

REMI - Reduced Envelope Multi-Spectral Imager for Sustained Land Imaging

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Reduced Envelope Multispectral Imager (REMI)



- Airborne, multispectral VIS-SWIR demonstrator for Sustainable Land Imaging
- In fabrication phase through September 2018
- Flight tests: Fall 2018, Spring 2019
- The REMI architecture meets SLI req's with an instrument that is much smaller than OLI by using a precision 2-axis mechanism to stabilize the scene during image acquisition.
- Features of the REMI architecture:
 - Single, reflective aperture that can also support the thermal infrared channels for full SLI spectral coverage
 - Mechanism and associated electronic control with space-flight heritage from TEMPO and GEMS
 - Much simpler focal plane than OLI and OLI-2 with associated lower costs and risks for acquisition and integration and test





All VSWIR Spectral Bands Demonstrated



Full VSWIR optical solution for SLI-T demo

- Proposal: 4 visible bands and 2 SWIR bands
 - Demonstrate step-stare approach
 - Multiple optical paths with single aperture
- Baseline: Enable all 5 visible bands, the Cirrus band and both SWIR bands

TABLE A.2 SLI-T REFERENCE MISSION SPECTRAL IMAGE PERFORMANCE REQUIREMENTS

Band #	Band Name	Band #	Center Wavelength (nm)	Center Wavelength Tolerance (nm)	Minimum Lower Band Edge (nm)	Maximum Upper Band Edge (nm)
1	Coastal Aerosol	1	448	2	443	453
2	Blue	2	482	5	450	515
3	Green	3	562	5	525	600
4	Red	4	655	5	630	680
5	NIR	5	865	5	845	885
6	SWIR 1	6	1610	10	1560	1660
7	SWIR 2	7	2200	10	2100	2300
8	Panchromatic	N/A	590	10	500	680
9	Cirrus	9	1375	5	1360	1390
10	Thermal 1	N/A	10800	200	10300	11300
11	Thermal 2	N/A	12000	200	11500	12000



Ball Aerospace

Instrument Design Accommodates VNIR to TIR **VNIR Optical Path SWIR Optical Path TIR Optical Path**

- The VNIR and SWIR channels are within scope of the REMI program
- Instrument design can accommodate and additional TIR channel
 - Compatible with the current optical design
 - Some modifications to instrument base plate and housing would be required

Image Motion Control Enables Smaller Aperture



- In optical instruments, SWaP scales with aperture
- SLI funded REIS (NASA SLI) identified that the driving requirement for aperture size was relative edge response (RER)
 - RER performance is a function of
 - Motion blur
 - Optics blur
 - Detector footprint
 - Platform jitter
- REMI utilizes active image motion control to
 - Reduce motion blur and platform jitter that offset the impacts to smaller aperture
 - Enable smaller aperture / instrument footprint while meeting SLI imaging requirements

Scan Approach Opens the Design Space



- Whisk Broom: LandSat 1-7
- Push Broom: LandSat 8 & 9
- Step-Stare with Image Motion Correction: SLI-T/REMI



Comparison of three different scan methodologies: Whisk Broom, Push Broom, and Step-Stare.



(OLI equivalent VIS IFOV = 42 urad)





Scan Modeling (On-Orbit Case)

(OLI equivalent VIS IFOV = 42 urad)



Aircraft is the more challenging environment

REMI Program Status



- REMI design and all drawings are complete
 - Parts are on order and are planned to arrive when needed
- First stage lens bonding completed
- Second stage lens bonding in process
 - SWIR optics currently on laser alignment station
 - VNIR optics staged for second stage lens bonding
- WASM manufacturing complete
 - In final calibration
 - Preparing for rate table testing
- Flight mechanical housing complete
 - Mounting plate complete as is instrument cover and thermal interface and plumbing
- REMI control electronics box complete
- Preparing for system integration later this summer

Rate table testing planned for July 2018









3-axis angular vibration and rate table

Test will verify scan mirror performance for both spacecraft and Twin Otter disturbance environments

Optical alignment in process













Flight Mechanical Housing and Control Assembly Complete





Flight mechanical baseplate

Housing and thermal subsystem



Control Assy



SWIR Camera Thermal Testing

- Ball
- The SWIR camera was tested using the REMI thermal system to predict performance in flight configuration
 - Verified that the radiometric accuracy of the flight system will be limited by the shot noise of the detector







Twin Otter Integration Configuration Complete







Attachment to Mount





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Flight Plans





Summary



- REMI utilizes a high performance scan mirror to achieve Landsat requirements with significant reduction in SWaP
 - Reduces image smear typical of Landsat instruments
 - Enables smaller aperture
 - Enables use of simple detector types
 - Key scan mirror technology has been space qualified on the GEMS and TEMPO programs
- REMI is currently in integration and test
- Engineering flights planned for Fall 2018, followed by science flights in 2019.

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THANK YOU!