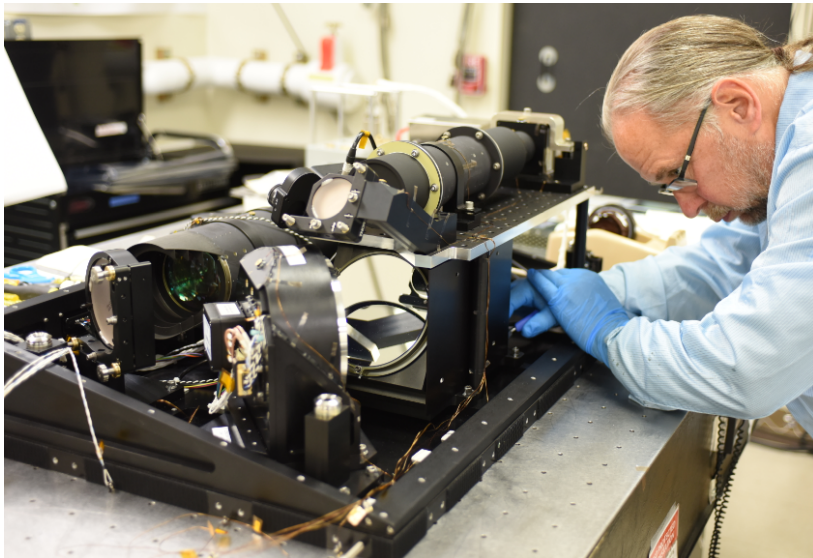# REMI - Reduced Envelope Multi-Spectral Imager for Sustained Land Imaging

Dennis Nicks, PI

James Howell, PM

Earth Science Technology Forum

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Sustainable Land Imaging (SLI) Program

- The Sustainable Land Imaging program aims to continue the 45+ year Landsat through the development of a new generation of smaller, less costly instruments that meet or exceed current Landsat imaging capabilities

- NASA ESTO SLI-Technology (2015 ROSES A.47)
  - New measurement technologies that enable future Land Imaging
  - Reduce overall program cost

NASA / USGS
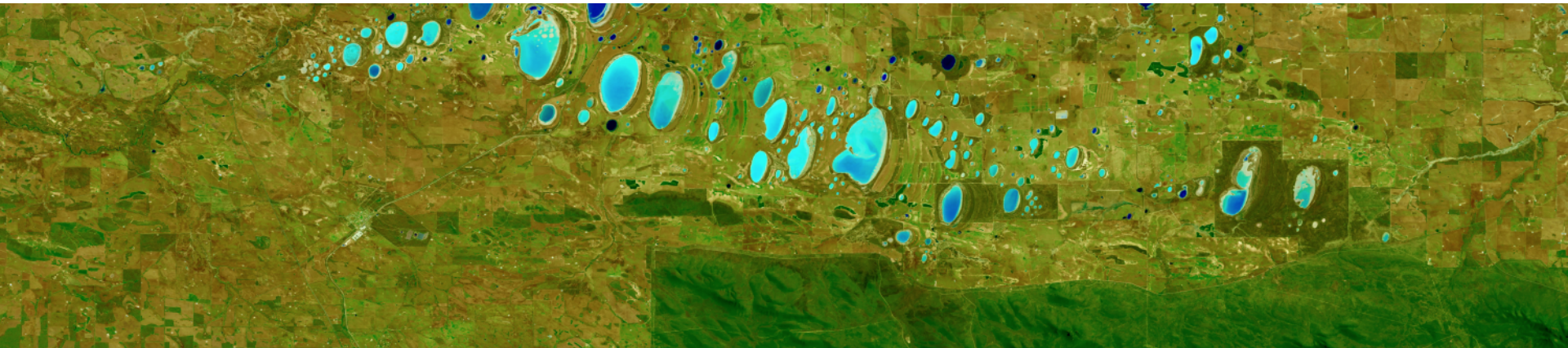
Reduced Envelope Multispectral Imager (REMI)

- Airborne, multispectral VIS-SWIR demonstrator

- Sensor fully integrated into aircraft test configuration

- Engineering flight completed – April 2019

- The REMI architecture meets SLI req's

- Features of the REMI architecture:

  - Small Size/Cost

  - One FPA per spectral range (1 @VNIR, 1 @ SWIR)

  - Step-Stare scanning enables high SNR while reducing aperture size

  - Incorporates Landsat spectral bands

  - Extensible to include TIR channels
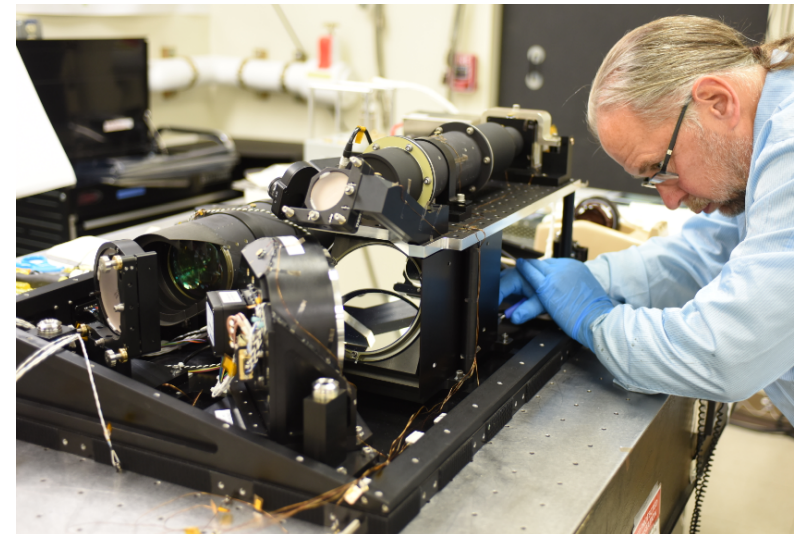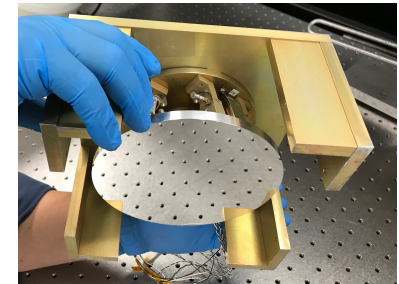
Top-Level REMI Mission Requirements

| Parameter	Value
Aircraft	Twin Otter De Havilland DHC-6
Flight altitude	4000 m AGL nominal; 5486 m (18,000 ft) max
Ground sampling distance	< 30 m equivalent at SLI altitude of 705 km
Spectral coverage	See next slide
Aircraft velocity	50 m/sec (~110 mph)
Spectral coverage	400 – 2500 nm; Bands 1-5 (VIS) and Bands 6,7,9 (SWIR)
Step-n-Stare frequency	15 – 20 Hz
Instrument environment	Pressure-controlled enclosure; thermally controlled
In-flight calibration	Baseline: On-board radiometric calibration before and after every flight. Support capability for in-flight calibrations.
Typical flight day	4.0 hours; 3.25 hours of data collection
On-board electronics & data system	Controls data acquisition, telemetry and data archiving; real-time display of data; instrument control
Data volume, typical flight day	~ 3.5 Tb
Ground data system	Management and archiving of data acquired during flights; Quick-look data assessment

All VSWIR Spectral Bands Demonstrated

Full VSWIR optical solution for SLI-T demo

- Baseline: Enable all 5 visible bands, the Cirrus band and both SWIR bands

- 1 Visible + 1 SWIR detector

VIS FPA

2496 pixels

Coastal A. Band 1 | Blue Band 2 | Green Band 3 | Red Band 4 | NIR Band 5

3360 pixels

SWIR FPA

1024 pixels

Band 7 | Band 9 | Band 6

1280 pixels

**TABLE A.2 SLI-T REFERENCE MISSION SPECTRAL IMAGE PERFORMANCE REQUIREMENTS**

Band #	Band Name	Band #	Center Wavelength (nm)	Center Wavelength Tolerance (nm)	Minimum Lower Band Edge (nm)	Maximum Upper Band Edge (nm)
1	Coastal Aerosol	1	448	2	443	453
2	Blue	2	482	5	450	515
3	Green	3	562	5	525	600
4	Red	4	655	5	630	680
5	NIR	5	865	5	845	885
6	SWIR 1	6	1610	10	1560	1660
7	SWIR 2	7	2200	10	2100	2300
8	Panchromatic	N/A	590	10	500	680
9	Cirrus	9	1375	5	1360	1390
10	Thermal 1	N/A	10800	200	10300	11300
11	Thermal 2	N/A	12000	200	11500	12000
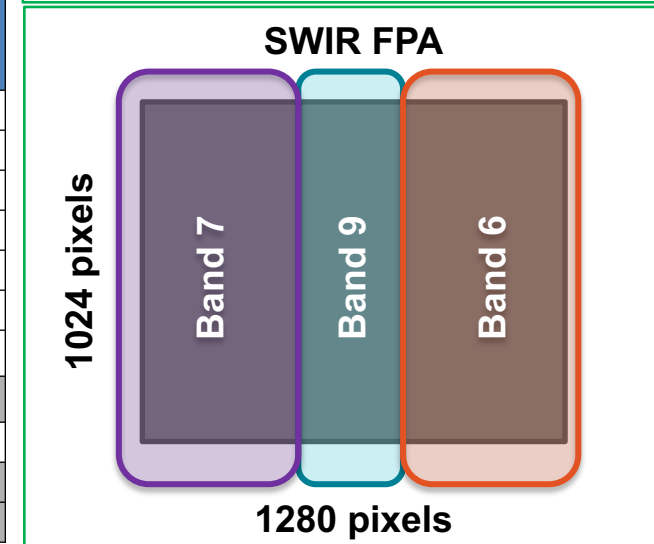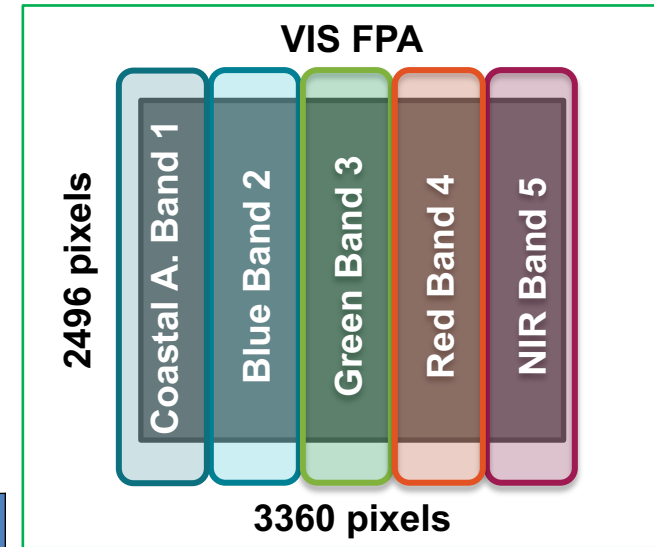
Instrument Design Accommodates VNIR to TIR
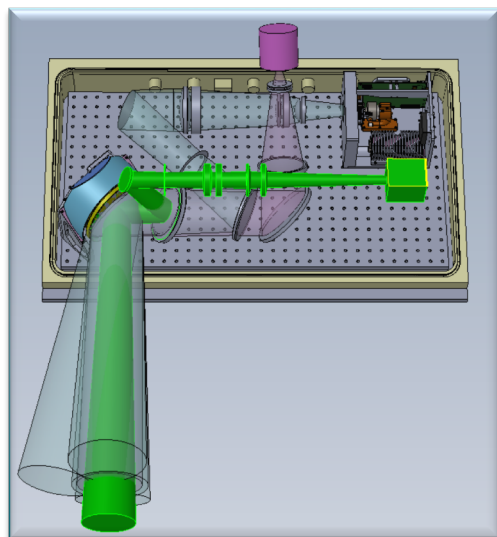
VNIR Optical Path
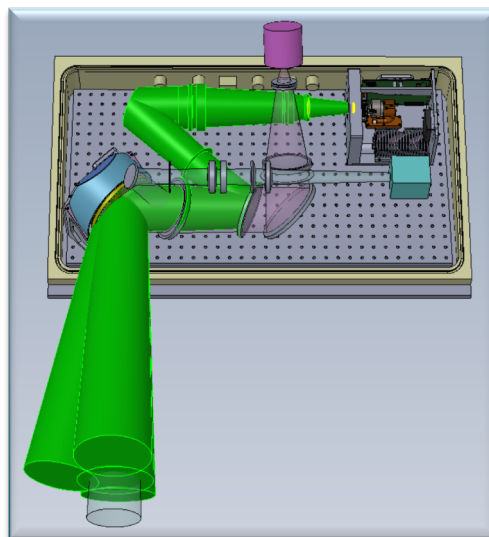
SWIR Optical Path

TIR Optical Path
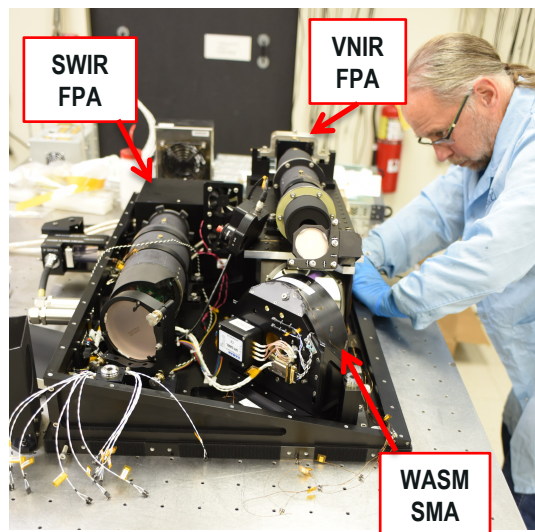

SWIR FPA

VNIR FPA

WASM SMA
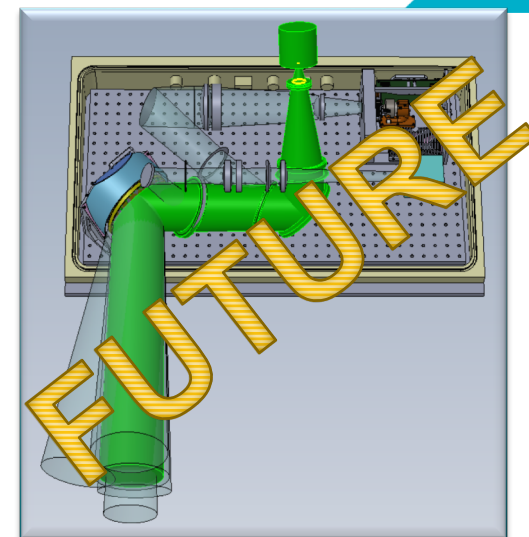
Image Motion Control Enables Smaller Aperture

- In optical instruments, SWaP scales with aperture

- SLI funded REIS (NASA SLI) identified that the driving requirement for aperture size was relative edge response (RER)
  - RER performance is a function of
    - Motion blur
    - Optics blur
    - Detector footprint
    - Platform jitter

- REMI utilizes active image motion control to
  - Reduce motion blur and platform jitter that offset the impacts to smaller aperture
  - Enable smaller aperture / instrument footprint while meeting SLI imaging requirements

- Scan mech reduces number of required detector arrays
  - REMI = 2x detectors for VNIR – SWIR
  - OLI = 14x detectors for VNIR - SWIR

Scan Approach Opens the Design Space

- **Whisk Broom: LandSat 1-7**
- **Push Broom: LandSat 8 & 9 (14 detector modules)**
- **Step-Stare with Image Motion Correction: SLI-T/REMI**
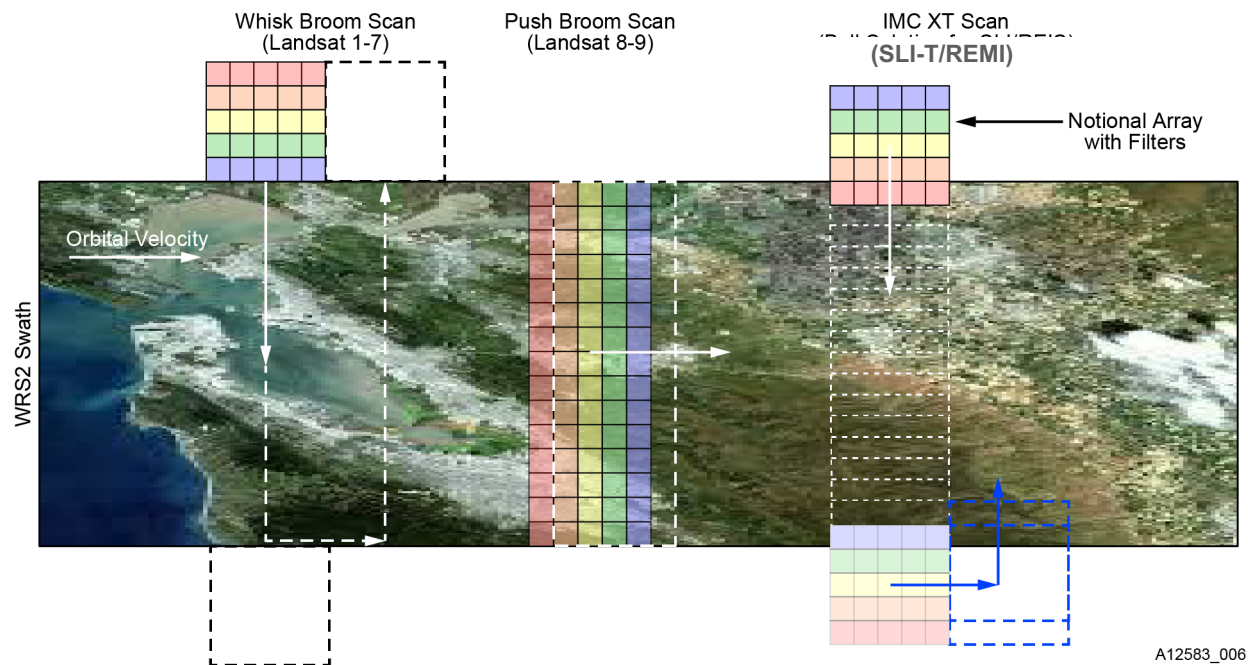
Comparison of three different scan methodologies:
Whisk Broom, Push Broom, and Step-Stare.
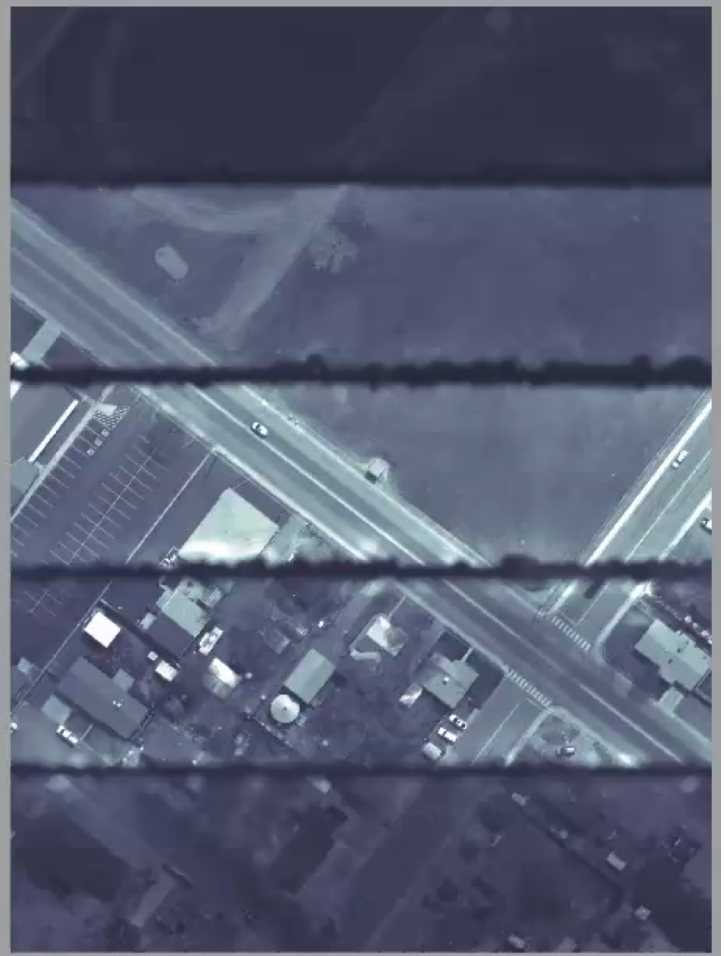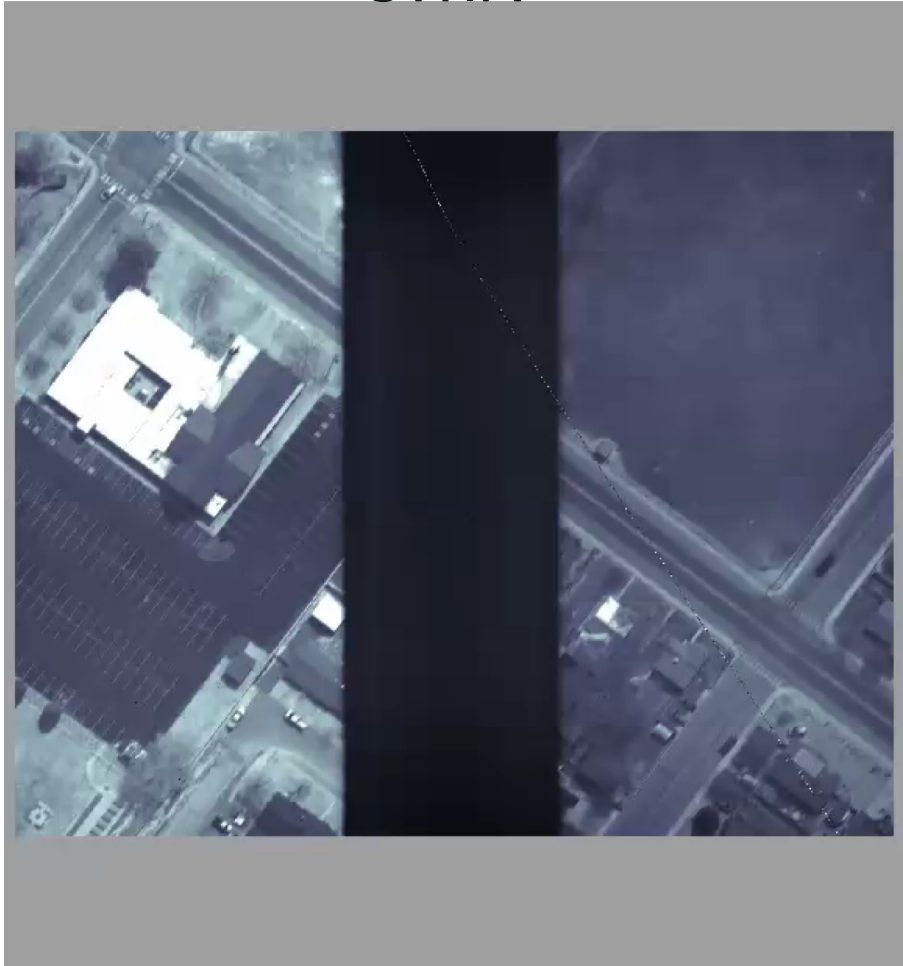
REMI Program Status

- Flight instrument integration complete
- Laboratory testing complete
- Engineering flight testing complete
- Review of engineering flight data in progress
- Planning for science flight in progress
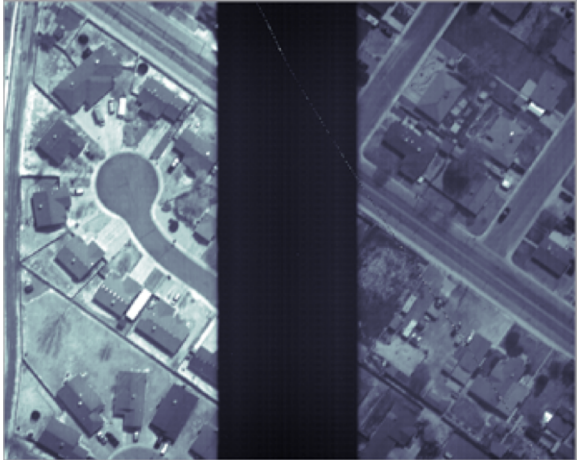
Quick-Look Imagery

SWIR

VNIR

Flight Images / Data

Band 7 2200 nm   Band 6 1610 nm
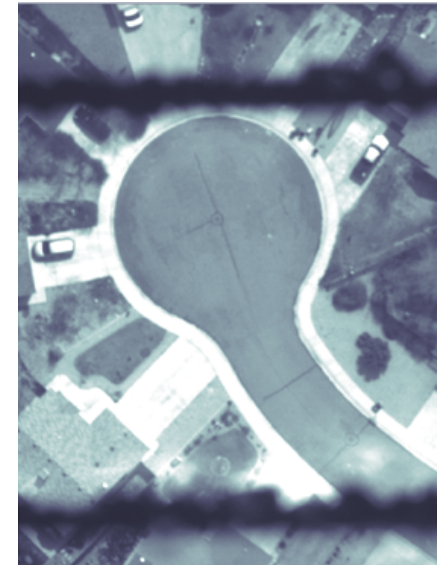   Band 9 1375 nm

Band 1 448 nm

Band 2 482 nm

Band 3 562 nm

Band 4 655 nm

Band 5 865 nm

Preliminary analysis indicates:
- Step-stare control and stabilization works well
- Active stabilization results in image stability on the order of optical quality
  - Aircraft environment more challenging than space
- In-process: Georegistration / mosaic image generation

Next Steps

- Preparing for Science Flights (September 2019)
  - Further laboratory verification
  - Software enhancements
  - Additional Radiometric/Spectral Calibration
- Science Flight Objectives
  - Landsat/Sentinel-2 under-flight
  - Vicarious calibration sites
  - Inland water collects
- Spaceborne instrument concept underway

Summary

- REMI utilizes a high performance scan mirror to achieve Landsat requirements with significant reduction in SWaP
  - Reduces image smear typical of Landsat instruments
  - Enables smaller aperture
  - Enables use of simple detector types
  - Key scan mirror technology has been space qualified on the GEMS and TEMPO programs
- REMI is fully integrated and tested
- Engineering flight completed
  - *Demonstrated successful use of scan mechanism for step-stare sampling of FOR and image stabilization*
- Science flights planned for later in 2019

Acknowledgements

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  - Paul Kaptchen, AI&T / Aircraft Ops
  - Nathan Leisso, Calibration
  - Nathan Showalter, Technical Support / Flight Team
  - Betsy Farris, Systems Engineering / Flight Team
  - Tom Kampe, Optical Engineering
  - Emily Mrkvicka, Optical Engineering
  - Bob Warden, Mechanical
  - Kyle Solander, Electronics
  - Jonathan Fox, Software / Flight Team
  - Homero Gutierrez, Scan Mechanism Controls
  - Nick Polaski, Controls Implementation
  - Lyle Ruppert, L1B Data Processing

THANK YOU!