
Temporal Experiment for Storms and Tropical Systems Technology Demonstration (TEMPEST-D) to Enable Temporally-Resolved Observations of Clouds and Precipitation on a Global Basis using 6U-Class Satellite Constellations

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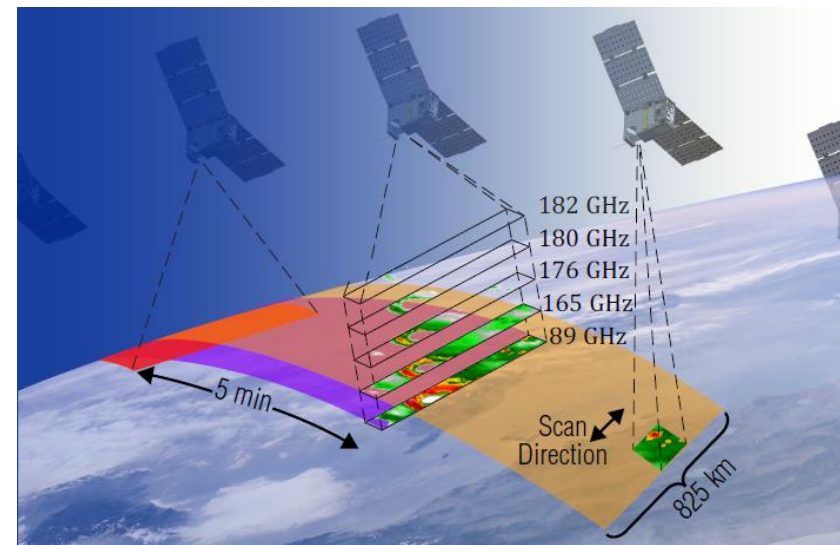
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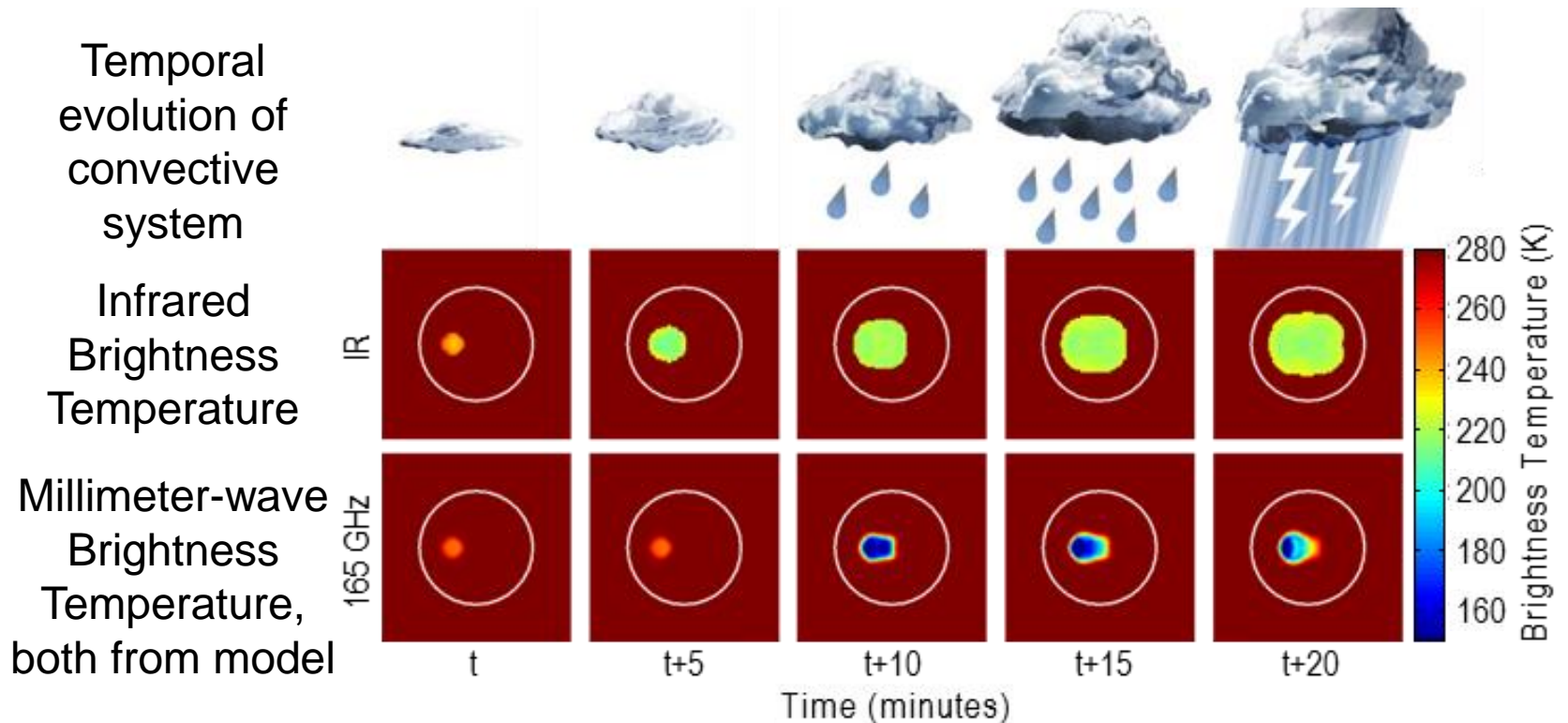
TEMPEST addresses 2017 Earth Science Decadal Survey Question W-4:

- *Why do convective storms, heavy precipitation, and clouds occur exactly when and where they do?*
 - Provides global, *temporally-resolved* observations of cloud and precipitation processes using a 6U-Class satellite constellation
 - Constrains weather and climate models using millimeter-wave radiometer observations
- TEMPEST-D technology demonstration began in August 2015 as a partnership among CSU, JPL and BCT.
 - Delivered 6U flight system with integrated payload to NanoRacks for launch integration on March 22, 2018.
- NASA CubeSat Launch Initiative (CSLI)
- Launched by Orbital ATK on CRS-9 from NASA Wallops to ISS on May 21, 2018
- Planned deployment into orbit by NanoRacks from JEM in July/August 2018



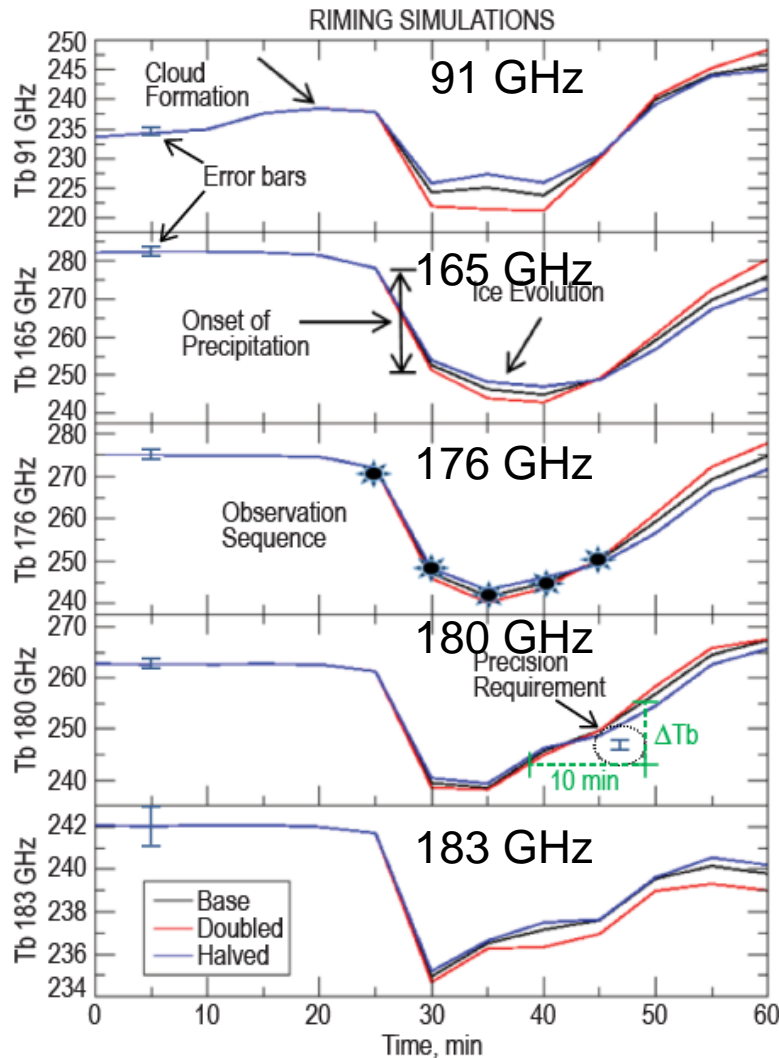
5 identical 6U small sats, each with an identical 5-channel radiometer, flying 5 minutes apart

Observations of Transition from Clouds to Precipitation

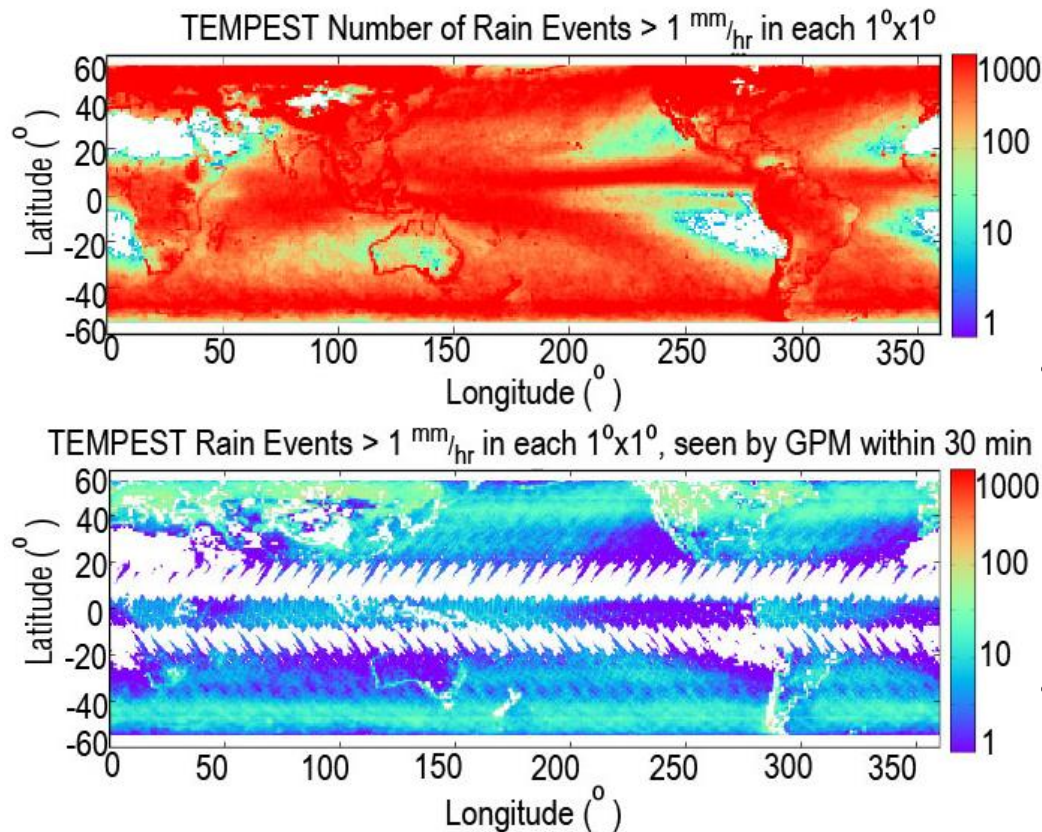


- Infrared brightness temperatures (middle row, available from GEO) show cloud top temperatures, locations and morphology.
- Onset of precipitation clearly detected at millimeter-wave frequencies on TEMPEST constellation, including 165 GHz (bottom row).
- TEMPEST minimum spatial resolution of 25 km is shown (circles).

Temporal Development of Ice in Cloud-Scale Models



- Modeled brightness temperatures at the five TEMPEST frequencies with 25-km spatial resolution
- Simulations compare different rates of supercooled water droplets collecting on ice crystals (riming efficiency).
- Rate varies from baseline (black) to twice (red) and half (blue).
- Measurable difference between curves is 4 K or greater in 5 minutes at onset of ice formation. Instrument precision requirement is 1 K in 5 minutes.
- Ice remaining in clouds after precipitation has substantial effects on climate. Residual ice can be compared to W-band radar observations from NASA's CloudSat or ESA's EarthCARE.



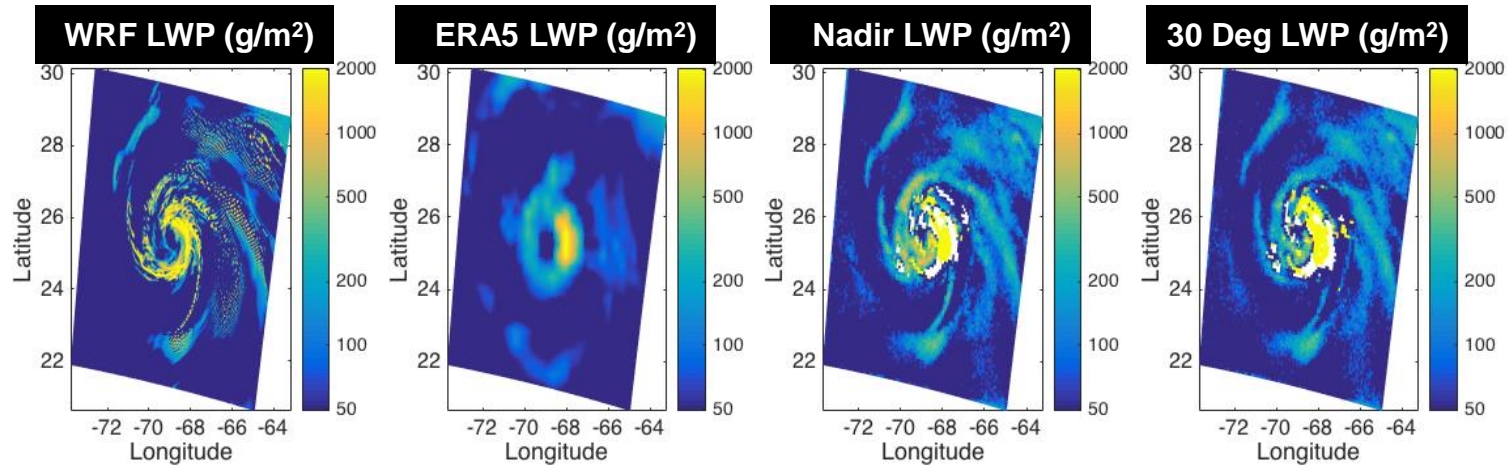
- During a future one-year mission, TEMPEST constellation could make more than 3,000,000 time-resolved observations of precipitation ($> 1 \text{ mm/hr}$), including 100,000+ deep-convection events
- Could perform more than 50,000 precipitation observations coincident (within 30 minutes) with NASA/JAXA Global Precipitation Mission (GPM) core satellite
- Assumes nominal TEMPEST orbit for deployment from ISS at 400-km altitude and 51.6° inclination.
- Precipitation estimates from AMSR-E satellite radiometer data with oceanic observations only.
- TEMPEST mission observations will be complementary to NASA CYGNSS and NASA TROPICS Earth Venture missions.

“Truth”

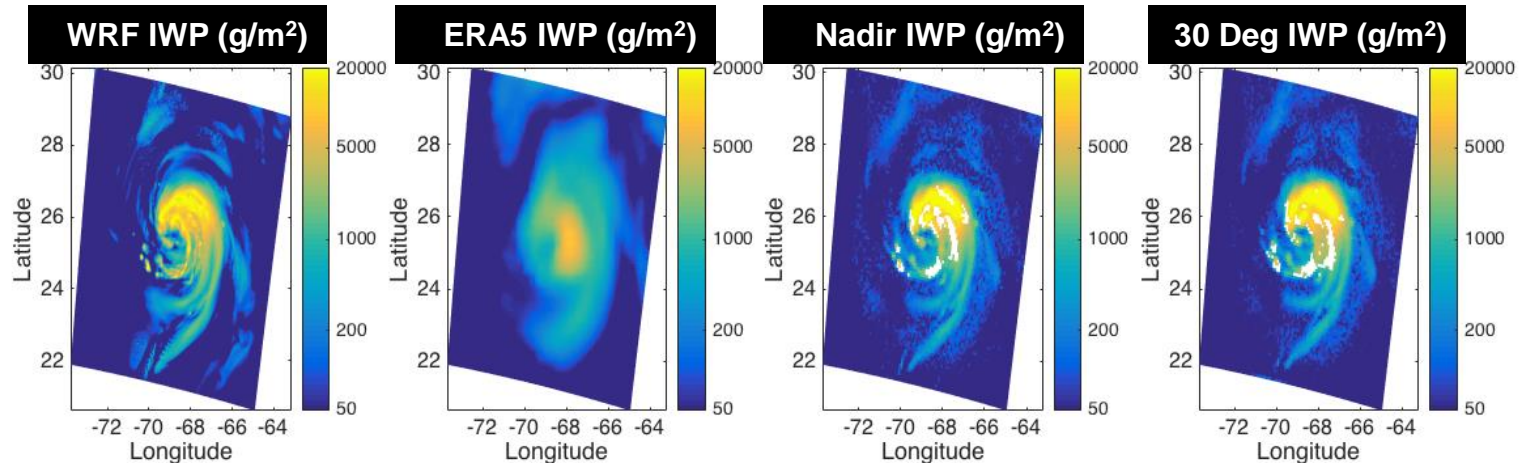
A-priori

Retrievals

Cloud Liquid

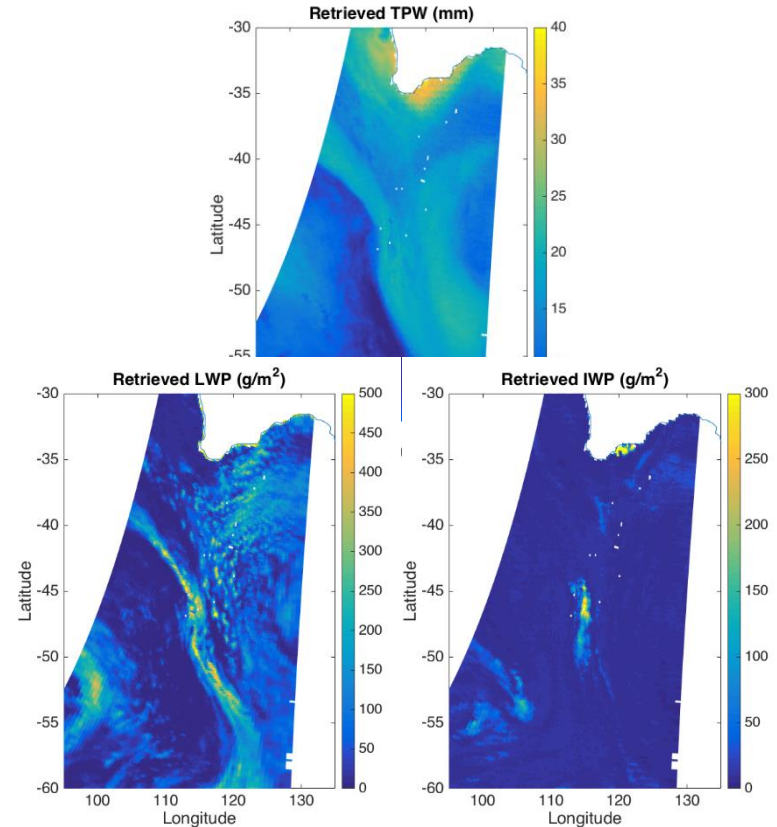


Cloud Ice



From WRF model of Hurricane Gonzalo in Oct. 2014

- Builds on CSU 1D-Var (Duncan and Kummerow, 2016); observed brightness temperatures balanced against a-priori knowledge to find most likely state vector
- Retrieves water vapor profile (3 EOFs) and cloud liquid water path (LWP), with option to retrieve cloud ice water path (IWP)
- View-angle-dependent measurement error covariance matrix (\mathbf{S}_y) greatly reduces biases near the edge of the swath
- Applicable to both TEMPEST-D risk reduction and improved TEMPEST-D vs. MHS single and double difference validation activities.



1D-Var retrieval of total precipitable water, liquid water path, and ice water path from the Microwave Humidity Sounder (MHS)

TEMPEST-D Demonstration: Motivation and Objectives

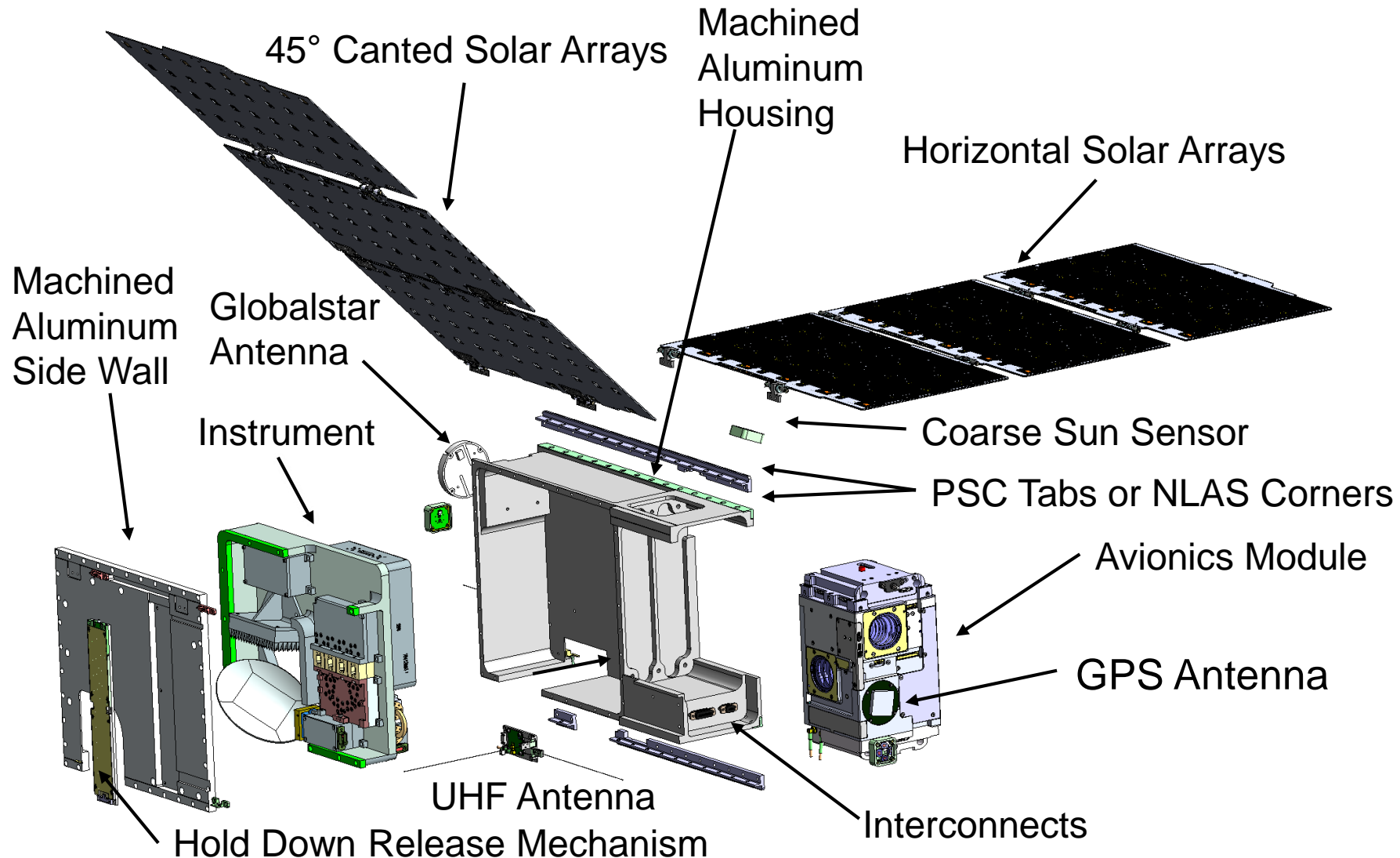


- Demonstrate capability of 6U CubeSats to contribute to NASA Earth Science measurements in a 90-day technology demonstration mission
- Reduce **risk**, cost and development time for future measurements of Earth science processes using CubeSat constellations
- Raise the technology readiness level (TRL) of the TEMPEST mm-wave radiometer instrument from 5 to 9 (scanning reflector to 7)
- Provides the first in-space demonstration of a millimeter-wave radiometer based on an InP HEMT low-noise amplifier front-end (LNA) for Earth Science measurements.

Success Criteria:

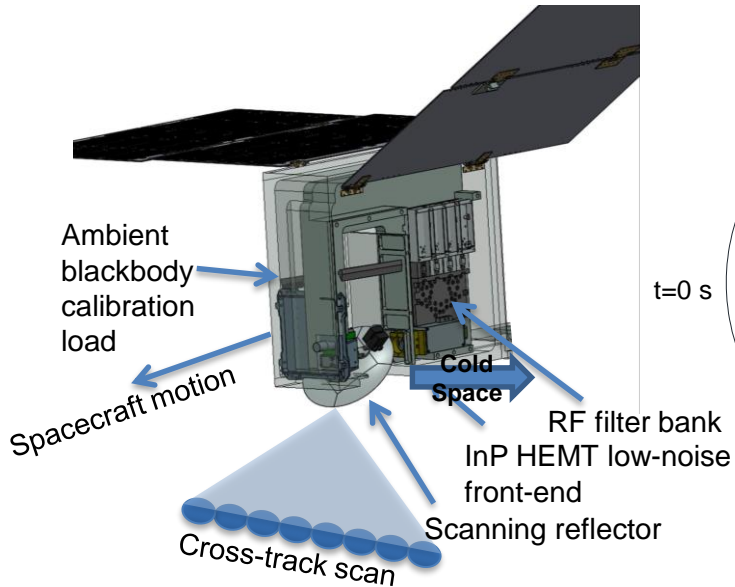
- Demonstrate feasibility of differential drag maneuvers to achieve required time separation of 6U-Class satellites in same orbital plane
- Demonstrate cross-calibration between TEMPEST mm-wave radiometers and NASA/JAXA Global Precipitation Mission Microwave Imager and/or Microwave Humidity Sounder (MHS, on two NOAA satellites and two ESA/EUMETSAT satellites) with 2 K precision and 4 K accuracy.

TEMPEST-D 6U-Class BCT Spacecraft Bus based on XB1

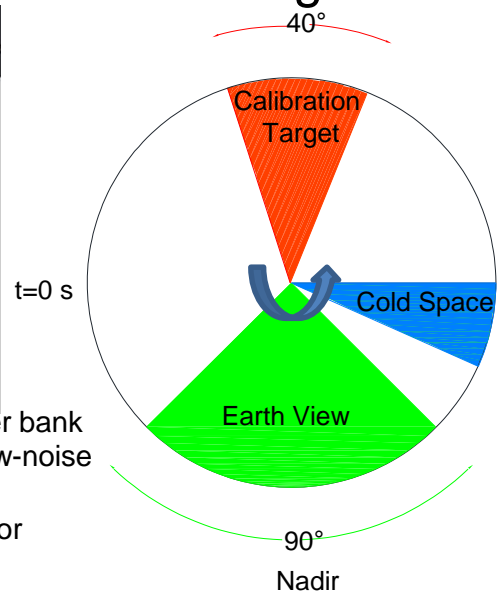


TEMPEST-D Instrument: Radiometer Calibration

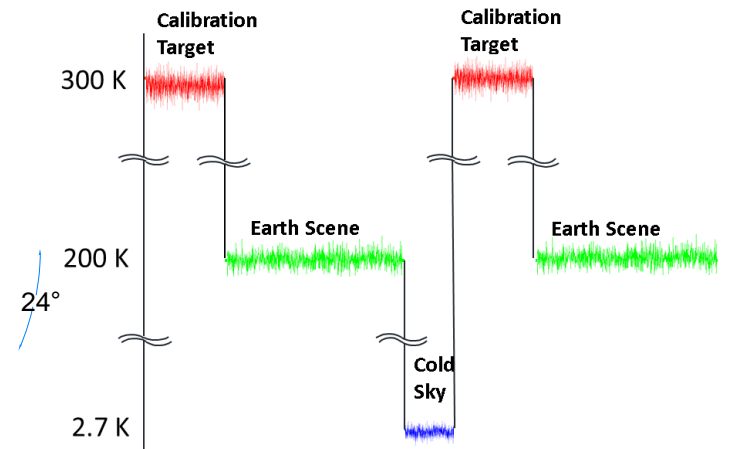
TEMPEST-D Instrument



Observing Profile

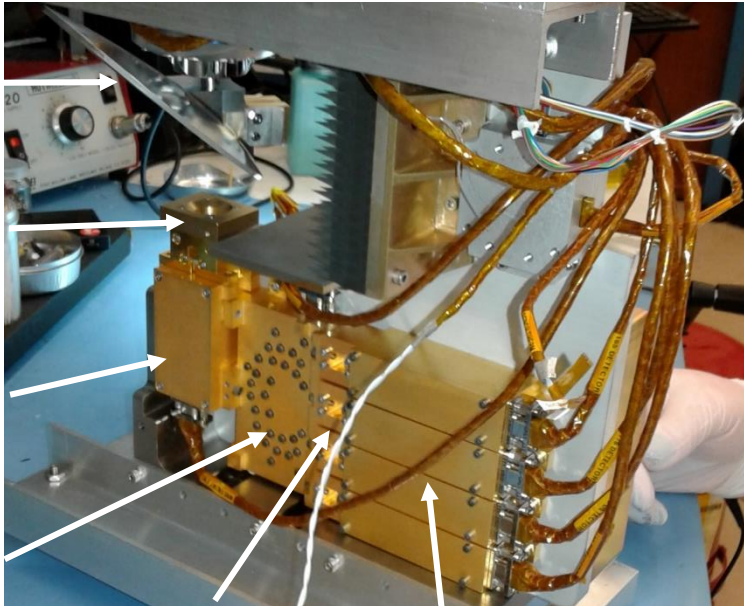


Time Series of Output Data



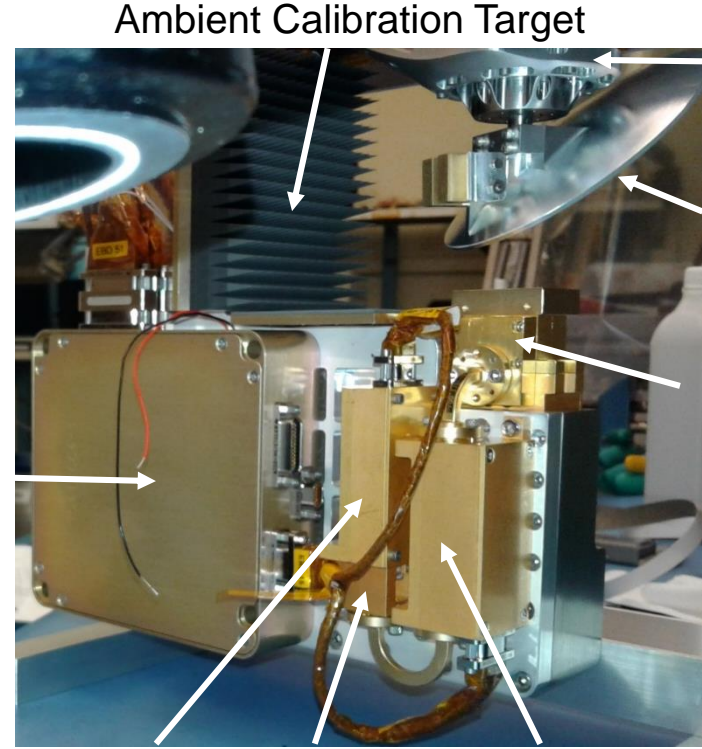
- Five-frequency millimeter-wave radiometer measures Earth scene over $\pm 45^\circ$ nadir angles, for an 825-km swath width from a nominal orbit altitude of 400 km. Spatial resolution ranges from 13 to 25 km for 89 to 182 GHz.
- TEMPEST-D performs two-point end-to-end calibration every 2 sec. by measuring cosmic microwave background at 2.73 K (“cold sky”) and ambient blackbody calibration target each revolution (scanning at 30 RPM).

Flight Model Radiometer Instrument Bench-top Integration at JPL



Scanning Reflector
Dual-Frequency Feed horn
165-182 GHz Radiometer Front-end
165-182 GHz Power Divider

165-182 GHz Filter Bank
165-182 GHz Detectors
Command & Data Handling and Power Distribution Subsystem



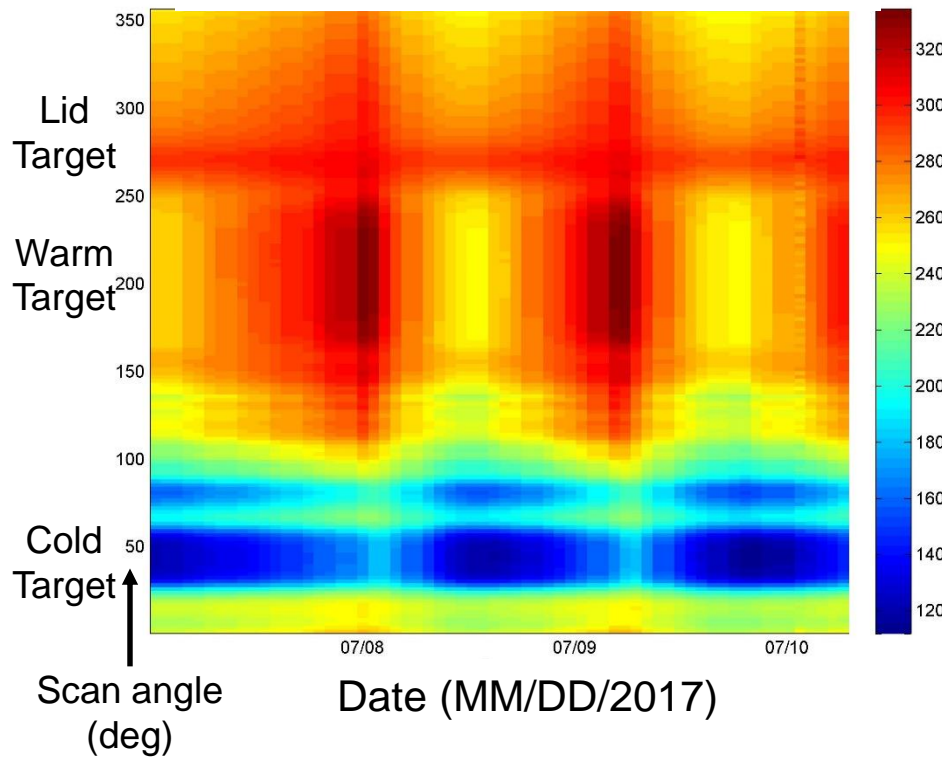
Ambient Calibration Target

Scanning Motor
Scanning Reflector
Dual-Frequency Feed horn

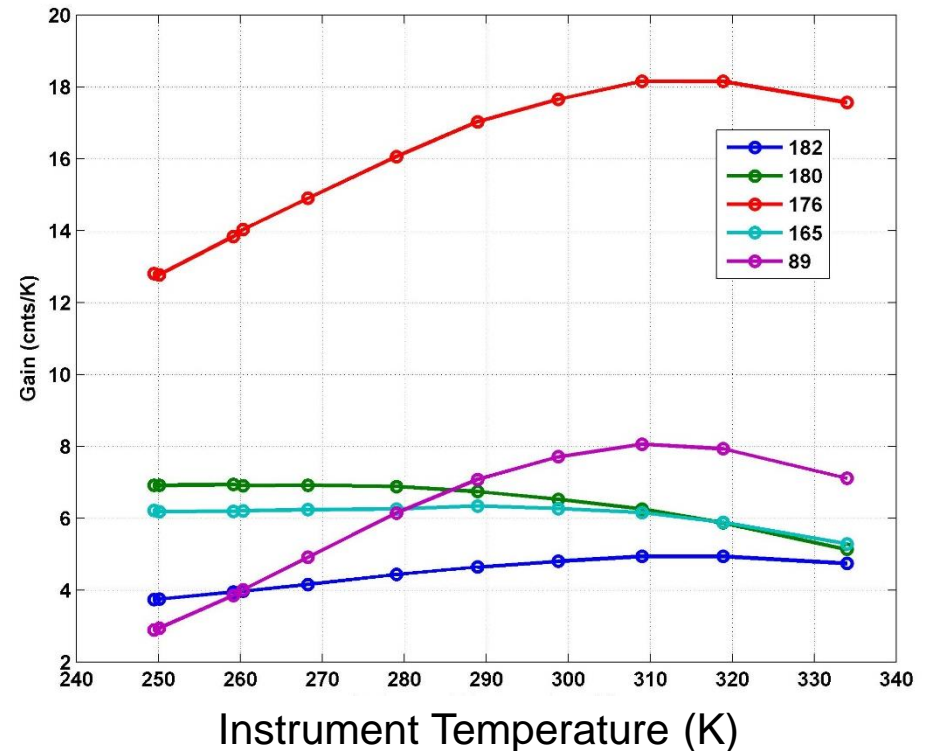
89 GHz Radiometer Front-end
89 GHz BP Filter
89 GHz Detector

Thermal Vacuum Testing Results for Flight Instrument (Jul. 2017)

89 GHz Antenna Temperature (K)

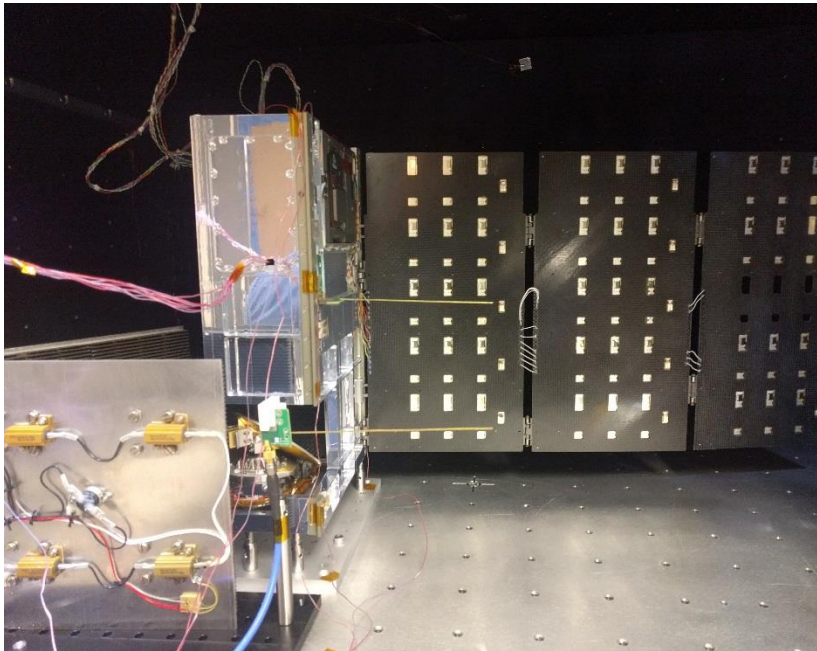


Gain vs. Temperature

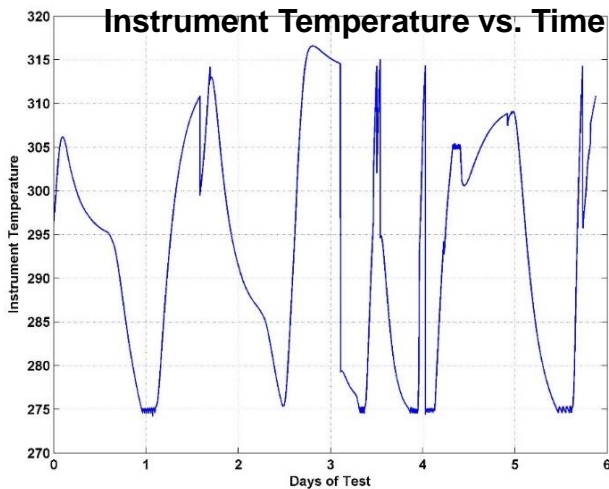
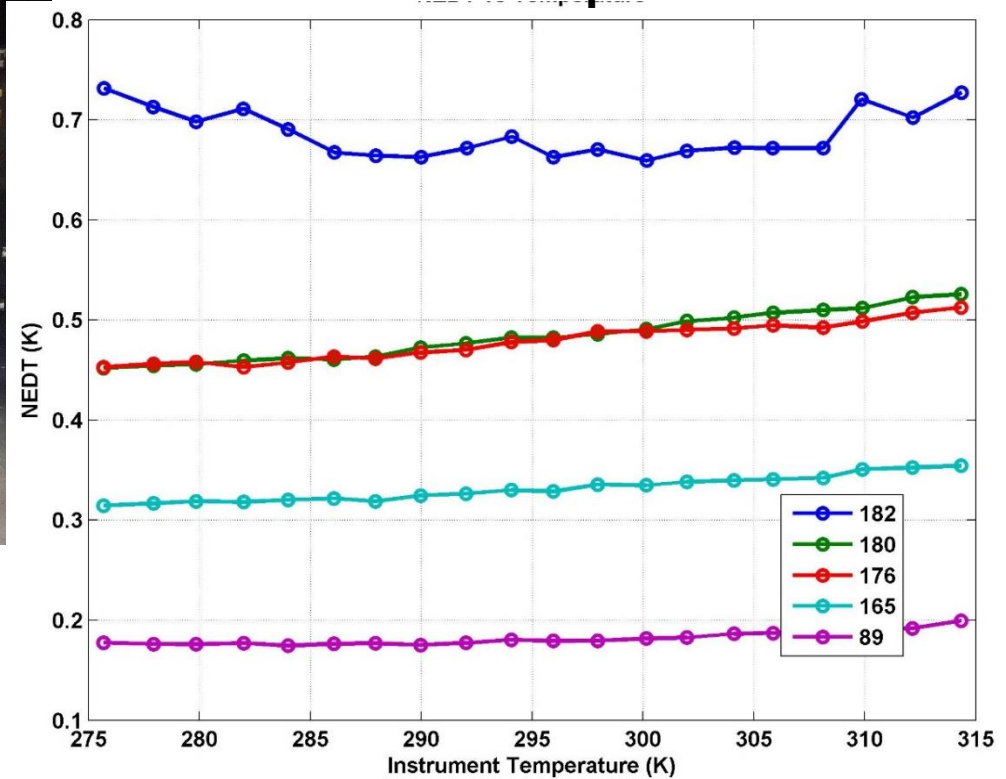


Gain measured while viewing blackbody calibration target at chamber temperature varying from -25°C to $+60^{\circ}\text{C}$.

Instrument Performance during Spacecraft TVac Testing (Jan. 2018)



Radiometric Resolution vs. Instrument Temperature



Measured radiometric resolution values meet total noise requirements of 1.4 K for all five millimeter-wave radiometer channels.

Radiometer Instrument Technical Resources Summary

Resource	CBE	Allocation	Margin (Actual)
Radiometer Mass (kg)*	3.75	4.0	6.25%
Radiometer Power (W)	6.0	6.5	8%
Radiometer Data Rate (Kbps)**	10.3	12.3	16%
Radiometer Precision (K)	0.4-0.95	1.4	71-32%
Radiometer Accuracy (K)	3.5	4	13%

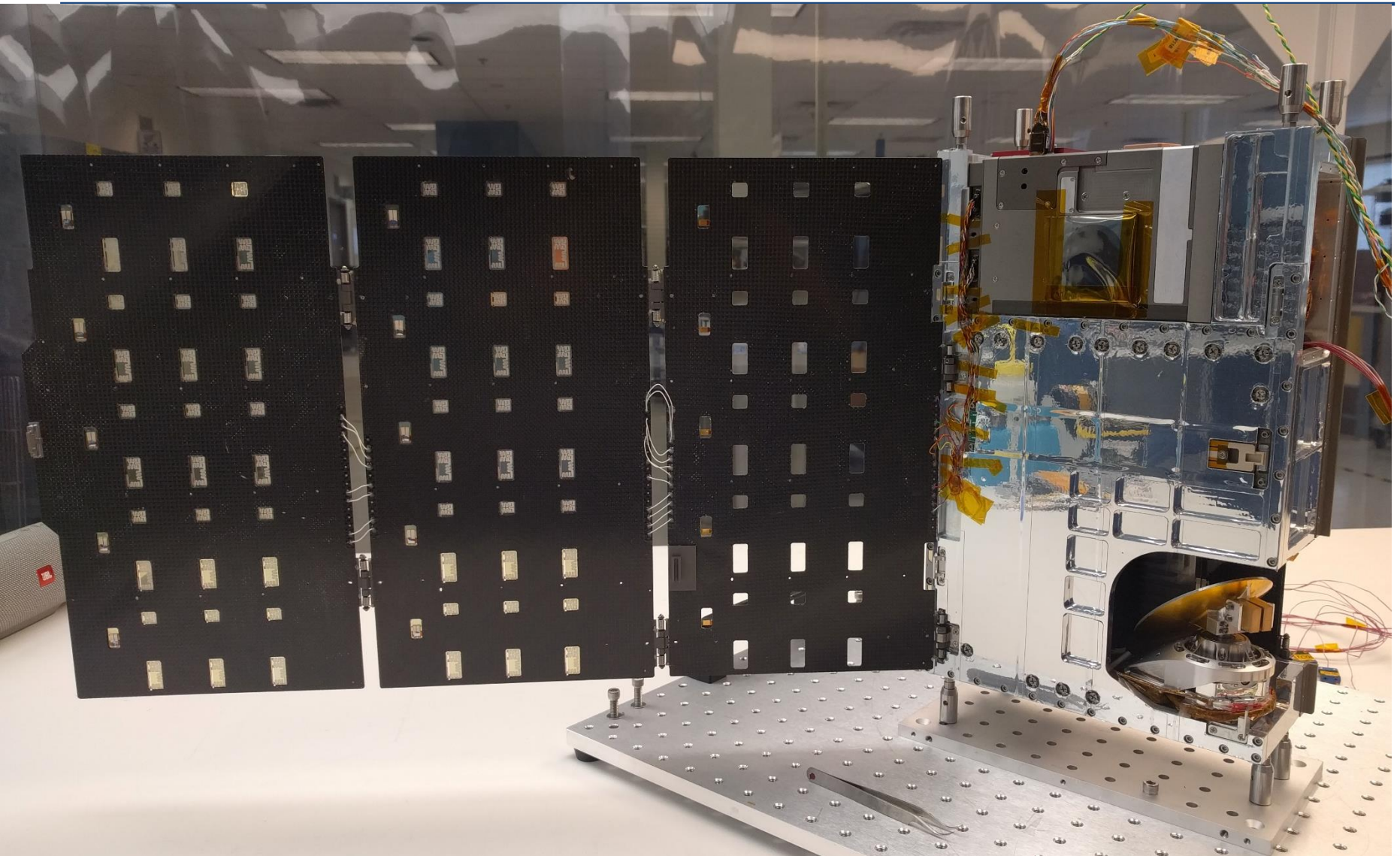
$$MARGIN = 100 \times \frac{Allocation - CBE}{Allocation}$$

*Change due to titanium standoff replacement with aluminum

**Includes spacecraft state-of-health telemetry

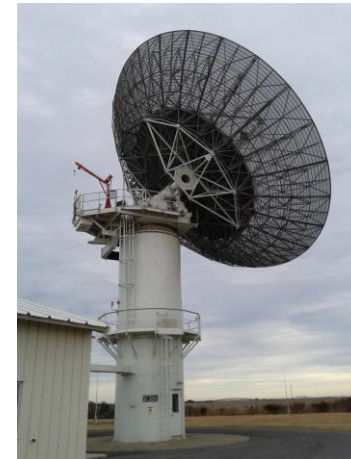
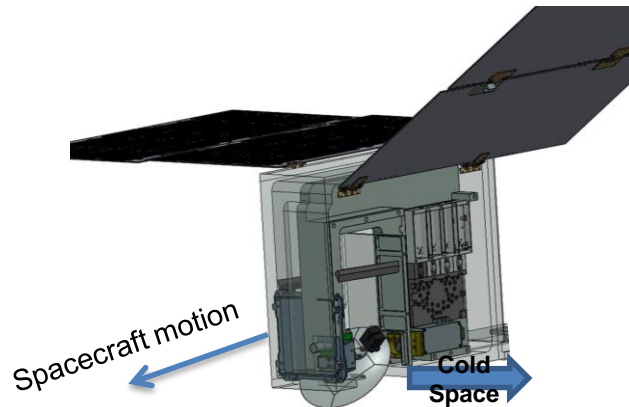
All excess margin can now be released to the spacecraft.

TEMPEST-D Flight Unit Integrated at BCT (Feb. 2018)



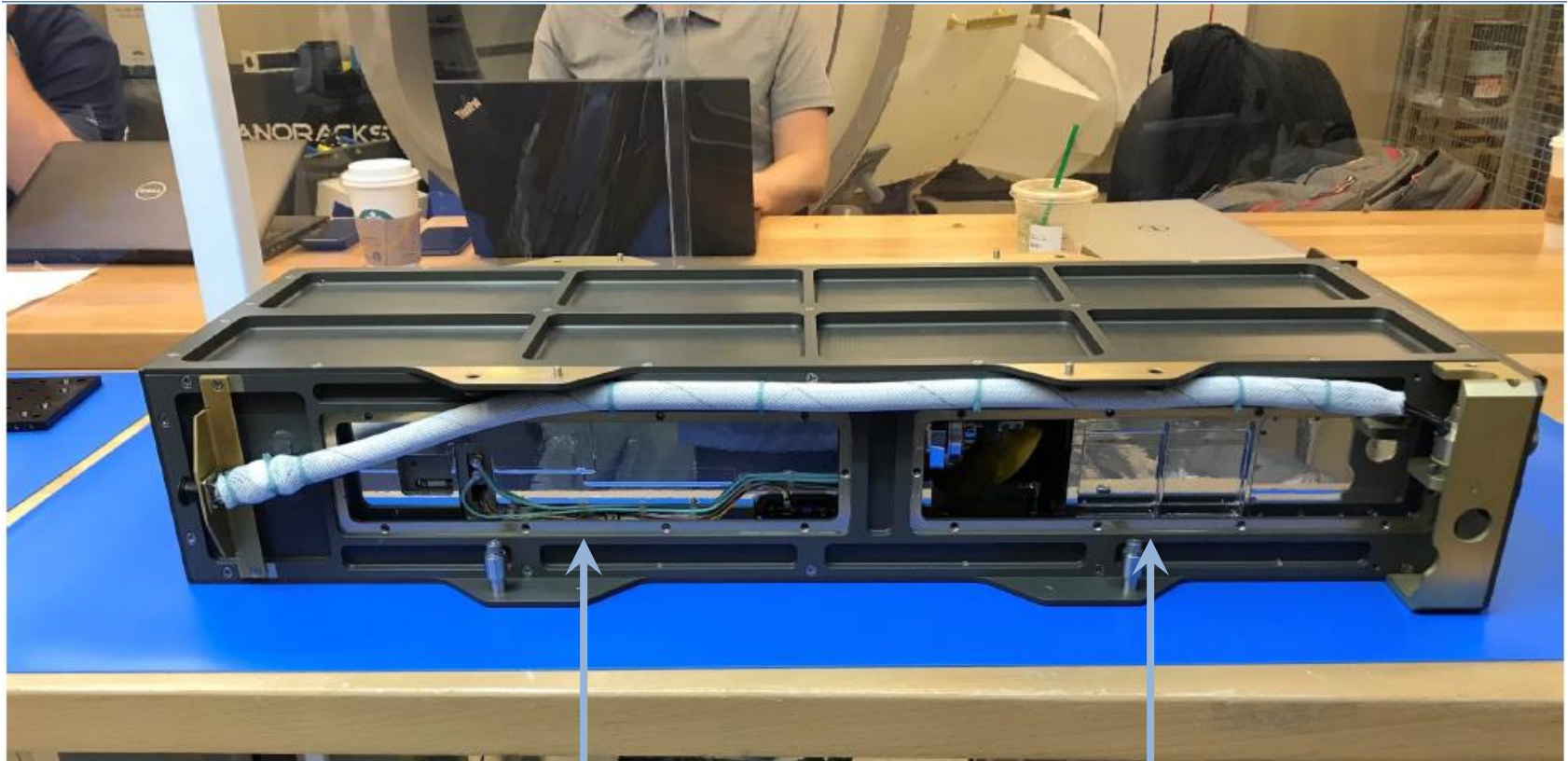
TEMPEST-D Mission: Recent Milestones

- Ground compatibility testing completed at NASA Wallops on Feb. 16, 2018
- Workmanship vibration penalty testing completed on Feb 22, 2018.
- Thermal balance penalty testing completed on Mar. 2, 2018.
- FCC granted radio communication license on Mar. 8, 2018
 - Primary Comms: Cadet-U to NASA Wallops, UHF ~450 MHz Uplink and ~470 MHz Downlink
 - Secondary Comms: S-band Globalstar to MEO: ~1.6 GHz
- Successfully integrated into NanoRacks 6U Deployer along with CubeRRT on Mar. 22, 2018.
- Launched to ISS on OA-9 on May 21, 2018



NASA Wallops 18-m dish

Final Installation in Deployer at NanoRacks in Houston, Mar. 2018



CubeRRT

TEMPEST-D

Launched on Orbital ATK OA-9 from NASA Wallops to ISS on May 21, 2018



Photo Credit: NASA

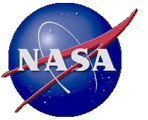


Orbital ATK Cygnus Arrived at ISS on May 24, 2018



Photo Credit: NASA

Summary



- TEMPEST-D mission to demonstrate capability of 6U CubeSats to perform global temporally-resolved observations of cloud and precipitation *processes*
- Reduces risk, cost and development time for future CubeSat constellation missions to perform repeat-pass radiometry to measure temporal signatures of atmospheric processes
- Provides first in-space technology demonstration of a millimeter-wave radiometer based on an InP HEMT low-noise amplifier front-end for Earth Science measurements
- Demonstrates feasibility of differential drag maneuvers to achieve required time separation of 6U CubeSats in the same orbital plane
- Demonstrates cross-calibration of TEMPEST radiometers with NASA/JAXA GPM Microwave Imager and/or MHS with 2 K precision and 4 K accuracy
- Demonstrates capability for rapid development of CubeSats for Earth science, about 2.5 years from project start to delivery for launch integration
- Launched on Orbital ATK CRS-9 from NASA Wallops to ISS on May 21, 2018
- Planned for deployment into orbit from ISS by NanoRacks in July/Aug. 2018



Thank you for your kind attention. Many thanks to NASA Earth Ventures for their support and to the NASA Earth Science Technology Office for program management.