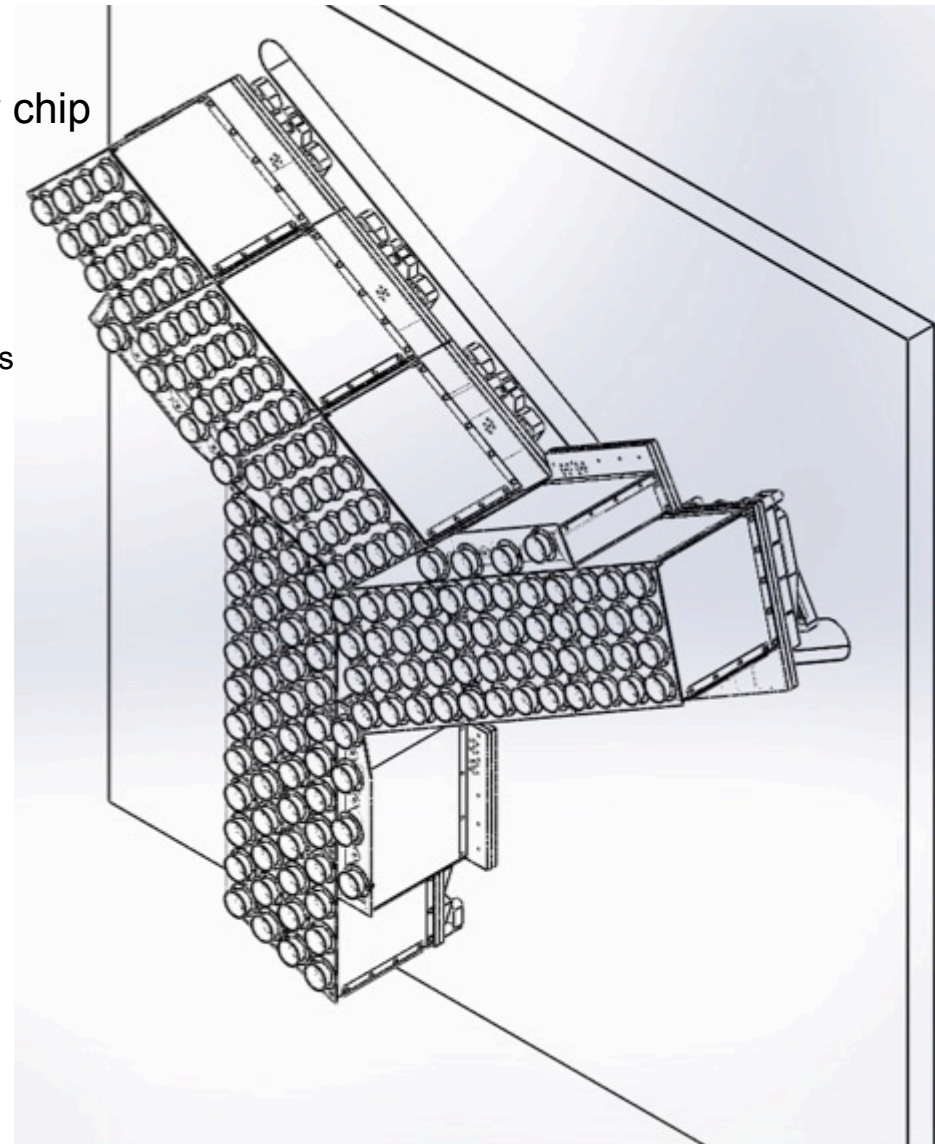
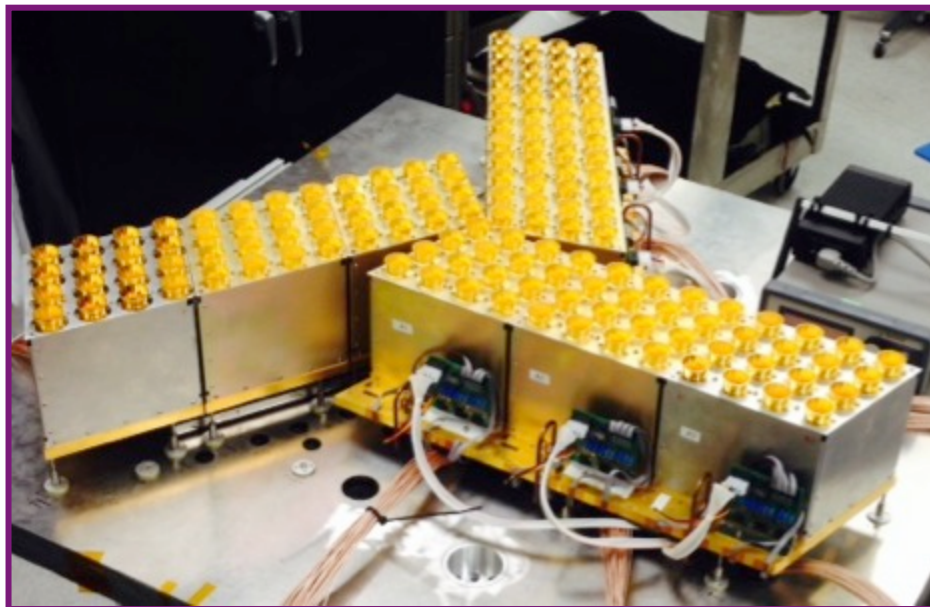




# IIP-10: System testing

## Awaiting completion of correlator test board

- Further testing of correlator ASIC
  - Complete full correlator board hosting new chip
  - Full functional testing of new chip
- System testing
  - Assemble small 183-GHz antenna array
    - Using miniature ultra-low-power MIMRAM receivers
  - Integrate full system with correlator
  - Characterize system performance
  - Imaging demonstration

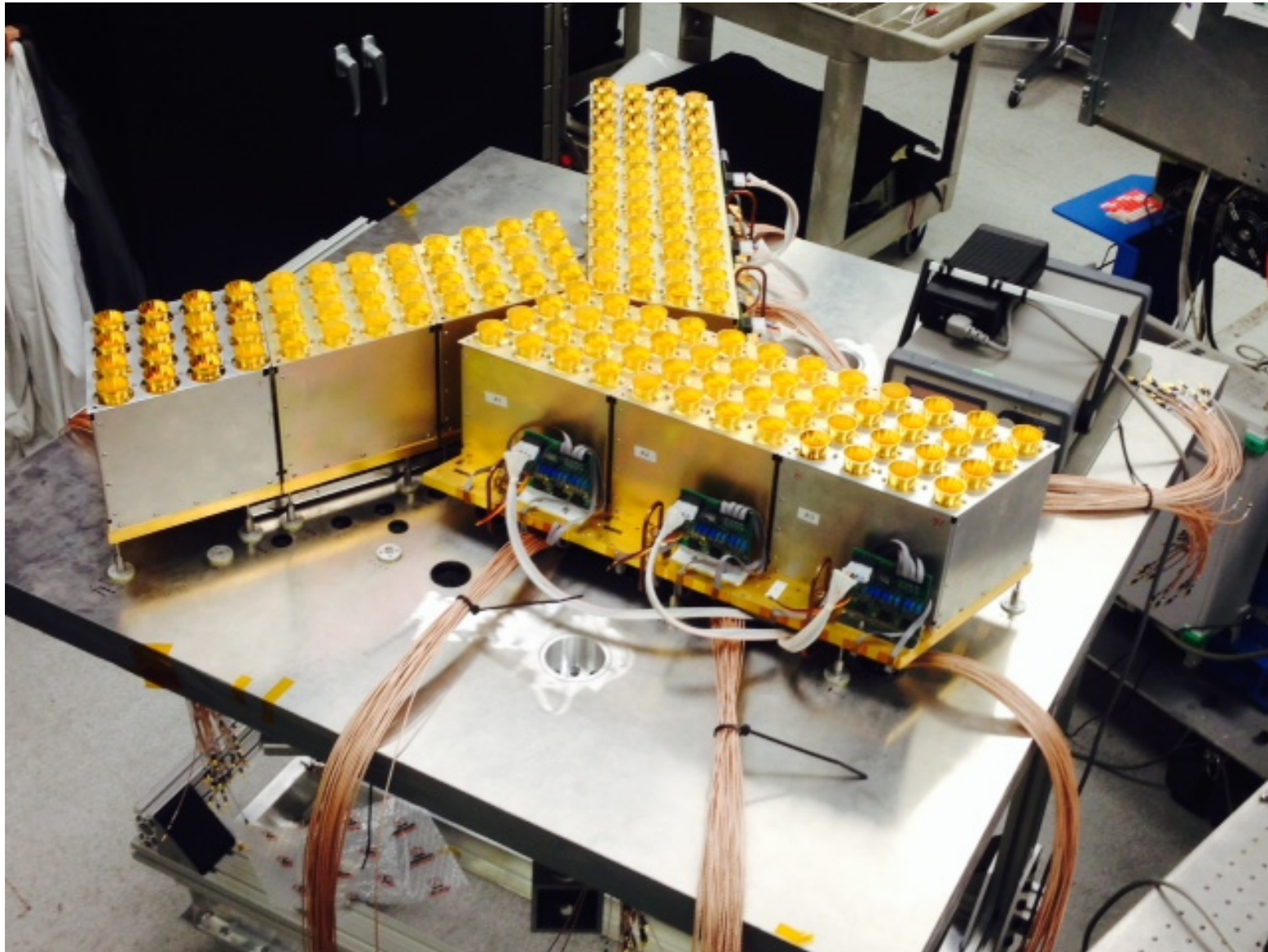






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Pasadena, California

# System Integration





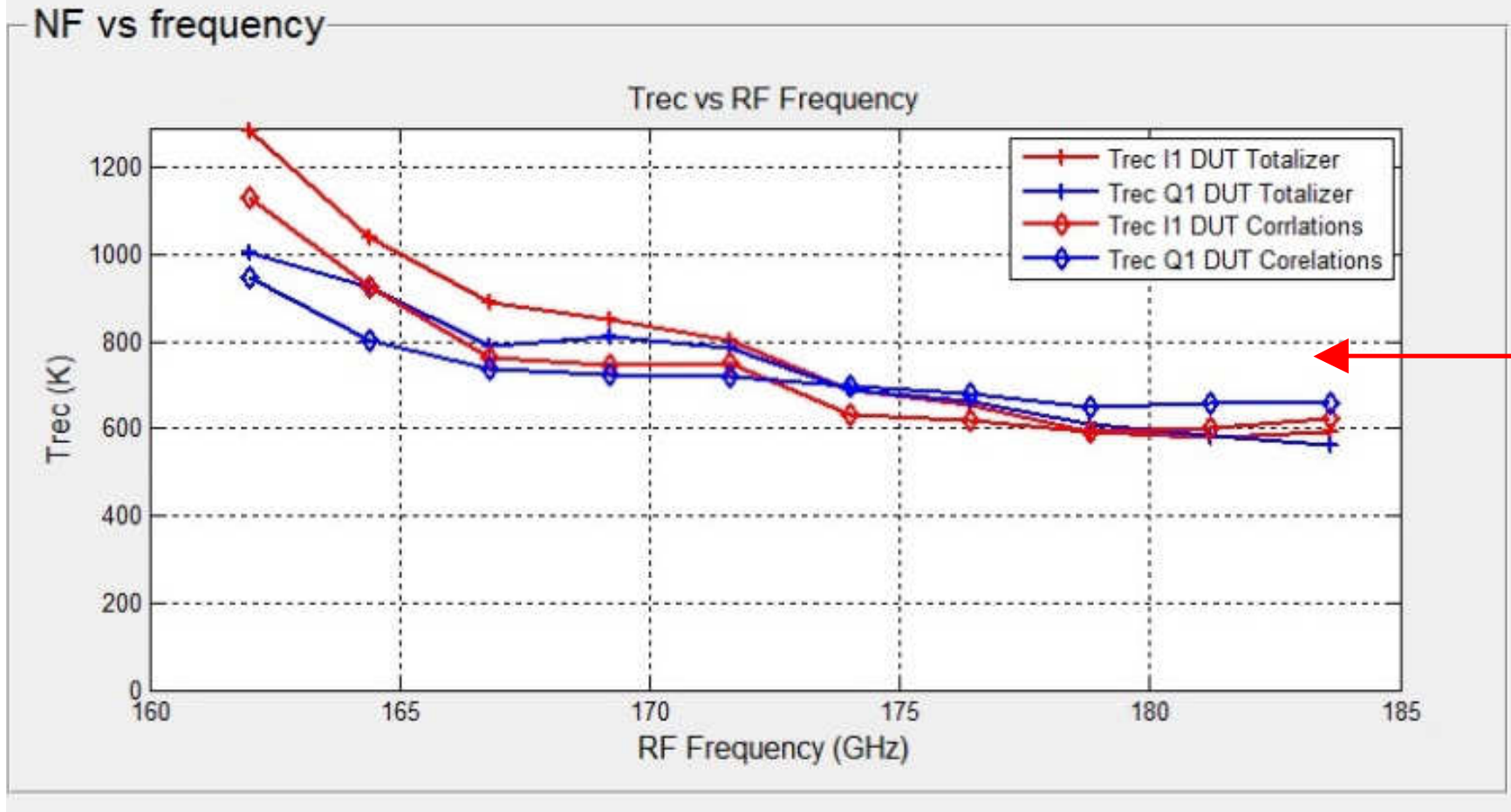
# System Integration



- Table assembled
- Tiles assembled
- Power supplies integrated
- LO integrated
- IF cables installed
- Synthesizer Installed

# 135 Receivers Populated

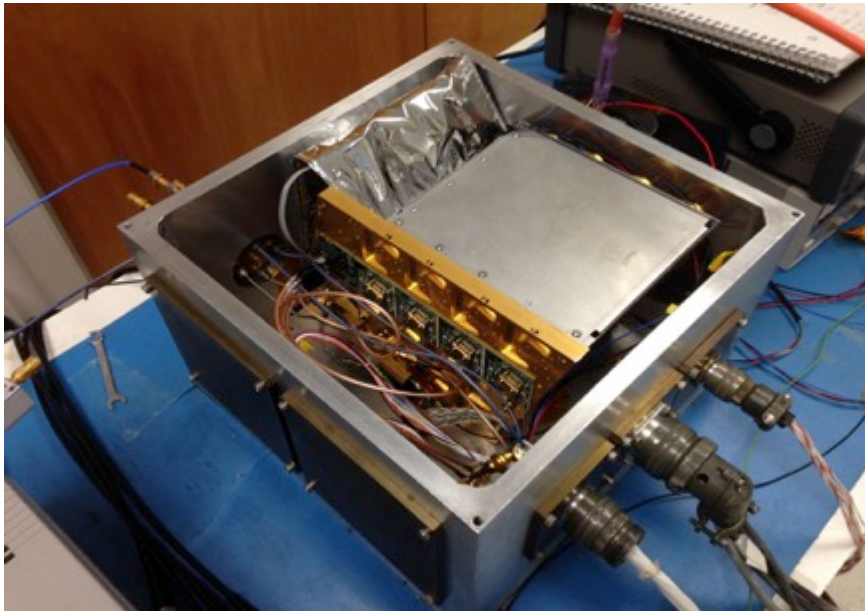
FPGA-based system level performance for a typical receiver.





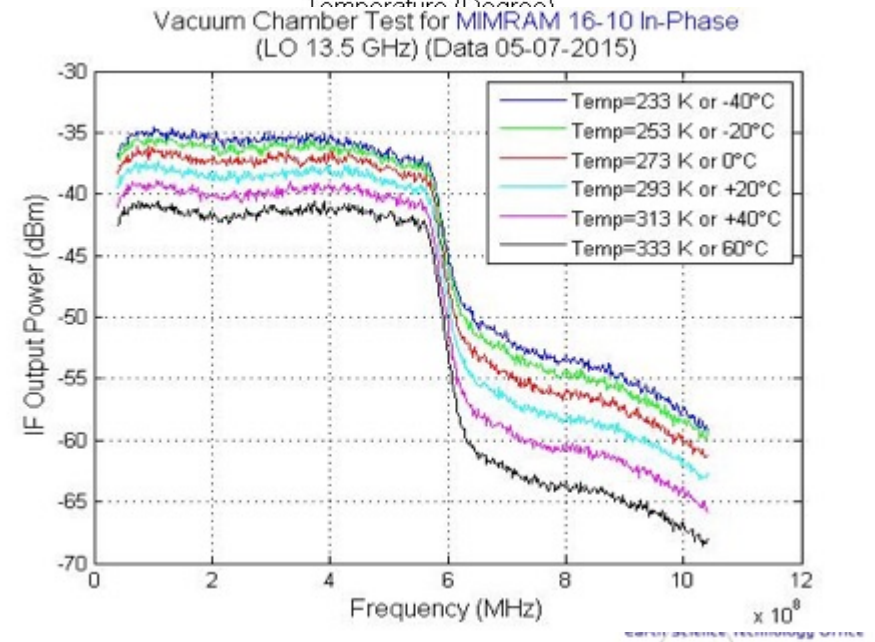
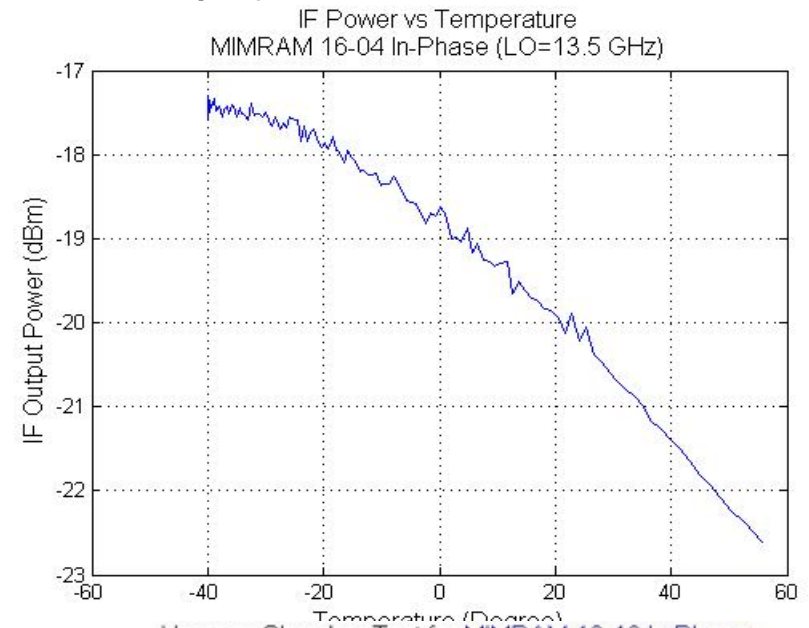
# Tile TVAC Tests-Raising TRL

(beyond proposed scope)



Tested in vacuum ( $<10^{-4}$  torr) over  $AFT \pm 10^\circ C$

4 IFs monitored, one continuously





# Tile/Board/MIMRAM Random Vibe Tests

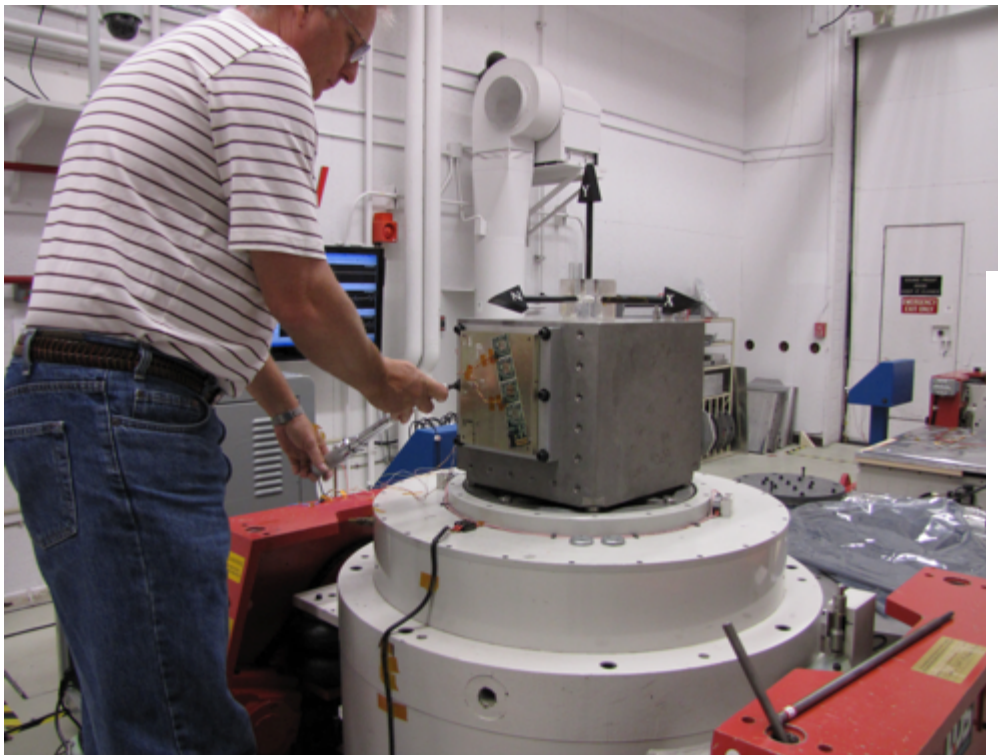
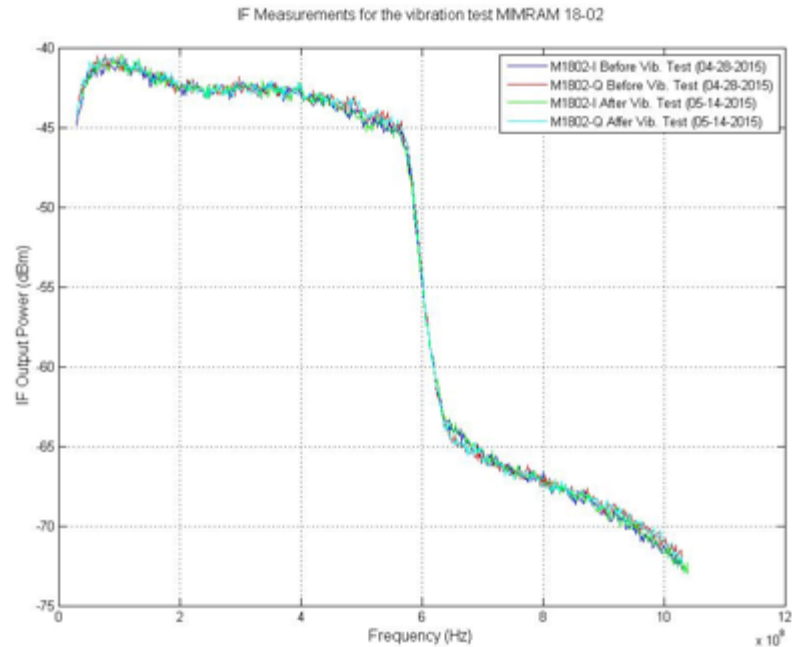


Table 5.2 - GEO Random Vibration Environment

Frequency (Hz)	Level
20	0.4 g <sup>2</sup> /Hz
20 – 50	+3 dB /octave
50 – 500	1.0 g <sup>2</sup> /Hz
500 – 2000	-4 dB /octave
2000	0.16 g <sup>2</sup> /Hz
Overall G <sub>rms</sub>	32.1

Test duration: 120 seconds

## Pre- and post-vibration data



# What can we do with GeoSTAR?

## TIME TESTED MEASUREMENTS AND DATA PRODUCTS USING MATURE ALGORITHMS

GeoSTAR will make similar measurements from GEO as AMSU currently does from LEO, but every 15 minutes vs. 2 times per day  
 High-intensity events can be sampled in 5 minutes or less

GeoSTAR will uniquely provide measurement of  
***Temperature/moisture/clouds; Wind; Precipitation***  
 simultaneously, continuously and in 3 dimensions

Parameter	Horizontal	Vertical	Temporal	Precision	Accuracy	Thermodyn.	Microphys.	Dynamics	
Brightness temperatures	25 - 50 km	N/A	5-20 min.	0.5-1.5 K	0.5 K				
Temperature	25 - 50 km	2-3 km	10-20 min	1.5 K	0.5 K	√			
Water vapor				25%	10%	√			
Wind vector (u,v)				5 m/s	2 m/s				√
Reflectivity				4 dBZ	2 dBZ				√
Rain rate		5 mm/hr		50%	√	√			
LWP		N/A		25%	10%	√			
IWP				25%	20%	√	√		

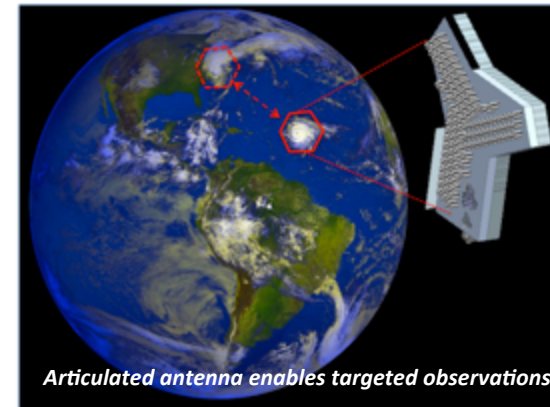
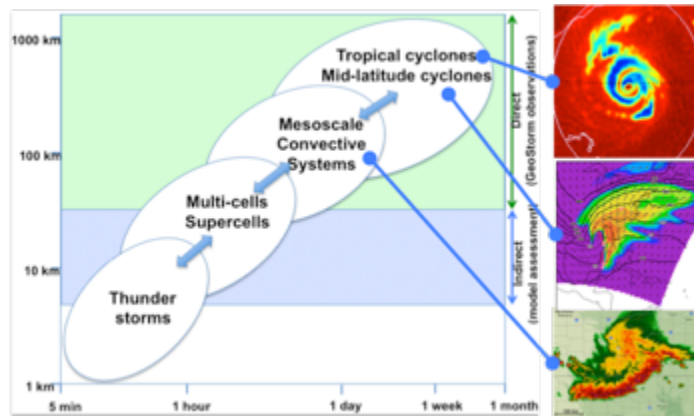
Precision & accuracy reflect performance of MIRS (except for reflectivity)



# GeoStorm

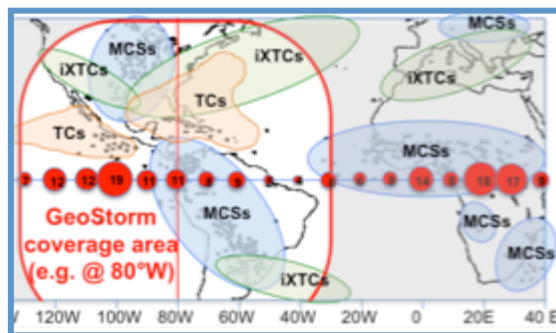
*Improve our understanding of sudden and unpredicted change in intensification and motion of destructive storms:*

- *hurricanes*
- *severe thunderstorms and mesoscale convective systems*
- *mid-latitude cyclones and winter storms*



Low cost as a hosted payload

Many hosting opportunities in GEO:



There are more than 80 GEO comm-sats that provides a view of the Americas, being replaced at a rate of 5-6 per year

## GeoStorm Highlights

Targeted observations	Life cycle storm tracking
Time-continuous	Capture dynamic processes; diurnal cycle fully resolved
Multiple simultaneous key parameters	Temperature, humidity, precipitation, wind
All-weather	Cloud/rain-penetrating
3-D observations	1000 km dia x 15 km vert. (volume); 25 km dia x 3 km vert. (resolution)
Wide coverage	All storms visible from GEO

# Summary

- STAR concept demonstrated in IIP-03
  - Developed a functional 50-GHz STAR demonstrator
- Key technologies developed in IIP-07
  - Developed miniature low-power 183-GHz MMIC receivers
  - Developed new alias-rejecting antenna array design
- Ready for PATH mission after IIP-10
  - Full-size 64x64 correlator ASIC is a success!
  - Can start development ~2015 → launch ~2020
- We have advanced the technology from Tier-3 level to Tier-1 level – a major achievement